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Active Contraction of the Left Ventricle with Cardiac Tissue Modelled as a Micromorphic Medium

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Declaration of Authorship

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Abstract

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Marina Kamper

In this study we investigate the behaviour of the left ventricular myocardium during a heartbeat. The myocardium is composed of interconnected cardiac fibres which are responsible for contraction of the heart chambers. Computational modelling of cardiac tissue presents several challenges because of its anisotropic, non-linear and time-dependent behaviour [85]. In addition to this, cardiac muscle tissue has a complex hierarchical material structure.

In general, cardiac tissue is treated as a non-linear elastic and incompressible material. Most computational studies employ the theories of classical continuum mechanics to model the passive response of the myocardium. Various classical continuum model have been successfully implemented to describe the behaviour of cardiac tissue. To take into account the distinct behaviour in the fibre and the sheet direction, these model typically assume the myocardium to be either transversely isotropic or orthotropic. Classical continuum models, however, do have several limitations. Most notably, they are unable to account for internal lengths or describe scale effects. If the characteristic length of the continuum is of the same order as the internal characteristic length, classical continuum models produce inadequate results [31].

Instead of a classical continuum formulation, we use a micromorphic continuum description for cardiac tissue. The complex microstructure and deformation experienced by cardiac tissue motivate the use of a micromorphic model. The micromorphic theory may be viewed as an extension of the classical continuum theory. Within a micromorphic continuum, particles are endowed with extra degrees of freedom that describe the deformation the continuum particle. Additional vectors, referred to as directors, are attached to the continuum particles. In this research, the directors are chosen such that they represent the deformation experienced by the cardiac fibres.

Furthermore, the total ventricular wall stress is additively decomposed into an active and a passive stress component. The active tension in the heart is taken to be a function of the sarcomere length, intracellular calcium concentration and the time after the onset of contraction. Contraction takes place in the fibre direction, described by the director field, and all the cardiac fibres are assumed to contract simultaneously.

The simulated results presented correspond well with typical mechanics observed in clinical experiments. This work demonstrates the potential of a micromorphic formulation for analysing and better understanding ventricular mechanics.

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Chapter 1

Introduction

Computational modelling of biological phenomena has attracted a great deal of attention over the last few decades. It allows researchers to predict the behaviour of complex biological systems and has the advantage that thousands of computational experiments can be performed to investigate the responses of these systems [7].

One of the most important biological structures in the human body is the heart. Computer-based cardiac¹ models provide insight into the interactions between physical phenomena and anatomical structures of the heart and there has been a steady rise in the accuracy and complexity of these models [109]. With the improvement and advent of new imaging modalities, it has become possible to accurately reconstruct the geometry and structure of the human heart. Additionally, an increase in computer power has led to faster and easier simulations [77].

Extensive computational heart modelling in the last half-century has brought about a variety of cardiac models utilised in different applications. Cardiac models are typically used to simulate the electrical activity and analyse the mechanical performance of the heart or to model the electromechanical coupling in heart fibres, see for example studies performed by Trayanova [123], Göktepe et al. [47] and Wong et al. [137]. The ultimate goal is to have these models be used in the medical environment to provide guidance during treatment of heart diseases. The rationale is that individual patients will respond differently to the same clinical intervention. Rather than applying a clinical treatment that suits the average patient, a physician can use a computational model to predict how a patient will respond to a specific intervention [48]. This is referred to as *patient-specific* treatment: a treatment is designed to fit the individual instead of the average patient.

1.1 Background

1.1.1 The Heart and Its Function

The heart is responsible for pumping blood through the cardiovascular system². As blood travels through the body, it supplies the organs with vital nutrients and also assists in the removal of metabolic waste [38]. The heart is composed of two pumps, a right and a left pump [79]. The right pump receives oxygen-poor blood from the body and pumps it to the lungs so that the blood

¹Cardiac: relating to the heart.

²Cardiovascular system: system in the body that circulates blood.

may become oxygenated. Oxygen-rich blood travels from the lungs to the left pump from where it is distributed to the rest of the body. As it moves through the body, the blood's oxygen content decreases and it returns to the right pump for the cycle to repeat. This path of blood flow is vital for survival.

Each pump is composed of an atrium and a ventricle. The two atria are responsible for receiving blood, whereas the two ventricles perform the main pumping action of the heart. By raising the blood pressure, the ventricles are able to distribute the blood through the cardiovascular system. The heart wall consists mainly of interconnected cardiac muscle fibres supported by soft connective tissue. The increase in blood pressure is accomplished by the contraction of these cardiac muscle fibres. Contraction is regulated by electrical impulses that are generated by the heart's natural pacemaker.

1.1.2 Modelling the Behaviour of Cardiac Tissue

Many studies related to cardiac modelling centre around simulating the behaviour of cardiac muscle tissue. Soft biological tissues, such as cardiac tissue, are anisotropic and heterogeneous materials that display time-dependent and non-linear behaviour [3, 42, 85]. Cardiac muscle fibres may experience changes in lengths of more than 20% during a heartbeat. It is therefore necessary to analyse its behaviour using a large (finite) deformation theory [85].

The mechanical response of cardiac tissue is typically separated into two components, the passive response and the active response. The former is usually described and predicted with the theories of classical continuum mechanics. Within the field of classical continuum mechanics, one neglects the intrinsic microstructure and only considers the macroscopic system [56]. Atomic-scale properties are averaged and the discontinuous atomic and molecular micro-structures are replaced by a smooth continuum body. According to Holzapfel [56], a classical continuum is a body composed of a continuous collection of point particles. Figure 1.1(a) illustrates the motion of a classical continuum body from its undeformed state (initial configuration) to its current state (deformed configuration). The reference and current positions of a point particle contained within the continuum are also shown.

The behaviour of all continua is governed by so-called balance laws, while material models (or constitutive equations) enable us to describe the responses of specific materials [100]. Material models for passive cardiac tissue have been documented extensively in the literature. Initially the passive response was described with simple isotropic models [27, 82], but more recent analyses employ transversely isotropic [25, 50] and orthotropic [58, 125] materials models.

The active component of the mechanical behaviour of cardiac tissue may be included in a number of ways. The active behaviour or active response refers to the ability of cardiac tissue to generate its own tension and contract when it is electrically stimulated. The contraction of cardiac tissue occurs as a result of a series of phenomena that take place on different length scales. To avoid over-complexity, an active model should yield results in line with experimental observations, but not explicitly include all the underlying mechanics [102]. The most common approach to incorporate the active response in a continuum formulation is with an *active-stress* model [2]. With this approach the stress experienced by the heart wall is additively decomposed into active and passive components and the active component is then simply included in the balance laws [2].

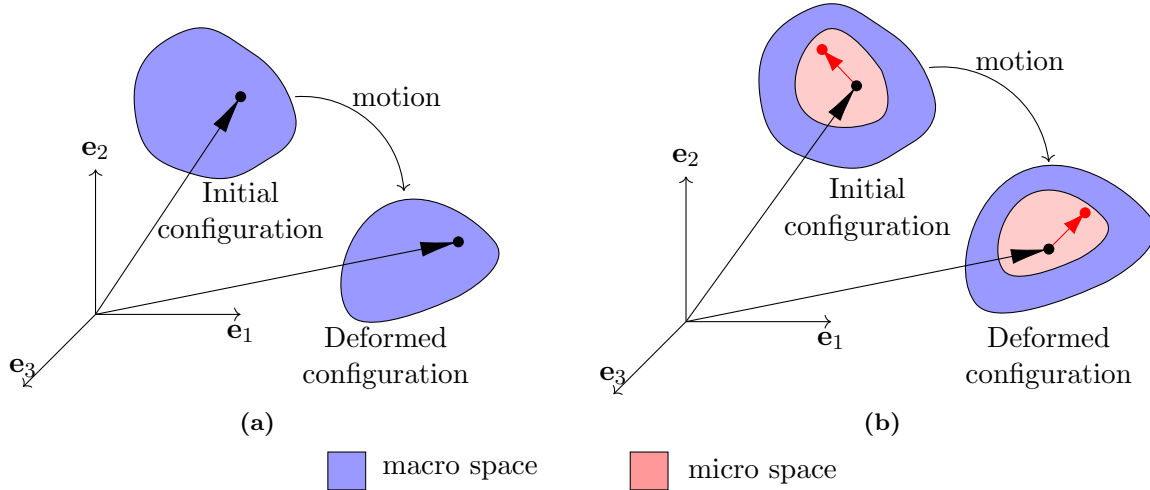


Figure 1.1: The difference between (a) the classical continuum that consists of a collection of point particles and (b) the micromorphic continuum composed of deformable particles.

1.1.3 Classical Continuum Theories vs Microcontinuum Modelling

Although the theories of classical continuum mechanics are valid for a wide range of scientific and engineering problems, there are a number of cases in which the classical theory cannot be justified. Classical continuum theories will fail to accurately predict the response of a body whose characteristic length is comparable to the size of the body constituents since these constituents may have individual responses that differ from the overall response of the body [30]. One such example is the flow of blood through capillaries. The size of the capillaries (characteristic length) is of the same order as the size of cells found in blood. The use of a classical continuum formulation to describe this behaviour yields results that are not in agreement with experimental observations. Other cases in which classical continuum mechanics might give unsatisfactory results include modelling materials such as foams, solids with micro-cracks, bones and materials with large porous structures [32].

To overcome some of these drawbacks, classical continuum theories can be extended to so-called *microcontinuum* theories where, instead of having infinitesimally small point particles, a continuum consists of a set of *deformable* particles. This is achieved by attaching additional vectors (also referred to as the *directors*) to the point particle, as illustrated by the red vector in Figure 1.1(b). This vector represents the deformation of the particle and as the microcontinuum experiences deformation, the director also deforms. The benefit is that a microcontinuum model takes into account material properties on the micro-scale and therefore gives further insight into multiscale phenomena.

One of the first studies to propose the idea of deformable particles was conducted by the Cosserat brothers [21]. In order to include characteristics of the material micro-structure, the Cosserat brothers introduced the concept of a rigid micro-particle. In this formulation a continuum particle is endowed with three additional degrees of freedom to describe the rotation of the particle. The particle is therefore able to experience macroscopic displacement as well as micro-rotation. Materials that exhibit this type of behaviour are referred to as *micropolar* or *Cosserat* continua.

The micropolar theory is fairly popular in bone mechanics. Bone, like cardiac tissue, has a hierarchical material structure. At the millimetre scale, bone appears to be a homogeneous material, but

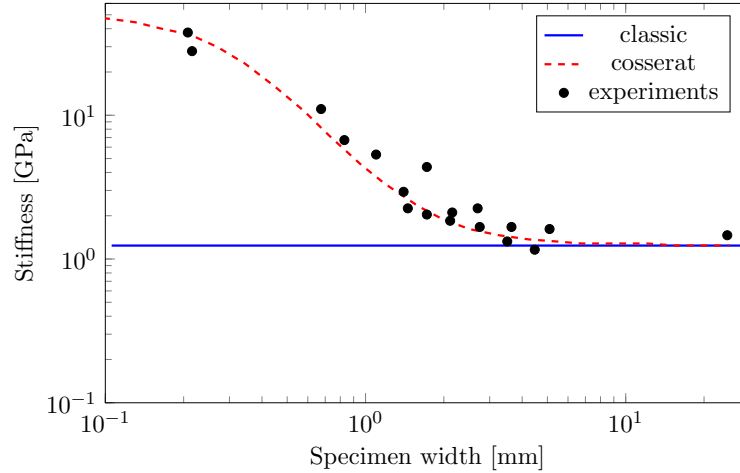


Figure 1.2: A collection of several datasets where the shear stiffness of bone were experimentally obtained using different sample sizes, adapted from [13].

if one considers scales closer to the micron scale, the heterogeneous structure becomes more evident [78]. Figure 1.2 presents experimental results from torsional testing performed on bone samples. The torsional stiffness of bone was obtained from specimens with different diameters. With bigger specimen samples, the stiffness is fairly well predicted by classical continuum mechanics. The reason is that as the specimen size becomes larger, the heterogeneity on the micro-scale becomes less dominant [78]. However, if one considers smaller sample sizes, the predicted classical stiffness deviates significantly from the actual stiffness. This is because the classical theory is only valid if the size of the micro-constituents is much smaller than the specimen size [13]. Since a micropolar formulation accounts for size-dependent material responses, it performed much better in predicting the torsional stiffness. The figure clearly demonstrates the advantage of a microcontinuum formulation.

Apart from the micropolar theory there are two other subclasses of microcontinuum theories, namely *microstretch* and *micromorphic*. A subclass is based on the constraints placed on the deformation of the particle [31]. In the case of a microstretch medium, a continuum particle is allowed to experience micro-rotation and micro-stretch. In the more general case of a micromorphic material, a particle is allowed to experience micro-rotation, micro-stretch as well as micro-shear.

1.2 Aims and Objectives

In this work we investigate the behaviour of a healthy heart, more specifically its bio-mechanical behaviour, i.e. we study the deformations and stresses experienced by the heart wall during a heartbeat. The left ventricle has to sustain much higher pressures compared to the right ventricle and is considered by some to be of higher importance [17]. Moreover, right ventricular infarction³ is very rare. Therefore this study, like many other computational studies of the heart, focuses on the left ventricle.

We propose the use of a micromorphic formulation to model the behaviour of a patient-specific left ventricle. A micromorphic model allows us to include features of the micro-structure and to model relative deformation between the cardiac fibres and the surrounding tissue. Microcontinuum

³Myocardial infarction (known as a heart attack): a sudden event where cardiac muscle is deprived of blood [79].

modelling has only recently been extended to modelling the response of cardiac tissue, see for example Sack et al. [111], Thureau et al. [121] and Von Hoegen et al. [131]. This work therefore aims to investigate whether a micromorphic model is suitable to describe cardiac tissue.

The layout and orientation of cardiac muscle fibres in the heart wall have a significant effect on the mechanical response and deformation of the heart [124]. With a micromorphic model, this project aims to capture more realistic material responses of cardiac muscle tissue during the filling, contraction and ejection stages of a heartbeat. In this work, one micro-director is defined for each continuum particle and the directors are chosen to align with the cardiac fibres. The directors in a micromorphic medium are allowed to distort [37], and this formulation therefore provides additional information about the deformation experienced by cardiac fibres.

A micromorphic material model that describes the passive behaviour of cardiac tissue has previously been implemented by the Computational Continuum Mechanics (CCM) Research Group at the University of Cape Town. The main goal of this project is to develop and implement an active-stress model that couples with the existing micromorphic model. To this end, we define the following objectives:

1. Review the literature on the mechanical behaviour of the heart, including constitutive models, heart geometries and the physiology of a healthy heart.
2. Develop and implement an active contraction model on SESKA, an in-house finite element software, that can be used together with an existing micromorphic model to simulate the mechanical behaviour of cardiac tissue. The main effort is to determine the additional contributions to the stiffness matrix and force vectors of the finite element formulation.
3. Perform simulations on SESKA:
 - (a) Simulate a heartbeat by applying applicable boundary and loading conditions.
 - (b) Calibrate the material model to match experimental pressure-volume data.
 - (c) Examine the deformation of the left ventricle and the ventricular wall stresses.
4. Determine the suitability of the micromorphic model by comparing the results to a heart simulation that uses a classical continuum formulation.

1.3 Thesis Outline

The thesis is structured as follows. In Chapter 2 the physiology of a healthy heart is discussed. While we largely consider the macroscopic structure and mechanics of the heart, we also look at the structure and behaviour of cardiac tissue at lower scales. Chapter 3 outlines the theories of classical continuum mechanics and introduces important terminology and nomenclature that are used throughout the remainder of this work. Additionally, Chapter 3 briefly explains the finite element procedure used when dealing with non-linear elastic materials. A number of cardiac models, including passive and active models, are reviewed in Chapter 4. The evolution of heart geometries used in computational models is also addressed. Chapter 5 serves as a basic introduction to the subject of micromorphic modelling and also gives examples from the literature where microcontinuum modelling is utilised to describe the behaviour of biological tissue. Chapter 6 provides the formulation of an active-stress model. The objective is to adapt a model documented in the literature and make

it compatible with the existing micromorphic model. Chapter 7 explains the methodology used to simulate the different phases of the heart cycle and briefly explains how the ventricular geometry was reconstructed from magnetic resonance images. In Chapter 8 we present the results from the micromorphic model and the classical continuum model. Finally, the thesis is concluded in Chapter 9 with a summary as well as suggestions for future work.

Chapter 2

Heart Physiology

This chapter reviews the workings of a healthy heart, including the mechanics of the ventricles, the different phases of the cardiac cycle, the layout of cardiac tissue in the heart wall and the contraction of cardiac fibres. The purpose is to present the medical nomenclature that is relevant to subsequent chapters.

2.1 Gross Structure of the Heart

The heart is often described as a muscular pump with its main purpose being to supply the organs of the body with blood [11, 69, 126]. Oxygen-rich blood provides cells with food while at the same time also removing left-over products from metabolic processes [38].

Figure 2.1 presents a cross-sectional view of the human heart. This muscular organ, located between the two lungs, consists of four chambers: the right and left atria and the right and left ventricles. The interventricular septum, a thick muscular wall, separates the two ventricle. Together the right atrium and right ventricle form the right pump of the heart and similarly the left atrium and left ventricle serve as the left pump.

Four one-way valves are found in the heart. The mitral valve and tricuspid valve, also referred to as the atrioventricular valves, separate the two atria from their adjoining ventricles. In a healthy heart, these valves permit blood flow in one direction only (from the atrium to its ventricle) and thereby preventing blood from flowing back into the atria. The semilunar valves, the pulmonary valve and the aortic valve, open to allow blood flow away from the ventricles to the body.

2.2 Circulation of Blood

As blood travels through the body it becomes partially depleted of oxygen. The oxygen-depleted blood returns to heart and enters the right atrium at a low pressure. The blood passes from the atrium through the tricuspid valve into the right ventricle. Contraction of the right ventricle increases the blood pressure and the blood is ejected into the pulmonary artery. The pulmonary arteries transfer the blood to the lungs where it is enriched with oxygen. Pulmonary veins then return the blood to the left atrium. The blood moves through the left atrium and bicuspid valve and fills the left ventricle. The ventricle contracts and blood is ejected into the aorta from where it

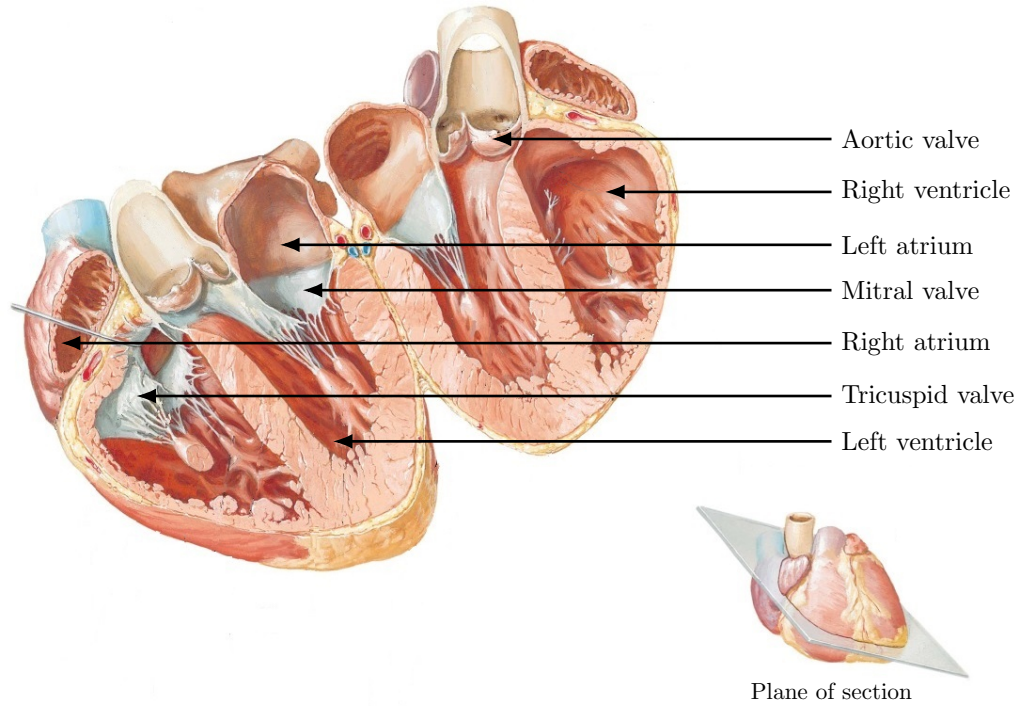


Figure 2.1: Cross-section of the human heart showing the four heart chambers and heart valves, adapted from [86].

is transported to the rest of the organ system. The phase in which the relaxed ventricles fill with blood is referred to as diastole while the phase in which the ventricles contract is termed systole.

The cardiovascular system can be separated into two main circulation loops: the pulmonary circuit and the systemic circuit. The pulmonary circuit is the path of blood as it moves from the right ventricle to the lungs and back to the left atrium. Blood passes through the systemic circuit as it travels from the left ventricle to the rest of the body and returns to the right atrium. The right ventricle is responsible for blood flow in the pulmonary circuit and likewise the left ventricle pumps blood through the systemic circuit.

2.3 Ventricular Mechanics

The main pumping power of the heart is supplied by the ventricles and as a result the ventricular walls are subjected to pressures that are much higher than the pressures experienced by the atrial walls. In this section we consider the function of the ventricles, specifically the pressure and volume changes during a heartbeat.

2.3.1 Ventricular Geometry

The idealised shape of the left ventricle is similar to an elongated ellipsoid truncated at its top [85, 113], while the right ventricle forms a crescent-like shape around the left ventricle [66, 85]. Since the ventricles develop higher pressures than the atria, the heart wall surrounding the ventricles is much thicker compared to that surrounding the atria [85]. Overall, the heart wall is thickest in the

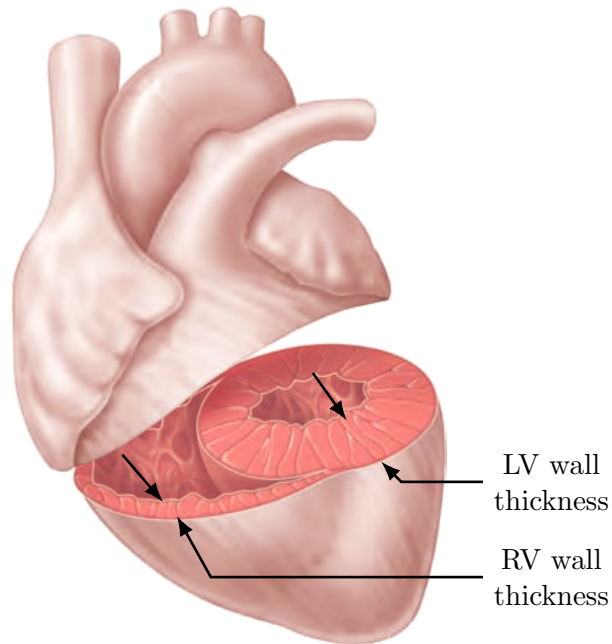


Figure 2.2: Cross-sectional view of the two ventricles illustrating the difference in wall thickness of the left and right ventricles, adapted from [79].

left ventricle, as illustrated in Figure 2.2. This is because the left ventricle is responsible for blood flow in the systemic circuit. The systemic circuit offers a greater resistance to blood flow than the pulmonary circuit. The left ventricle has to develop high pressures to overcome this resistance and as a result left ventricular pressure at the end of contraction is about three times as high as the pressure in the right ventricle [66]. The thick wall of the left ventricle ensures that the ventricle can sustain these high pressures.

2.3.2 Pressure-Volume Relationship

The pressure-volume loop shown in Figure 2.3 provides the relationship between the cavity volume and cavity pressure of a ventricle. It is a simple method to represent a full heartbeat. The curve is divided into four distinct phases namely (i) diastolic filling, (ii) isovolumetric contraction, (iii) ejection and (iv) isovolumetric relaxation.

At the start of diastolic filling (point A), the atrioventricular valve (inlet valve) of the ventricle opens and blood enters the chamber from the atrium. Ventricular filling continues until the ventricle reaches its final volume, known as the end-diastolic volume (EDV). At this point the atrioventricular valve closes, preventing any blood from further entering the ventricle. Point B marks the beginning of isovolumetric contraction. An electrical impulse activates the cardiac fibres and causes the ventricle to contract. Because the inlet and outlet valves are closed, the ventricular volume remains constant. This phase is also characterised by a rapid increase in pressure. Once the pressure in the ventricle reaches that of the great artery, the semilunar valve (outlet valve) opens. Blood is ejected into the artery from where it is distributed to the body. The closing of the semilunar valve marks the end of ejection (Point D). At the end of ejection the ventricular volume is at its lowest. This volume is known as the end-systolic volume (ESV). After the semilunar valve has closed, the

ventricle relaxes while its volume remains relatively unchanged. The ventricular pressure decreases until it is below that of the atrium. The pressure difference between the atrium and ventricle causes the atrioventricular valve to open. The opening of the inlet valve allows blood to flow into the ventricle and indicates the start of a new cycle. It is important to note that the pressure-volume curves for the left and right ventricles are similar in shape. However, the pressure and volume values of points A through D are different for the two ventricles.

The stroke volume (SV) is defined as the difference between the end-systolic and the end-diastolic volume,

$$SV = EDV - ESV. \quad (2.1)$$

The stroke volume is an important measurement to determine cardiac output [69]. The ejection fraction (EF) is a popular measure of ventricular function. It is the amount of blood ejected divided by the end-diastolic volume, that is

$$EF = \frac{SV}{EDV}. \quad (2.2)$$

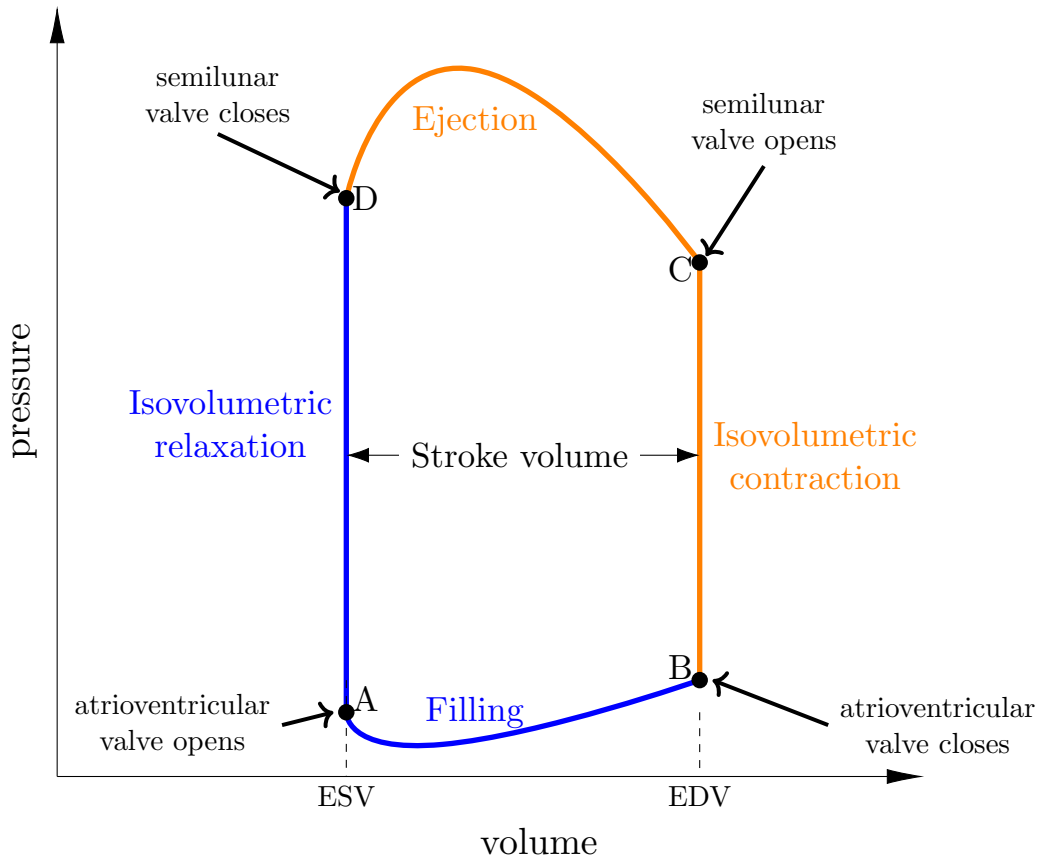


Figure 2.3: The pressure-volume relationship for the ventricle consists of four distinct phases. Diastole, indicated in blue, refers to the period during which the ventricle relaxes and subsequently fills. Systole, denoted in orange, comprises the contraction and the ejection phase.

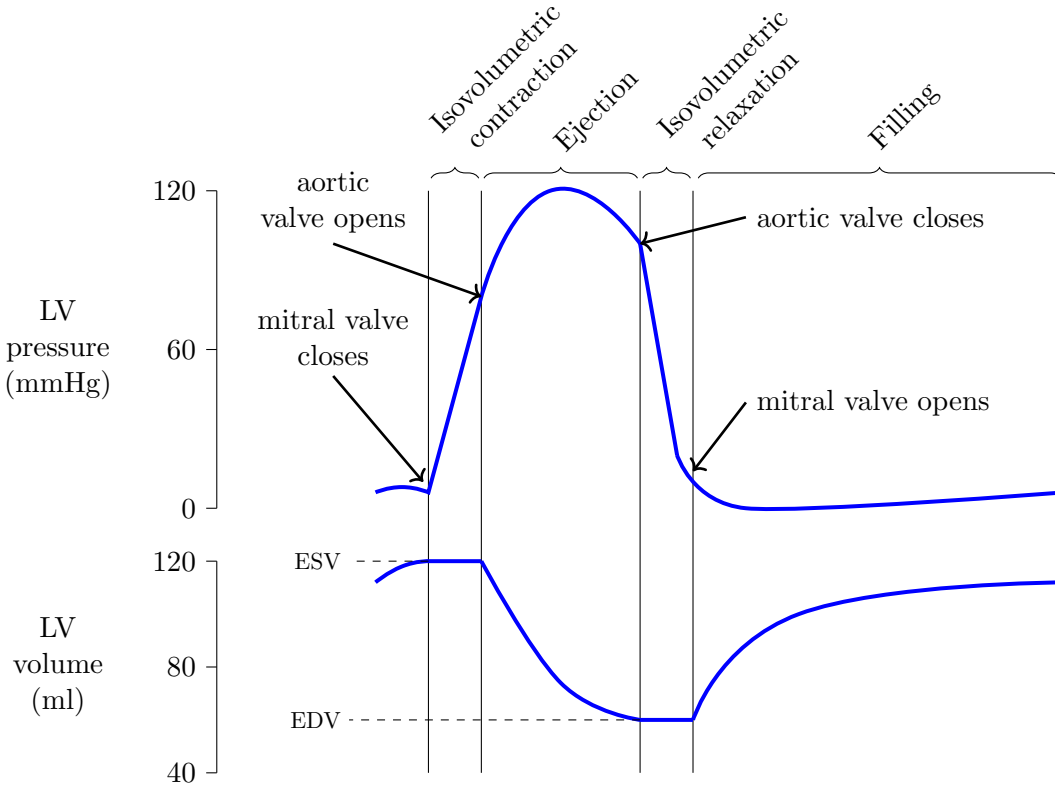


Figure 2.4: Variation in the left ventricular pressure and volume over time.

Figure 2.4 presents the temporal variation in the pressure and volume of the left ventricle. Similar to the pressure-volume loop, Figure 2.4 is divided into the four phases of the cardiac cycle, that is ventricular filling, isovolumetric contraction, ejection and isovolumetric relaxation. From Figure 2.4 we see that isovolumetric contraction and relaxation occur over relatively short time periods while the period of ventricular filling is the longest phase of the cardiac cycle.

2.3.3 Ventricular Motion

During systole the walls of the ventricles thicken [118] and the left ventricle makes a twisting motion about its long axis [108, 113]. Buchalter et al. [10] used magnetic resonance image tagging to measure the torsion experienced by the left ventricle. They found that when viewed from the apex the ventricle experiences a counterclockwise rotation, as seen in Figure 2.5. Studies have also observed longitudinal displacement of the ventricular base [106]. The base moves down towards the apex, causing a decrease in the distance between the base and apex. This results in an overall long-axis shortening of the ventricles. Additionally, Codreanu et al. [20] have observed an upward motion of the apex during systole.

In early diastole the ventricles begin to relax [69] and the left ventricle untwists [98]. As the ventricles fill with blood, they expand. The ventricular walls move outward and the distance between the apex and the base increases. The motion patterns during systole and diastole may be explained by the arrangement of cardiac fibres in the ventricles [20].

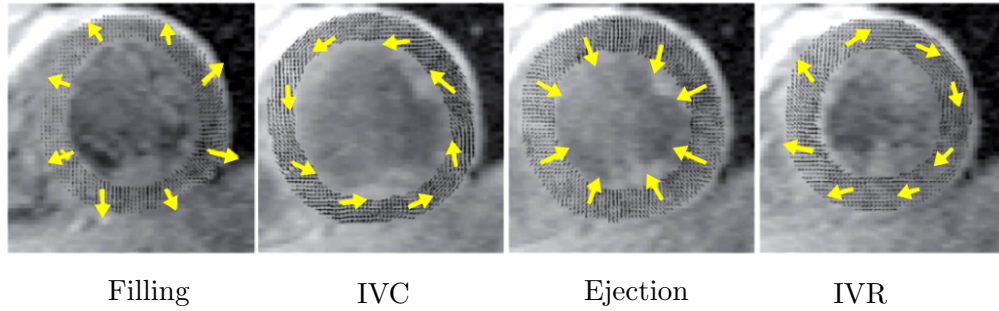


Figure 2.5: Short axis images of the mid-ventricle from [65] showing the motion of the left ventricle during diastolic filling, isovolumetric contraction (IVC), ejection and isovolumetric relaxation (IVR). Arrows represent the in-plane velocity field of the ventricle.

2.4 Cardiac Muscle Tissue

The wall of the heart is divided into three layers: the endocardium, myocardium and epicardium. The endocardium (*endo* = within) is the thin inner layer of the heart wall. It is a serous membrane that lines the inside of the atria and ventricles [16, 58]. The thick middle layer, or myocardium (*myo* = muscle), consists of contracting muscle tissue and makes up the largest part of the heart wall [61]. Cardiac muscle fibres are joined together to form the complex structure of the myocardium and it is these fibres that are responsible for the pumping action of the heart. The myocardium has a non-uniform thickness. It is thickest in the left ventricle and thinnest in the atria. The final layer, the epicardium (*epi* = above), forms the thin outer layer of the heart wall.

2.4.1 Structure of Cardiac Tissue

The three major types of muscle tissue found in the body are skeletal, smooth and cardiac muscle tissue. The latter is found only in the heart [38]. Cardiac muscle tissue consists of short branched cardiac muscle fibres, also referred to as cardiac muscle cells or myocytes. The length of a cardiac muscle fibre varies between 80 and 100 μm while its diameter ranges from 10 to 20 μm [85]. The fibres are connected at intercalated discs, which appear as dark lines under a microscope, see Figure 2.6. Because intercalated discs form a mechanical and electrical connection between adjacent cells, the myocardium functions as a single unit [38]. Figure 2.6 also shows the striated appearance (alternating light and dark lines) of cardiac muscle tissue. The light and dark lines are referred to as I-bands and A-bands respectively.

A single muscle fibre consists of subunits called fibrils. The striated appearance of the muscle tissue is a result of different filaments found in a fibril. The dark A-bands contain so-called thick filaments, which consist mainly of the protein myosin, while the light I-bands contain thin filaments, composed of the protein actin. The thick filaments cause the darker appearance of the A-band. Figure 2.7 illustrates the hierarchical structure of a cardiac fibre. The dark line in the middle of the I-band is called a Z-line. As indicated in Figure 2.7, a subunit from Z-line to Z-line is referred to as a sarcomere. Titin, a large protein, joins the thick filaments to the Z-lines. The ends of the thin filaments do not stop at the edge of the I-band. Rather, they extend into the A-band and overlap with the thick filaments, as shown in Figure 2.8(a).

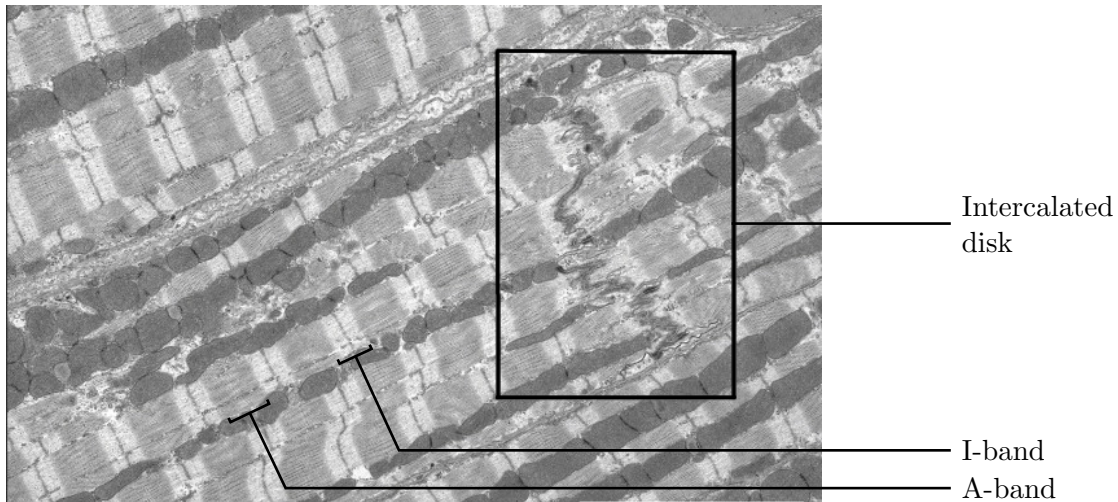


Figure 2.6: Cardiac muscle tissue as seen from under an electron microscope, modified from [91]. The electron micrograph shows the branched shape and striated appearance of cardiac muscle tissue.

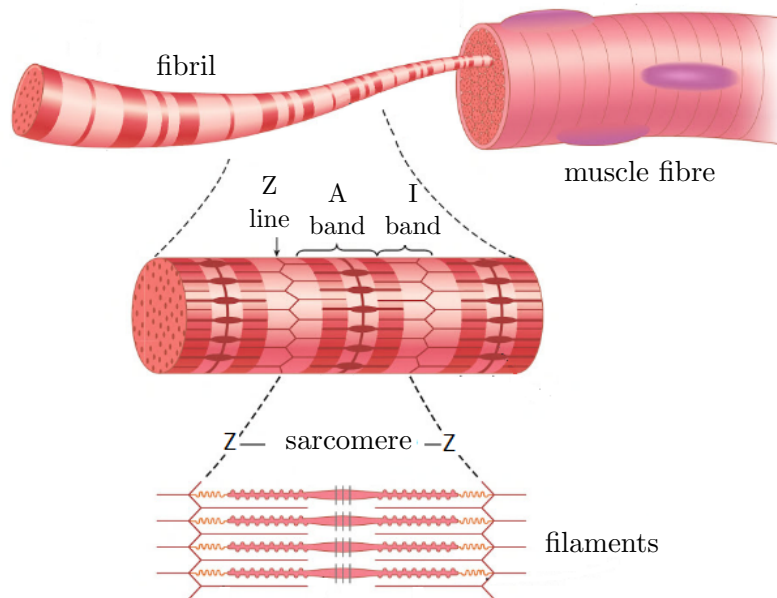


Figure 2.7: Schematic diagram modified from [52] showing the hierarchical structure of muscle tissue. The alternating thick and thin filaments give cardiac muscle tissue its striated appearance. A sarcomere extends from Z-line to Z-line and is the contractile unit of a muscle fibre.

2.4.2 Contraction of Cardiac Tissue

Contraction of muscle tissue is accomplished by the contraction of individual fibres. At microscopic level, the contraction of fibres is often described with the *sliding filament theory* [38]. To produce fibre contraction, thin filaments in the fibril slide further in between the thick filaments, see Figure 2.8(a). Because the overlap between the thick and thin filaments increases, the dark A-bands move closer together. It is important to note that the lengths of the filaments remain unchanged. Cross-bridges (also referred to as myosin heads), which form part of the thick filaments, pull the thin filaments to achieve this sliding motion. As the distance between neighbouring Z-lines decreases, the sarcomeres and hence the fibrils shorten. The shortening of the fibrils results in the contraction of the muscle fibre.

Contractile proteins in the heart regulate the contraction and relaxation of cardiac muscle fibres. Apart from actin, thin filaments also contain tropomyosin and the three proteins of the troponin complex, namely troponin-I (TN-I), troponin-T (TN-T) and troponin-C (TN-C). These proteins are shown in Figure 2.8(b). For a fibre to contract cross-bridges have to interact with binding sites on the actin protein. In the relaxed muscle the tropomyosin is held in position by the troponin complex such that it inhibits the cross-bridges from attaching to the thin filaments.

Calcium ions are produced when the fibre is stimulated by an electrical impulse. The calcium ions bind to troponin-C causing the tropomyosin to move away and expose active binding sites for the cross-bridges. Conversely, when the calcium ions are removed, the tropomyosin resumes its inhibitory position. It is clear that energy is spent on the micro-scale to obtain deformation on the macro-scale [2].

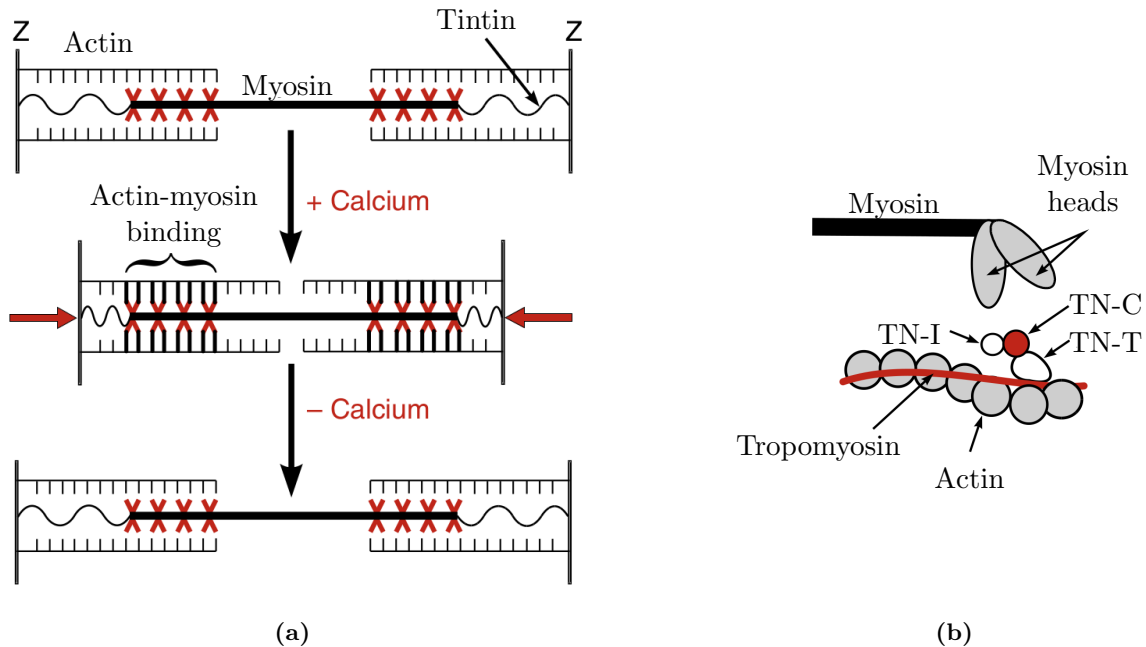


Figure 2.8: Schematic diagrams illustrating (a) the sliding filament theory that is used to explain the mechanics of muscle contraction on the microscopic level and (b) the different proteins found in the myofilaments, from [69].

2.4.3 Electrical Conduction

To contract, cardiac muscle fibres need to be electrically excited [96]. In a healthy heart, the electrical activity is regulated by the natural pacemaker of the heart, the sinoatrial node located in the right atrium. The sinoatrial node consists of special cells that are capable of generating electrical impulses. The electrical impulse travels from fibre to fibre across the atria causing the right and left atria to contract. The atria are separated from the ventricles through a layer of connective tissue that surrounds the atrioventricular valves. This layer prevents the impulse from travelling directly to the ventricles and special conducting tissue is therefore required [38]. This conducting tissue allows the impulse to travel through the ventricles. The left and right ventricles contract approximately 0.1 to 0.2 seconds after the atria have contracted [38].

2.4.4 Tissue Layout in the Ventricles

The cardiac muscle fibres are arranged in a complex pattern in the myocardium. Surrounding the muscle fibres are types I and II collagen, which make up the majority of the extracellular matrix [61]. The organisation of the muscle fibres in the myocardium is often described in one of two ways [45, 58]. The first considers the heart to be one continuous muscle that is coiled into two loops or helices [12]. The second assumes the heart to consist of layers (or sheets), which contain predominately cardiac muscle fibres [142]. The layers are usually three to four fibres thick [58] and are allowed to slide with respect to each other during contraction and filling [75]. Several authors adopt the latter approach, see for example [3, 47, 58, 85]. Figure 2.9 illustrates the layout of fibres in the ventricle and depicts the muscle layers as red lines. A micrograph of cardiac tissue is given in Figure 2.10 and clearly shows its laminar structure.

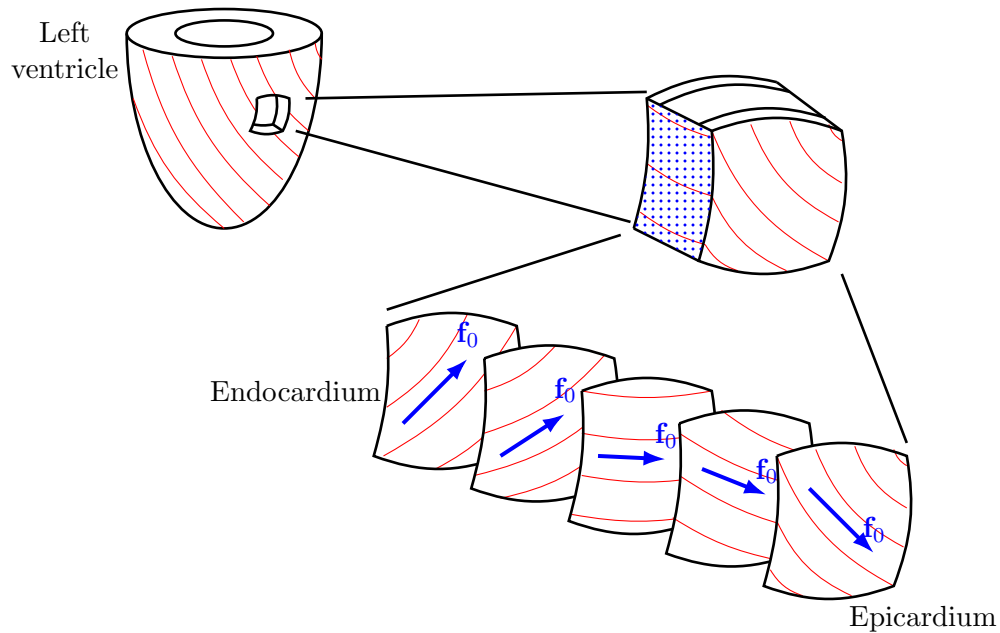


Figure 2.9: Schematic illustration of the cardiac tissue layout in the left ventricle as well as the transmural variation in the fibre direction \mathbf{f}_0 , adapted from [97].

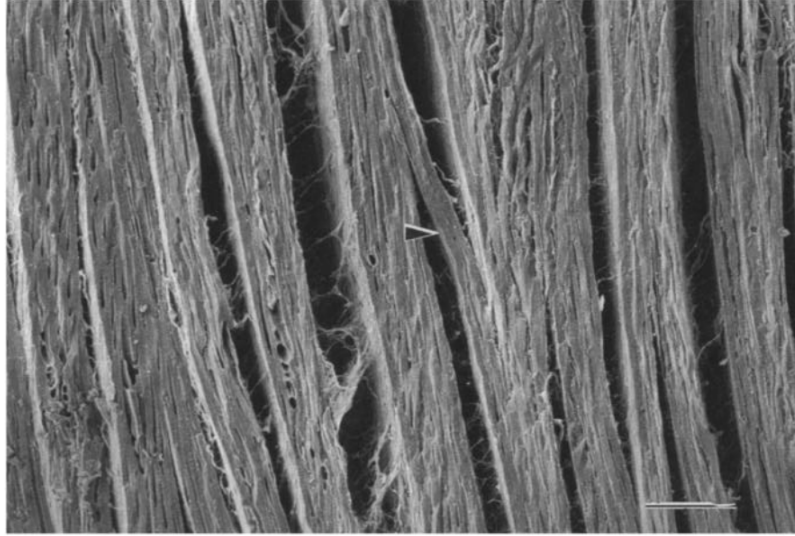


Figure 2.10: Electron micrograph showing the layered organisation of cardiac tissue in the heart wall, from [74].

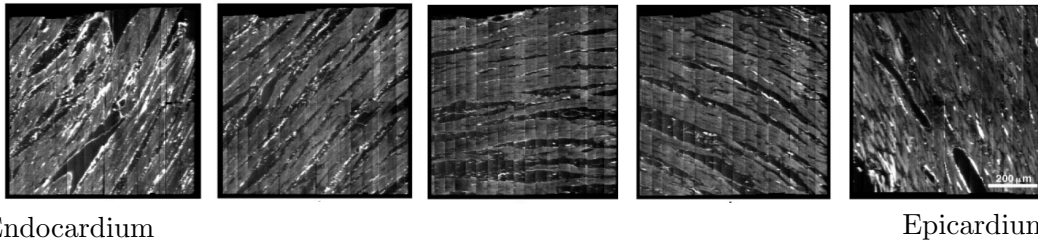


Figure 2.11: Microscopic images of cardiac tissue reproduced from [142] showing the variation in the fibre direction across the ventricular wall.

The longitudinal axis of the muscle fibre is referred to as the fibre direction. Since the electrical impulse travels approximately 3 times faster along the longitudinal axis of the fibre than in the transverse direction, the propagation of the impulse is highly dependent on the fibre layout [18]. At the epicardial region, the fibres are arranged in a left-handed spiral, which gradually changes to a right-handed spiral pattern near the endocardium. Cross-sectional images of the ventricular wall are provided in Figure 2.11 and show the variation of the fibre orientation through the myocardial wall. At the epicardium, the fibre angle is between -50° and -70° . The angle changes through the wall to an angle of $+50^\circ$ to $+70^\circ$ at the endocaridial region [61, 118, 142].

2.5 Summary of the Cardiac Physiology

The purpose of this chapter was to review the function of a healthy heart and to discuss briefly the micro-mechanics of fibre contraction. We noted that a heartbeat is divided into four phases with distinct pressure-volume relations. To perform a computational investigation of the heart it is necessary determine an appropriate method to simulate these phases, as presented in later chapters.

For cardiac tissue, like most biological tissues, one finds internal structures at different length scales. The macroscopic behaviour of the heart is determined by the dynamics of cardiac tissue

on the microscale. For example, the sliding of filaments on the microscale is responsible for the contraction of muscle fibres and therefore also the contraction of the heart chambers.

We also discussed the layout of cardiac fibres in the heart and noted that the direction of the fibres varies across the ventricular wall. For example, at the outer surface of the left ventricle the fibres form a left-handed helix whereas the cardiac fibres near the inner surface are arranged in a right-handed helix. Cardiac function is highly dependent on this arrangement. The layout of the cardiac fibres causes the ventricles to experience different types of motions such as twisting, wall thickening and untwisting.

Chapter 3

Non-Linear Continuum Mechanics

The theories of classical continuum mechanics are utilised extensively in the engineering environment to describe the behaviour of structures and materials. A classical continuum formulation neglects features on the anatomic and molecular scales and assumes that a body is composed of a continuous, rather than a discrete, distribution of continuum particles. A comprehensive overview of continuum mechanics can be found in [56].

In general, the continuum mechanics approach consists of (i) deriving general principles that are valid for all materials be it solids or fluids and (ii) developing constitutive models to describe the behaviour of individual materials. Both of these components are dealt with in subsequent sections. Because we ultimately want to simulate the behaviour of cardiac tissue, this chapter also introduces the concept of hyperelasticity, a popular type of constitutive model used to describe non-linear elastic material behaviour. Finally, the chapter reviews the finite element method and briefly describes curvilinear coordinate systems.

3.1 Tensor Preliminaries

A second-order tensor \mathbf{T} is a mathematical entity that linearly maps a vector to another vector,

$$\mathbf{b} = \mathbf{T}\mathbf{a} . \quad (3.1)$$

In the above, the tensor \mathbf{T} acts on vector \mathbf{a} to produce vector \mathbf{b} . Vectors are also referred to as first-order tensors. We can extend Equation (3.1) to higher order tensors, for example

$$\mathbf{A} = \mathbf{CB} , \quad (3.2)$$

where \mathbf{C} is a fourth-order tensor and \mathbf{A} and \mathbf{B} are second-order tensors. In the Cartesian space, we express tensors in terms of their components as follows,

$$\begin{array}{ll} \text{scalar (zeroth-order): } \rho & \text{first-order: } \mathbf{a} = a_i \mathbf{e}_i \\ \text{second-order: } \mathbf{A} = A_{ij} \mathbf{e}_i \otimes \mathbf{e}_j & \text{fourth-order: } \mathbf{A} = \mathbb{A}_{ijkl} \mathbf{e}_i \otimes \mathbf{e}_j \otimes \mathbf{e}_k \otimes \mathbf{e}_l \end{array}$$

where \otimes is the dyadic or tensor product. In this text, scalars are denoted with lowercase Greek symbols, first-order tensors with lowercase boldface Roman letters, second-order tensors with uppercase boldface Roman letters and fourth-order tensors with blackboard bold letters.

The previous expressions utilise the Einstein summation convention where a repeated index denotes a summation over that index,

$$a_i b_i c_j = \sum_{i=1}^n a_i b_i c_j . \quad (3.3)$$

The repeated index i is referred to as the dummy index and indices that appear only once are called free indices.

3.1.1 Symmetry

Any tensor \mathbf{T} can be additively decomposed into a symmetric component \mathbf{S} and a skew-symmetric component \mathbf{W} . That is,

$$\mathbf{T} = \mathbf{S} + \mathbf{W} , \quad (3.4)$$

where \mathbf{S} and \mathbf{W} are defined as

$$\mathbf{S} = \mathbf{T}^{\text{symm}} = \frac{1}{2}(\mathbf{T} + \mathbf{T}^T) , \quad (3.5)$$

$$\mathbf{W} = \mathbf{T}^{\text{skew}} = \frac{1}{2}(\mathbf{T} - \mathbf{T}^T) . \quad (3.6)$$

\mathbf{T}^T in the above denotes the transpose of the tensor \mathbf{T} . A symmetric tensor \mathbf{S} and skew-symmetric tensor \mathbf{W} satisfy the following,

$$\mathbf{S} = \mathbf{S}^T \quad \text{and} \quad \mathbf{W} = -\mathbf{W}^T . \quad (3.7)$$

3.1.2 Eigenvalue Problem

A second-order tensor \mathbf{T} has the following property,

$$\mathbf{T}\mathbf{n} = \lambda\mathbf{n} , \quad (3.8)$$

where \mathbf{n} is the eigenvector and λ the eigenvalue of \mathbf{T} . Equation (3.8) is known as the eigenvalue problem and to solve it we rewrite the expression as

$$(\mathbf{T} - \lambda\mathbf{I})\mathbf{n} = \mathbf{0} , \quad (3.9)$$

where \mathbf{I} is the identity matrix. For non-trivial solutions of \mathbf{n} , the inverse of $(\mathbf{T} - \lambda\mathbf{I})$ does not exist and as a result

$$\det(\mathbf{T} - \lambda\mathbf{I}) = 0 . \quad (3.10)$$

Solving Equation (3.10) leads to the characteristic equation,

$$\lambda^3 - I_1\lambda^2 + I_2\lambda - I_3 = 0 , \quad (3.11)$$

where I_i are the scalar invariants of \mathbf{T} . I_i remain unchanged irrespective of the coordinate system in which the tensor is defined. This is an important property that is frequently exploited in continuum mechanics. For example, material laws are often defined in terms of the scalar invariants to ensure that it is independent of the chosen coordinate system.

The invariants of \mathbf{T} are given as

$$\begin{aligned} I_1 &= \text{tr} \mathbf{T} , \\ I_2 &= \frac{1}{2} ((\text{tr} \mathbf{T})^2 - \text{tr} \mathbf{T}^2) , \\ I_3 &= \det \mathbf{T} . \end{aligned} \quad (3.12)$$

The tensor \mathbf{T} can also be expressed in terms its eigenvalues,

$$\mathbf{T} = \begin{bmatrix} \lambda_1 & 0 & 0 \\ 0 & \lambda_2 & 0 \\ 0 & 0 & \lambda_3 \end{bmatrix} . \quad (3.13)$$

3.2 Kinematics

The deformation of a three-dimensional continuum body in the Euclidean reference frame is shown in Figure 3.1. At time t_0 the body is in its undeformed state and occupies the space \mathcal{B}_0 . The undeformed state is also called the initial or reference configuration. The body undergoes deformation and at some time t , it occupies the space \mathcal{B} . The deformed state is referred to as the current or spatial configuration. The motion φ from the reference position to the current position is given by

$$\mathbf{x} = \varphi(\mathbf{X}, t) . \quad (3.14)$$

The displacement of the continuum particle P is defined as

$$\mathbf{u}(\mathbf{X}, t) = \mathbf{x}(\mathbf{X}, t) - \mathbf{X} , \quad (3.15)$$

where \mathbf{X} is the initially position of the particle and \mathbf{x} the current position at time t .

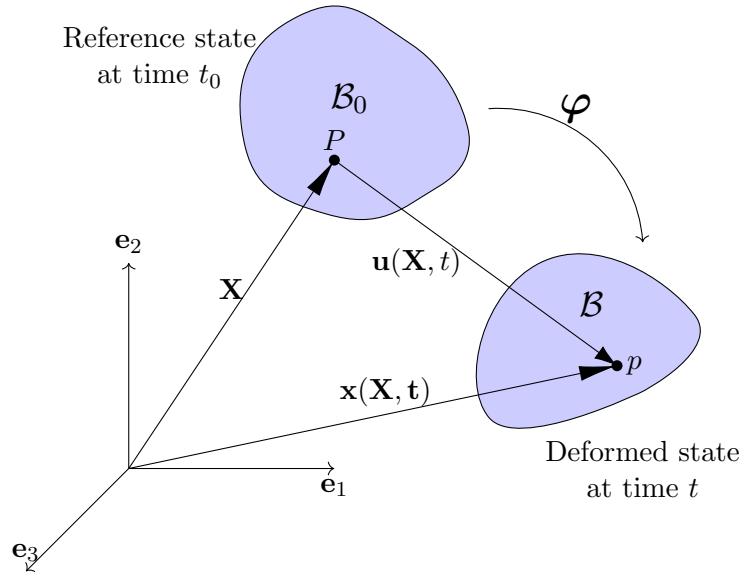


Figure 3.1: Motion of a continuum body in the Euclidean space.

The convention is to use upper and lower case letters to denote the initial and current configurations respectively. The gradient of a field (\bullet) may be with respect to either the reference or the current configuration,

$$\text{Grad}(\bullet) = (\bullet) \otimes \nabla_{\mathbf{X}} = \frac{\partial(\bullet)}{\partial X_j} \otimes \mathbf{e}_j, \quad \text{grad}(\bullet) = (\bullet) \otimes \nabla_{\mathbf{x}} = \frac{\partial(\bullet)}{\partial x_j} \otimes \mathbf{e}_j. \quad (3.16)$$

The divergence of a vector field (\bullet) is defined as the scalar product of the gradient operator and the field,

$$\text{Div}(\bullet) = (\bullet) \cdot \nabla_{\mathbf{X}} = \frac{\partial(\bullet)}{\partial X_j} \cdot \mathbf{e}_j, \quad \text{div}(\bullet) = (\bullet) \cdot \nabla_{\mathbf{x}} = \frac{\partial(\bullet)}{\partial x_j} \cdot \mathbf{e}_j. \quad (3.17)$$

The deformation gradient \mathbf{F} is the derivative of the spacial position \mathbf{x} with respect to the reference position,

$$\mathbf{F}(\mathbf{X}, t) = \frac{\partial \mathbf{x}}{\partial \mathbf{X}} = \frac{\partial(\mathbf{X} + \mathbf{u})}{\partial \mathbf{X}} = \mathbf{I} + \nabla_{\mathbf{X}} \mathbf{u}, \quad (3.18)$$

where \mathbf{I} is the identity matrix. From the definition of differential $d\mathbf{x}$,

$$d\mathbf{x} = \frac{\partial \mathbf{x}}{\partial \mathbf{X}} d\mathbf{X} = \mathbf{F} d\mathbf{X}, \quad (3.19)$$

we see that the deformation gradient maps an infinitesimal line element in the reference configuration to a line element in the deformed configuration. The determinant of the deformation gradient gives the change in volume of an infinitesimal volume element,

$$J = \det \mathbf{F} = \frac{dv}{dV}, \quad (3.20)$$

where J is known as the Jacobian and dv and dV are the infinitesimal volumes in the current and the reference configuration respectively. Values of $J \leq 0$ result in either negative or zero volumes, i.e. unrealistic motion, and as a consequence only motions where $J > 0$ are allowed.

Nanson's formula provides a relationship between an infinitesimal area element $d\mathbf{A}$ defined in the reference configuration and an area element $d\mathbf{a}$ in the current configuration,

$$d\mathbf{a}\mathbf{n} = J\mathbf{F}^{-T} d\mathbf{A}\mathbf{N}, \quad (3.21)$$

where \mathbf{N} and \mathbf{n} are unit normals to the area elements. Another useful identity is

$$\frac{DJ}{Dt} = \dot{J} = J \text{div} \mathbf{v}, \quad (3.22)$$

where \dot{J} is the time derivative of the Jacobian and $\mathbf{v} = \frac{d\mathbf{x}}{dt}$ is the velocity field.

3.2.1 Time Derivatives

The velocity field \mathbf{v} and acceleration field \mathbf{a} in the material description, i.e. expressed in terms of the reference coordinates \mathbf{X} , are given as

$$\mathbf{v}(\mathbf{X}) = \frac{\partial \mathbf{x}(\mathbf{X}, t)}{\partial t}, \quad \mathbf{a}(\mathbf{X}) = \frac{\partial \mathbf{v}(\mathbf{X}, t)}{\partial t}. \quad (3.23)$$

If a scalar field ϕ is expressed in terms of the spatial coordinates \mathbf{x} , the chain rule is used to obtain the material time derivative,

$$\dot{\phi}(\mathbf{x}, t) = \frac{D\phi}{Dt} = \frac{\partial\phi}{\partial t} + \frac{\partial\phi}{\partial\mathbf{x}} \frac{\partial\mathbf{x}}{\partial t}. \quad (3.24)$$

Similarly, for the velocity field \mathbf{v} defined in the current configuration, we have

$$\begin{aligned} \dot{\mathbf{v}}(\mathbf{x}, t) = \frac{D\mathbf{v}}{Dt} &= \frac{\partial\mathbf{v}}{\partial t} + \frac{\partial\mathbf{v}}{\partial\mathbf{x}} \frac{\partial\mathbf{x}}{\partial t} \\ &= \frac{\partial\mathbf{v}}{\partial t} + (\text{grad}\mathbf{v})\mathbf{v}. \end{aligned} \quad (3.25)$$

The symmetric part of the velocity gradient is known as the rate of deformation \mathbf{d} and the skew-symmetric part is referred to as the spin $\boldsymbol{\omega}$,

$$\mathbf{d} = \mathbf{l}^{\text{symm}} = \frac{1}{2}(\mathbf{l} + \mathbf{l}^T), \quad (3.26)$$

$$\boldsymbol{\omega} = \mathbf{l}^{\text{skew}} = \frac{1}{2}(\mathbf{l} - \mathbf{l}^T), \quad (3.27)$$

where $\mathbf{l} = \text{grad}\mathbf{v}$ is the velocity gradient. The time derivative of the deformation gradient can be expressed in terms of the velocity gradient as

$$\dot{\mathbf{F}} = \mathbf{l}\mathbf{F}. \quad (3.28)$$

3.3 Strain Measures

Strain is a measure of the deformation experienced by the continuum body. In this section we define a number of strain measures often utilised in non-linear continuum mechanics.

One of the simplest and most commonly-used strain measures is the *average normal strain* ϵ , defined as

$$\epsilon = \frac{\Delta L}{L_0}, \quad (3.29)$$

where L_0 is the original length of the specimen and ΔL is the change in the specimen length. If ds and dS represent the magnitudes of the infinitesimal line elements $d\mathbf{x}$ and $d\mathbf{X}$ then we have,

$$(ds)^2 = d\mathbf{x} \cdot d\mathbf{x} \quad \text{and} \quad (dS)^2 = d\mathbf{X} \cdot d\mathbf{X}, \quad (3.30)$$

where $d\mathbf{x}$ and $d\mathbf{X}$ are defined in the current and the reference configuration respectively. Consider the difference between the squares of the magnitudes,

$$\begin{aligned} (ds)^2 - (dS)^2 &= d\mathbf{x} \cdot d\mathbf{x} - d\mathbf{X} \cdot d\mathbf{X} \\ &= \mathbf{F}d\mathbf{X} \cdot \mathbf{F}d\mathbf{X} - d\mathbf{X} \cdot d\mathbf{X} \\ &= d\mathbf{X} \cdot \mathbf{F}^T \mathbf{F} d\mathbf{X} - d\mathbf{X} \cdot d\mathbf{X} \\ &= \mathbf{F}^T \mathbf{F} : d\mathbf{X} \otimes d\mathbf{X} - \mathbf{I} : d\mathbf{X} \otimes d\mathbf{X} \\ &= (\mathbf{F}^T \mathbf{F} - \mathbf{I}) : d\mathbf{X} \otimes d\mathbf{X}. \end{aligned} \quad (3.31)$$

Two deformation tensors frequently found in continuum mechanics texts are the right Cauchy-Green deformation tensor \mathbf{C} and the Green-Lagrange strain tensor \mathbf{E} . These tensors are defined as

$$\mathbf{C} = \mathbf{F}^T \mathbf{F}, \quad (3.32)$$

$$\mathbf{E} = \frac{1}{2}(\mathbf{F}^T \mathbf{F} - \mathbf{I}) = \frac{1}{2}(\mathbf{C} - \mathbf{I}). \quad (3.33)$$

We can therefore write Equation (3.31) in term of the deformation tensors as

$$\begin{aligned} (ds)^2 - (dS)^2 &= (\mathbf{C} - \mathbf{I}) : d\mathbf{X} \otimes d\mathbf{X} \\ &= 2\mathbf{E} : d\mathbf{X} \otimes d\mathbf{X}. \end{aligned} \quad (3.34)$$

Similarly, it can be shown that

$$\begin{aligned} (ds)^2 - (dS)^2 &= (\mathbf{I} - (\mathbf{F}\mathbf{F}^T)^{-1}) : d\mathbf{x} \otimes d\mathbf{x} \\ &= (\mathbf{I} - \mathbf{b}^{-1}) : d\mathbf{x} \otimes d\mathbf{x} \\ &= 2\mathbf{e} : d\mathbf{x} \otimes d\mathbf{x}, \end{aligned} \quad (3.35)$$

where we have defined the left Cauchy-Green deformation tensor \mathbf{b} as

$$\mathbf{b} = \mathbf{F}\mathbf{F}^T, \quad (3.36)$$

and the Euler-Almansi strain tensor \mathbf{e} as

$$\mathbf{e} = \frac{1}{2}(\mathbf{I} - \mathbf{b}^{-1}). \quad (3.37)$$

3.4 Stress Measures

Forces acting on the body cause deformation. Within the framework of continuum mechanics, we identify two main types of forces. A *body force* \mathbf{b} acts throughout the entire volume of the body while a *traction* or *surface force* \mathbf{t} acts on some surface (external or internal) of the body.

Consider the surface shown in Figure 3.2 that passes through a point q in the body \mathcal{B} . At q the surface has a unit normal \mathbf{n} . If $\Delta \mathbf{f}$ is the resultant force acting on an infinitesimal area element Δa on the surface, then Cauchy's stress principle states that the traction is given by

$$\mathbf{t} = \mathbf{t}(\mathbf{x}, t, \mathbf{n}) = \lim_{a \rightarrow 0} \frac{\Delta \mathbf{f}}{\Delta a} = \frac{d\mathbf{f}}{da}. \quad (3.38)$$

The traction at the current position of point q is a function of the unit normal \mathbf{n} . This dependence can be expressed with a linear transformation,

$$\mathbf{t}(\mathbf{x}, t, \mathbf{n}) = \boldsymbol{\sigma}(\mathbf{x}, t) \mathbf{n}. \quad (3.39)$$

The linear transformation $\boldsymbol{\sigma}$ is known as the Cauchy stress tensor. The stress may also be expressed in the reference configuration ,

$$\mathbf{t}'(\mathbf{X}, t, \mathbf{N}) = \mathbf{P}(\mathbf{X}, t) \mathbf{N}, \quad (3.40)$$

where \mathbf{X} is the position in the reference state and \mathbf{N} the normal of the area element dA .

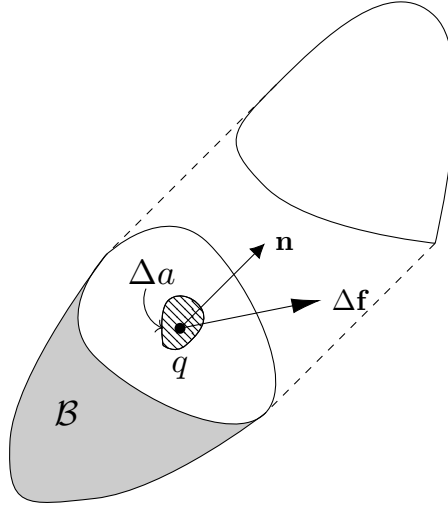


Figure 3.2: Force acting on a plane that passes through an arbitrary point in the body.

The linear transformation \mathbf{P} in Equation (3.40) is called the first Piola-Kirchhoff stress tensor. To relate the first Piola-Kirchhoff stress to Cauchy stress we define \mathbf{P} such that the force $d\mathbf{f}$ acting on da is the same as the force acting on dA . That is,

$$d\mathbf{f} = \mathbf{P}N dA = \mathbf{T}n da . \quad (3.41)$$

Furthermore, we utilise Nanson's formula,

$$\mathbf{P}N dA = \mathbf{T}n da = \mathbf{T}\mathbf{J}\mathbf{F}^{-T}N dA . \quad (3.42)$$

Therefore, the first Piola-Kirchhoff stress tensor expressed in terms of the Cauchy stress tensor is given by

$$\mathbf{P} = \mathbf{J}\mathbf{T}\mathbf{F}^{-T} . \quad (3.43)$$

The Cauchy stress tensor is symmetric and is defined in the current configuration, whereas the first Piola-Kirchhoff stress tensor is generally not symmetric and is defined in both the reference and current configurations. The symmetry of the Cauchy stress tensor is a result of the conservation of angular momentum, see [132]. Another useful stress tensor, the second Piola-Kirchhoff stress \mathbf{S} , is given as

$$\mathbf{S} = \mathbf{F}^{-1}\mathbf{P} = \mathbf{J}\mathbf{F}^{-1}\mathbf{T}\mathbf{F}^{-T} . \quad (3.44)$$

\mathbf{S} is a symmetric tensor that is defined in the reference configuration.

3.5 Conservation Laws

Balance laws are the fundamental laws that govern the behaviour of all continua. In this section the continuity equation and Cauchy's equation of motion are derived from the principle of mass conservation and the principle of linear momentum conservation.

3.5.1 Conservation of Mass

The principle of mass conservation states that the mass of an arbitrary subdomain Ω stays constant. Therefore the rate of change of mass of the subdomain is zero,

$$\underbrace{\frac{D}{Dt} \int_{\Omega} \rho dv}_{\text{rate of mass change}} = 0, \quad (3.45)$$

where $\frac{D}{Dt}$ denotes the total derivative and ρ the density of the continuum at time t . Using Equation (3.20), the preceding equation may be written with respect to the reference configuration Ω_0 . This allows one to take the derivative of the integrand since the reference configuration is independent of time,

$$\frac{D}{Dt} \int_{\Omega} \rho dv = \frac{D}{Dt} \int_{\Omega_0} \rho J dV = \int_{\Omega_0} \frac{D}{Dt} (\rho J) dV = 0. \quad (3.46)$$

We expand the above expression using the product rule,

$$\begin{aligned} \int_{\Omega_0} \frac{D}{Dt} (\rho J) dV &= \int_{\Omega_0} \left(\frac{D\rho}{Dt} J + \rho \frac{DJ}{Dt} \right) dV \\ &= \int_{\Omega_0} (\dot{\rho} J + \rho J \operatorname{div} \mathbf{v}) dV \\ &= \int_{\Omega_0} (\dot{\rho} + \rho \operatorname{div} \mathbf{v}) J dV \\ &= \int_{\Omega} (\dot{\rho} + \rho \operatorname{div} \mathbf{v}) dv, \end{aligned} \quad (3.47)$$

where we have utilised Equation (3.22). Since the principle of mass conservation is valid for any arbitrary domain Ω , the integrand is zero,

$$\dot{\rho} + \rho \operatorname{div} \mathbf{v} = 0. \quad (3.48)$$

This expression is known as the continuity equation. Using the definition of the material derivative (see Equation (3.24)), the continuity equation may also be given by

$$\frac{\partial \rho}{\partial t} + \operatorname{div}(\rho \mathbf{v}) = 0. \quad (3.49)$$

3.5.2 Conservation of Linear Momentum

The principle of linear momentum conservation states that the rate of change of linear momentum of any subdomain Ω is equal to the forces that act on the subdomain. That is,

$$\underbrace{\frac{D}{Dt} \int_{\Omega} (\rho \mathbf{v}) dv}_{\text{rate of change of linear momentum}} = \underbrace{\int_{\partial\Omega} \mathbf{t} ds}_{\text{surface force}} + \underbrace{\int_{\Omega} \rho \mathbf{b} dv}_{\text{body force}}, \quad (3.50)$$

where $\partial\Omega$ is the surface that bounds the domain Ω .

As in the previous section, the left-hand side may be expressed with respect to the reference configuration,

$$\frac{D}{Dt} \int_{\Omega} (\rho \mathbf{v}) dv = \frac{D}{Dt} \int_{\Omega} (\rho \mathbf{v} J) dV = \int_{\Omega_0} \frac{D}{Dt} (\rho \mathbf{v} J) dV. \quad (3.51)$$

Applying the product rule yields,

$$\int_{\Omega_0} \frac{D}{Dt} (\rho \mathbf{v} J) dV = \int_{\Omega_0} \frac{D\mathbf{v}}{Dt} \rho J + \mathbf{v} \frac{D(\rho J)}{Dt} dV = \int_{\Omega_0} \frac{D\mathbf{v}}{Dt} \rho J + \underbrace{J \mathbf{v} \frac{D(\rho)}{Dt} + \rho \mathbf{v} \frac{D(J)}{Dt}}_{\mathbf{v} \cdot (J \dot{\rho} + \rho J \operatorname{div} \mathbf{v})} dV. \quad (3.52)$$

From the principle of mass conservation, we see that sum of the last two terms in the integrand is zero. Therefore,

$$\int_{\Omega_0} \frac{D}{Dt} (\rho \mathbf{v} J) dV = \int_{\Omega_0} \frac{D\mathbf{v}}{Dt} \rho J dV = \int_{\Omega_0} \mathbf{a} \rho J dV = \int_{\Omega} \mathbf{a} \rho dv, \quad (3.53)$$

where \mathbf{a} denotes the acceleration. Using Equation (3.39) and the divergence theorem, the right-hand side of Equation (3.50) can be restated as,

$$\int_{\partial\Omega} \mathbf{t} ds + \int_{\Omega} \rho \mathbf{b} dv = \int_{\partial\Omega} \boldsymbol{\sigma} \mathbf{n} ds + \int_{\Omega} \rho \mathbf{b} dv = \int_{\Omega} \operatorname{div} \boldsymbol{\sigma} dv + \int_{\Omega} \rho \mathbf{b} dv. \quad (3.54)$$

Substituting Equations (3.53) and (3.54) back into Equation (3.50) leads to

$$\int_{\Omega} (\rho \mathbf{a} - \operatorname{div} \boldsymbol{\sigma} - \rho \mathbf{b}) dv = 0. \quad (3.55)$$

The preceding equation has to be valid for all Ω , hence the integrand is zero. This gives Cauchy's equation of motion,

$$\rho \mathbf{a} - \operatorname{div} \boldsymbol{\sigma} - \rho \mathbf{b} = 0. \quad (3.56)$$

If the body is in static equilibrium, Equation (3.56) becomes

$$\operatorname{div} \boldsymbol{\sigma} + \rho \mathbf{b} = 0, \quad (3.57)$$

which is referred to as the equation of equilibrium.

A boundary condition on a solid is categorised as either a Dirichlet or a Neumann boundary condition. The Dirichlet (or essential) boundary conditions are

$$\mathbf{u}(\mathbf{x}, t) = \bar{\mathbf{u}} \quad \text{on} \quad \partial\mathcal{B}_D \quad (3.58)$$

where $\bar{\mathbf{u}}$ is the prescribed displacement on the Dirichlet boundary \mathcal{B}_D . The Neumann (or natural) boundary condition is given as

$$\boldsymbol{\sigma} \mathbf{n} = \bar{\mathbf{t}} \quad \text{on} \quad \partial\mathcal{B}_N \quad (3.59)$$

where $\bar{\mathbf{t}}$ is the prescribed traction on the Neumann boundary \mathcal{B}_N . Together with the governing equation, the boundary conditions form the so-called *strong form* of the problem definition.

3.6 Hyperelasticity

The kinematic relations, stresses and balance laws stated in the previous sections are valid for any continuum body, regardless of whether it is a solid or fluid. To distinguish one material from another and to characterise its behaviour, material models or constitutive laws are required. A constitutive law should match the real behaviour of the material.

Biological tissue, including passive cardiac tissue, is usually modelled as a hyperelastic material [22]. Hyperelasticity refers to the ability of a material to respond elastically when subjected to large deformations, i.e. the material has the ability to recover its original configuration once the applied load is removed.

3.6.1 Strain Energy Functions

In the context of hyperelasticity, one assumes the existence of a strain energy function ψ which is defined per unit volume,

$$\psi = \psi(\mathbf{F}, \mathbf{x}) , \quad (3.60)$$

where ψ is a scalar-valued function that relates the amount of elastic energy stored to the deformation experienced by a body. If we limit ourselves to homogeneous materials then $\psi = \psi(\mathbf{F})$. We may also express the strain energy as a function of other strain tensors, such as the right Cauchy-Green deformation tensor or the Green-Lagrange strain tensor,

$$\psi(\mathbf{F}) = \psi(\mathbf{E}) = \psi(\mathbf{C}) . \quad (3.61)$$

Expressions for the stress tensors may be derived from the strain energy function with the use of the Clausius-Planck inequality [56]. For perfectly elastic materials without any thermal effects, the Clausius-Planck inequality reduces to

$$W_{int} - \dot{\psi} = 0 , \quad (3.62)$$

where $\dot{\psi}$ is the rate of internal energy and W_{int} is the rate of mechanical energy. The rate of internal energy is expressed with work conjugate pairs using either of the following,

$$W_{int} = \mathbf{P} : \dot{\mathbf{F}} , \quad W_{int} = \mathbf{S} : \dot{\mathbf{E}} , \quad W_{int} = \boldsymbol{\sigma} : \dot{\mathbf{d}} . \quad (3.63)$$

Considering the first conjugate pair, \mathbf{P} and $\dot{\mathbf{F}}$, Equation (3.62) can be restated as

$$\begin{aligned} 0 &= \mathbf{P} : \dot{\mathbf{F}} - \dot{\psi} \\ 0 &= \mathbf{P} : \dot{\mathbf{F}} - \underbrace{\frac{\partial \psi}{\partial \mathbf{F}} : \frac{\partial \mathbf{F}}{\partial t}}_{\dot{\psi}} \\ 0 &= \left(\mathbf{P} - \frac{\partial \psi}{\partial \mathbf{F}} \right) : \dot{\mathbf{F}} . \end{aligned} \quad (3.64)$$

If there are no constraints on the motion then the above expression is valid for all \mathbf{F} , hence

$$\mathbf{P} - \frac{\partial \psi}{\partial \mathbf{F}} = 0 \quad \Rightarrow \quad \mathbf{P} = \frac{\partial \psi}{\partial \mathbf{F}} . \quad (3.65)$$

We use similar approaches to find that,

$$\boldsymbol{\sigma} = \frac{\partial \psi}{\partial \mathbf{d}}, \quad (3.66)$$

$$\mathbf{S} = \frac{\partial \psi}{\partial \mathbf{E}} = 2 \frac{\partial \psi}{\partial \mathbf{C}}. \quad (3.67)$$

In the subsequent sections we restrict ourselves to three types of hyperelastic material models, namely isotropic, transversely isotropic and orthotropic models.

3.6.2 Isotropy

Isotropic materials behave the same in all directions. To ensure the strain energy is invariant under rotation, it is often expressed in terms of the strain invariants of \mathbf{C} (or \mathbf{b}). That is

$$\psi = \psi(I_1^C, I_2^C, I_3^C), \quad (3.68)$$

where I_i^C are the invariants of \mathbf{C} as defined in Equation (3.12). Using the chain rule we find the second-Piola Kirchhoff stress to be

$$\mathbf{S} = 2 \left(\frac{\partial \psi}{\partial I_1^C} \frac{\partial I_1^C}{\partial \mathbf{C}} + \frac{\partial \psi}{\partial I_2^C} \frac{\partial I_2^C}{\partial \mathbf{C}} + \frac{\partial \psi}{\partial I_3^C} \frac{\partial I_3^C}{\partial \mathbf{C}} \right). \quad (3.69)$$

Two common isotropic strain energy functions are the Mooney-Rivlin and the Neo-Hookean material model. For a compressible material, the Mooney-Rivlin model is given as

$$\psi = C_1(J^{-\frac{2}{3}}I_1^C - 3) + C_2(J^{-\frac{4}{3}}I_2^C - 3), \quad (3.70)$$

where C_1 and C_2 are material parameters determined from material testing. Setting $C_2 = 0$, we obtain the Neo-Hookean material model,

$$\psi = C_1(J^{-\frac{2}{3}}I_1^C - 3). \quad (3.71)$$

Both the Mooney-Rivlin and Neo-Hookean constitutive laws are often used to model the behaviour of rubber, see for example [103] and [89].

3.6.3 Transverse Isotropy

If a material has a distinct response in one direction it is termed transversely isotropic. As shown in Figure 3.3(a), the preferred direction is normal to the plane of isotropy, the plane in which the material behaves isotropic. Apart from the three invariants (I_1^C , I_2^C and I_3^C) it is necessary to compose additional invariants to take into account the preferred material direction \mathbf{f} . In this way the preferred direction can be included in the strain energy function, i.e. $\psi(\mathbf{C}, \mathbf{f})$. This is accomplished with a fourth and fifth invariant,

$$I_4^C = \mathbf{f} \cdot \mathbf{C} \mathbf{f} = \mathbf{C} : \mathbf{f} \otimes \mathbf{f}, \quad (3.72)$$

$$I_5^C = \mathbf{f} \cdot \mathbf{C}^2 \mathbf{f} = \mathbf{C}^2 : \mathbf{f} \otimes \mathbf{f}. \quad (3.73)$$

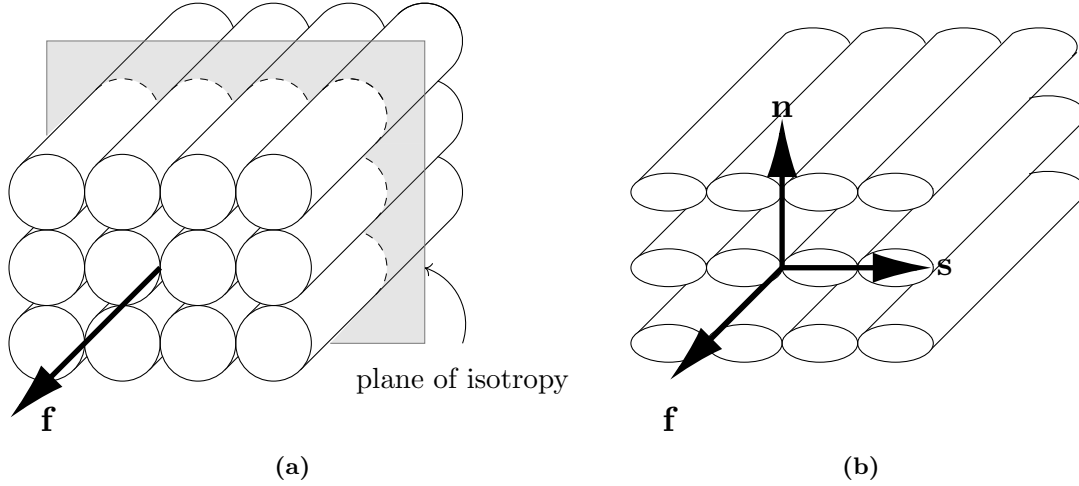


Figure 3.3: Schematic illustration of (a) a transversely isotropic material with one preferred material direction and (b) an orthotropic material with three orthogonal material directions.

The strain energy for a transversely isotropic material is often stated in the form

$$\psi = \psi_{iso}(I_1^C, I_2^C, I_3^C) + \psi_f(I_4^C, I_5^C), \quad (3.74)$$

where ψ_{iso} is the isotropic component and ψ_f is the transversely isotropic contribution to the strain energy. A simple example of a transversely isotropic material is

$$W = C_1(I_1^C - 3) + C_2(I_2^C - 3) + C_4(I_4^{C\frac{1}{2}} - 1)^2. \quad (3.75)$$

The above model was used by Brown and Smith [133] to model the behaviour of anisotropic elastomers reinforced with fibres. The preferred material direction corresponded with the reinforced fibre direction.

3.6.4 Orthotropy

An orthotropic material has three mutually orthogonal directions \mathbf{f} , \mathbf{s} and \mathbf{n} with different behaviour in each, see Figure 3.3(b). As with the transversely isotropic model, these orthogonal directions need to be included in the strain energy function. To account for orthotropy, we define additional invariants. For example,

$$I_4^C = \mathbf{f} \cdot \mathbf{C} \mathbf{f} = \mathbf{C} : \mathbf{f} \otimes \mathbf{f}, \quad (3.76)$$

$$I_5^C = \mathbf{f} \cdot \mathbf{C}^2 \mathbf{f} = \mathbf{C}^2 : \mathbf{f} \otimes \mathbf{f}, \quad (3.77)$$

$$I_6^C = \mathbf{n} \cdot \mathbf{C} \mathbf{n} = \mathbf{C} : \mathbf{n} \otimes \mathbf{n}, \quad (3.78)$$

$$I_7^C = \mathbf{n} \cdot \mathbf{C}^2 \mathbf{n} = \mathbf{C}^2 : \mathbf{n} \otimes \mathbf{n}. \quad (3.79)$$

One can also include coupling invariants [58] such as

$$I_8^C = \mathbf{f} \cdot \mathbf{C} \mathbf{n}. \quad (3.80)$$

An example of an orthotropic strain energy function is given below. The constitutive model was used by Holzapfel et al. [57] to describe the orthotropic mechanical response of arterial tissue.

$$\psi = \frac{c}{2}(\bar{I}_1 - 3) + \frac{k_1}{2k_2}(e^{k_2(\bar{I}_4 - 1)^2} - 1) + \frac{k_1}{2k_2}(e^{k_2(\bar{I}_6 - 1)^2} - 1). \quad (3.81)$$

The invariants in Equation (3.81) are defined as

$$\bar{I}_1 = \text{tr} \bar{\mathbf{C}}, \quad \bar{I}_4 = \bar{\mathbf{C}} : \mathbf{a}_{01} \otimes \mathbf{a}_{01}, \quad \bar{I}_6 = \bar{\mathbf{C}} : \mathbf{a}_{02} \otimes \mathbf{a}_{02}, \quad (3.82)$$

where $\bar{\mathbf{C}}$ is the distortional part of the right Cauchy-Green tensor and \mathbf{a}_{0i} are the directions that characterise the layout of the collagenous fibres in the arteries.

3.6.5 Incompressibility

A number of materials experience deformation without a significant volume change. Such materials may be idealised as incompressible materials. Since no volume change occurs we have that $\det \mathbf{F} = J = 1$. This constraint is enforced using the well-known method of Lagrange multipliers with the hydrostatic pressure p serving as the Lagrange multiplier [56]. The strain energy function for incompressible materials is given by

$$\psi = \psi(\mathbf{F}) - p(J - 1). \quad (3.83)$$

Furthermore, by differentiating the strain energy function in Equation (3.83) with respect to the deformation gradient we find that,

$$\mathbf{P} = \frac{\partial \psi(\mathbf{F})}{\partial \mathbf{F}} - p\mathbf{F}^{-T}. \quad (3.84)$$

The expressions for the Cauchy stress $\boldsymbol{\sigma}$ and the Second Piola-Kirchhoff stress \mathbf{S} for incompressible materials therefore reads (see Equations (3.43) and (3.44)),

$$\boldsymbol{\sigma} = \frac{\partial \psi(\mathbf{F})}{\partial \mathbf{F}} \mathbf{F}^T - p\mathbf{I}, \quad \mathbf{S} = 2 \frac{\partial \psi(\mathbf{C})}{\partial \mathbf{C}} - p\mathbf{C}^{-1}. \quad (3.85)$$

3.7 The Finite Element Method for Non-linear Elasticity

When dealing with complex geometries, loading conditions or material behaviour it is not possible to analytically solve the governing equations [76]. The finite element method is a numerical method used to find approximate solutions to partial differential equations, such as Cauchy's equation of motion or the equation of equilibrium. This section discusses the general finite element framework for non-linear elastic problems, such as those related to hyperelastic materials.

The governing equation derived in Section 3.5 and the boundary conditions for a solid body \mathcal{B} are repeated below,

$$\text{div} \boldsymbol{\sigma} + \rho \mathbf{b} = \rho \mathbf{a} \quad (3.86)$$

$$\mathbf{u} = \bar{\mathbf{u}} \quad \text{on} \quad \partial \mathcal{B}_D \quad (3.87)$$

$$\boldsymbol{\sigma} \mathbf{n} = \bar{\mathbf{t}} \quad \text{on} \quad \partial \mathcal{B}_N \quad (3.88)$$

The finite element method for non-linear problems generally consists of (i) casting the governing equation into a weak (variational) form, (ii) discretising the weak formulation by approximating field variables with shape functions, (iii) linearising the weak form using a Taylor expansion and (iv) solving the system with an iterative scheme such as a Newton-Raphson algorithm.

3.7.1 Principle of Virtual Work

The finite element formulation is based on the weak form of the governing equation. The weak form in solid mechanics is also referred to as the principle of virtual work. To obtain the weak form, we first consider a body subjected to an arbitrary displacement $\delta \mathbf{u}$. Equation (3.86) is multiplied by $\delta \mathbf{u}$ and integrated over the continuum domain \mathcal{B} ,

$$\int_{\mathcal{B}} (\rho \mathbf{a} - \text{div} \boldsymbol{\sigma} - \rho \mathbf{b}) \cdot \delta \mathbf{u} dV = 0, \quad (3.89)$$

where $\delta \mathbf{u}$ is also referred to as the virtual displacement. To ensure the boundary conditions are satisfied, $\delta \mathbf{u}$ vanishes on the Dirichlet boundary \mathcal{B}_D . The second term of the integrand in Equation (3.89) may be expanded with the identity, $\text{div} \boldsymbol{\sigma} \cdot \mathbf{a} = \text{div}(\boldsymbol{\sigma} \mathbf{a}) - \boldsymbol{\sigma} : \text{grad} \mathbf{a}$. Equation (3.89) can therefore be written as

$$\int_{\mathcal{B}} (\rho \mathbf{a} \cdot \delta \mathbf{u} - \text{div}(\boldsymbol{\sigma} \delta \mathbf{u}) + \boldsymbol{\sigma} : \text{grad} \delta \mathbf{u} - \rho \mathbf{b} \cdot \delta \mathbf{u}) dV = 0. \quad (3.90)$$

Applying the divergence theorem we obtain the weak form of the governing equation as

$$\int_{\mathcal{B}} -\boldsymbol{\sigma} : \text{grad} \delta \mathbf{u} dv + \int_{\partial \mathcal{B}} \mathbf{T} \delta \mathbf{u} \cdot \mathbf{n} ds + \int_{\mathcal{B}} \rho \mathbf{b} \cdot \delta \mathbf{u} dv = \int_{\mathcal{B}} \rho \mathbf{a} \cdot \delta \mathbf{u} dv. \quad (3.91)$$

If we consider the static case, then Equation (3.91) reduces to the weak form of the equilibrium equation,

$$\int_{\mathcal{B}} -\boldsymbol{\sigma} : \text{grad} \delta \mathbf{u} dv + \int_{\partial \mathcal{B}} \mathbf{T} \delta \mathbf{u} \cdot \mathbf{n} ds + \int_{\mathcal{B}} \rho \mathbf{b} \cdot \delta \mathbf{u} dv = 0. \quad (3.92)$$

The above can also be expressed in the reference configuration using Equations (3.20) and (3.44),

$$\int_{\mathcal{B}_0} \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV - \int_{\partial \mathcal{B}_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\mathcal{B}} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV = 0. \quad (3.93)$$

To rewrite the above in terms of the Green-Lagrange strain tensor, the first term is expanded using the symmetric and skew-symmetric parts:

$$\int_{\mathcal{B}_0} \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV = \int_{\mathcal{B}_0} \mathbf{S} : ((\mathbf{F}^T \text{Grad} \delta \mathbf{u})^{\text{symm}} + (\mathbf{F}^T \text{Grad} \delta \mathbf{u})^{\text{skew}}) dV. \quad (3.94)$$

Since the double dot product of a symmetric tensor and skew-symmetric tensor is zero we have

$$\begin{aligned} \int_{\mathcal{B}_0} \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV &= \int_{\mathcal{B}_0} \mathbf{S} : ((\mathbf{F}^T \text{Grad} \delta \mathbf{u})^{\text{symm}}) dV \\ &= \int_{\mathcal{B}_0} \mathbf{S} : \frac{1}{2} ((\mathbf{F}^T \text{Grad} \delta \mathbf{u})^T + (\mathbf{F}^T \text{Grad} \delta \mathbf{u})) dV. \end{aligned} \quad (3.95)$$

Using Equation (3.34), the variation of the strain tensor is given as

$$\delta \mathbf{E} = \delta \frac{1}{2} (\mathbf{F}^T \mathbf{F} - \mathbf{I}) = \frac{1}{2} (\mathbf{F}^T \text{Grad} \delta \mathbf{u} + \text{Grad}^T \delta \mathbf{u} \mathbf{F}) . \quad (3.96)$$

Therefore, the weak form of the equilibrium equation in terms of the Green-Lagrange strain tensor reads

$$\int_{\mathcal{B}_0} \mathbf{S} : \delta \mathbf{E} dV - \int_{\partial \mathcal{B}_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\mathcal{B}_0} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV = 0 . \quad (3.97)$$

Both of the weak forms stated in Equations (3.93) and (3.97) are often used to obtain the finite element stiffness matrix and the finite element force vectors for continua in static equilibrium.

3.7.2 Discretisation

The geometry of the body \mathcal{B} is discretised into a finite number of elements, as illustrated in the Figure 3.4. We divide the domain \mathcal{B} into n_e elements, such that

$$\mathcal{B} \approx \mathcal{B}^h = \bigcup_{e=1}^{n_e} \Omega_e, \quad (3.98)$$

where \mathcal{B}^h is the approximated body and Ω_e is the e^{th} element. The chosen element type depends on the problem at hand. Figure 3.5 shows examples of different element types. We not only approximate the geometry but also the primary field variables, in this case the displacement. The exact displacement solution is approximated using interpolation functions,

$$\mathbf{u}_{exact} \approx \sum_{I=1}^n N_I(\mathbf{X}) \mathbf{u}_I, \quad (3.99)$$

where \mathbf{u}_I is the unknown displacements at the nodes of an element with n nodes and N_I are the shape functions. Similarly we approximate the gradient of the displacement as

$$\text{Grad} \mathbf{u}_{exact} = \mathbf{u}_{exact} \otimes \nabla_X \approx \sum_{I=1}^n \mathbf{u}_I \otimes \frac{\partial N_I(\mathbf{X})}{\partial \mathbf{X}} . \quad (3.100)$$

In the Galerkin formulation the virtual displacement is approximated using the same shape functions, that is

$$\delta \mathbf{u}_{exact} \approx \sum_{I=1}^n N_I(\mathbf{X}) \delta \mathbf{u}_I, \quad \text{Grad} \delta \mathbf{u}_{exact} \approx \sum_{I=1}^n \delta \mathbf{u}_I \otimes \frac{\partial N_I(\mathbf{X})}{\partial \mathbf{X}} . \quad (3.101)$$

Discretising the weak form in Equation (3.93) with the approximations provided in Equations (3.98) through (3.101) leads to,

$$\bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \mathbf{F} \mathbf{S} : \left(\delta \mathbf{u}_I \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) d\Omega - \bigcup_{r=1}^{n_r} \sum_{I=1}^m \int_{\Gamma_r} \mathbf{t} \cdot N_I \delta \mathbf{u}_I d\Gamma - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \rho_0 \mathbf{b} \cdot N_I \delta \mathbf{u}_I d\Omega = 0, \quad (3.102)$$

where n_r is the number of element boundaries with traction loads and m is the number of nodes on the traction surface Γ_r . Note that we use \bigcup to denote an assembly process.

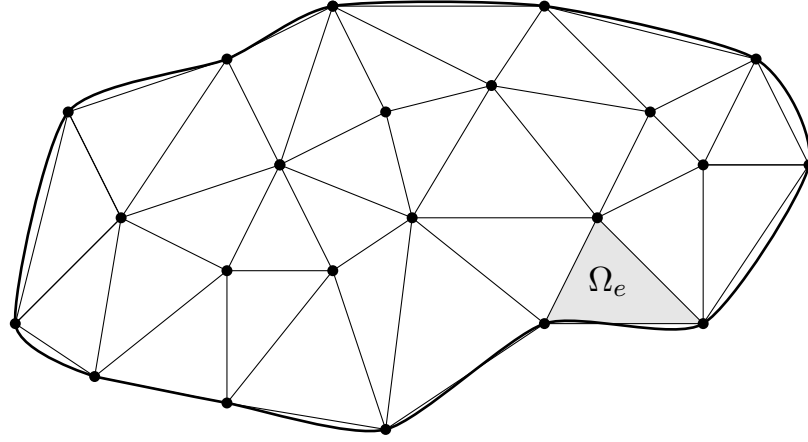


Figure 3.4: Discretisation of a continuum body into a finite number of elements.

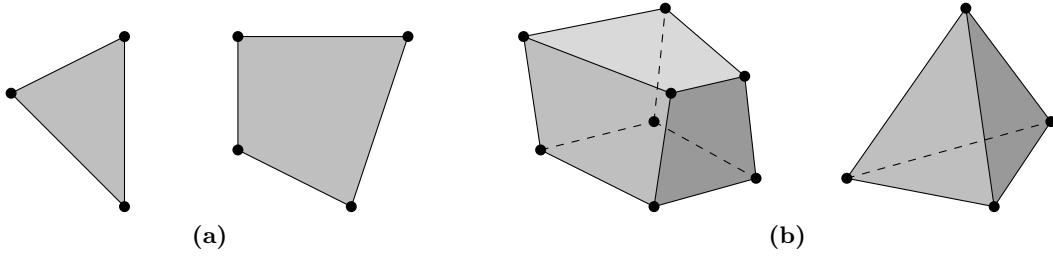


Figure 3.5: Examples of finite elements used to discretise the problem domain include (a) elements for two dimensional problems, such as triangular and quadrilateral elements and (b) elements for three-dimensional problems, such as hexahedral and tetrahedral elements.

Using the following rules $\mathbf{a} \cdot \mathbf{b} = \mathbf{b}^T \mathbf{a}$ and $\mathbf{T} : \mathbf{a} \otimes \mathbf{b} = \mathbf{a} \cdot \mathbf{T} \mathbf{b}$, we can rewrite the discretised weak form in Equation (3.102) as

$$\bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \mathbf{F} \mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} d\Omega - \bigcup_{r=1}^{n_r} \sum_{I=1}^m \delta \mathbf{u}_I^T \int_{\Gamma_r} N_I \mathbf{t} d\Gamma - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0. \quad (3.103)$$

Because the virtual displacement is arbitrary we have

$$\bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \mathbf{F} \mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} d\Omega - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Gamma_e} N_I \mathbf{t} d\Gamma - \bigcup_{r=1}^{r_e} \sum_{I=1}^n \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0. \quad (3.104)$$

The discretised weak formulation in Equation (3.104) is a set of non-linear algebraic equations that need to be solved using an iterative scheme.

3.7.3 Linearisation of the Weak Formulation

Finite deformations (large strains) and nonlinear material behaviour introduce nonlinearities into the system. We therefore linearise the weak form and use an iterative method, such as the Newton-

Raphson scheme, to find an approximate solution to the displacement field. First, let us define the residual G^1 as

$$G(\mathbf{u}) = \int_{\mathcal{B}_0} \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV - \int_{\partial \mathcal{B}_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\mathcal{B}_0} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV. \quad (3.105)$$

The objective is to find \mathbf{u} to minimise the residual. To find a solution that minimises the above, we consider a first-order Taylor expansion of the residual,

$$G(\mathbf{u}_{k+1}) = G(\mathbf{u}_k) + \frac{\partial G(\mathbf{u}_k)}{\partial \mathbf{u}} \cdot \Delta \mathbf{u}_k = G(\mathbf{u}_k) + \Delta G_k, \quad (3.106)$$

where $\Delta \mathbf{u}$ is the displacement increment, k is the current iteration step and ΔG is the variation in the residual. The updated or new displacement is the sum of the current displacement and the displacement increment, that is

$$\mathbf{u}_{k+1} = \mathbf{u}_k + \Delta \mathbf{u}_k. \quad (3.107)$$

If we assume that the traction and body forces are conservative, i.e. do not depend on the displacement, then the last two terms in $G(\mathbf{u})$ are independent of $\Delta \mathbf{u}$ and the variation in the residual is given as

$$\Delta G = \frac{\partial G(\mathbf{u})}{\partial \mathbf{u}} \cdot \Delta \mathbf{u} = \int_{\mathcal{B}_0} \Delta \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV + \int_{\mathcal{B}_0} \mathbf{S} : \Delta \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV, \quad (3.108)$$

where $\Delta \mathbf{F} = \frac{\partial \mathbf{F}}{\partial \mathbf{u}} \Delta \mathbf{u}$ and $\Delta \mathbf{S} = \frac{\partial \mathbf{S}}{\partial \mathbf{u}} \Delta \mathbf{u}$ are the variations in the deformation gradient and the Second Piola-Kirchhoff stress. We write the first term of the variation in the residual as

$$\begin{aligned} \int_{\mathcal{B}_0} \Delta \mathbf{S} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV &= \int_{\mathcal{B}_0} \mathbf{C} \Delta \mathbf{E} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV \\ &= \int_{\mathcal{B}_0} \mathbf{C} \mathbf{F}^T \text{Grad} \Delta \mathbf{u} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV \end{aligned} \quad (3.109)$$

where $\mathbf{C} = \frac{\partial \mathbf{S}}{\partial \mathbf{E}}$ is the fourth-order elasticity tensor. The elasticity tensor is a measure of the change in stress which occurs as a result of the change in strain. It has both major and minor symmetries. To obtain the second line in the above we used the definition of the strain tensor \mathbf{E} and exploited the symmetry of \mathbf{C} . Discretising the above equation we find

$$\int_{\mathcal{B}_0} \mathbf{C} \mathbf{F}^T \text{Grad} \Delta \mathbf{u} : \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV \approx \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \mathbf{C} \mathbf{F}^T \left(\Delta \mathbf{u}_J \otimes \frac{\partial N_J}{\partial \mathbf{X}} \right) : \mathbf{F}^T \left(\delta \mathbf{u}_I \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) d\Omega. \quad (3.110)$$

To rearrange the above, it is useful to express the integrand in index notation².

$$\mathbb{C}_{ijkl} F_{sk} \Delta u_{Js} \frac{\partial N_J}{\partial X_l} F_{ri} \delta u_{Ir} \frac{\partial N_I}{\partial X_j} = \delta u_{Ir} F_{ri} \frac{\partial N_I}{\partial X_j} \mathbb{C}_{ijkl} \frac{\partial N_J}{\partial X_l} F_{sk} \Delta u_{Js} \quad (3.111)$$

¹The residual can also be defined as $G(\mathbf{u}) = \int_{\mathcal{B}_0} \mathbf{S} : \delta \mathbf{E} dV - \int_{\partial \mathcal{B}_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\mathcal{B}_0} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV$

²This is because the components of tensors are scalars and therefore the operation is commutative.

The index notation in the previous expression allows us to rewrite Equation (3.110) as

$$\begin{aligned} \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \mathbf{C} \mathbf{F}^T \left(\Delta \mathbf{u}_J \otimes \frac{\partial N_J}{\partial \mathbf{X}} \right) : \mathbf{F}^T \left(\delta \mathbf{u}_I \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) d\Omega \\ = \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \left(\mathbf{F} \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) \mathbf{C} \left(\frac{\partial N_J}{\partial \mathbf{X}} \otimes \mathbf{F}^T \right) d\Omega \Delta \mathbf{u}_J, \end{aligned} \quad (3.112)$$

where $(\mathbf{F} \otimes \frac{\partial N_I}{\partial \mathbf{X}})$ is a third-order tensor.

The second term in Equation (3.108) we write as

$$\begin{aligned} \int_{\mathcal{B}_0} \mathbf{S} : \Delta \mathbf{F}^T \text{Grad} \delta \mathbf{u} dV &= \int_{\mathcal{B}_0} \mathbf{S} : (\text{Grad}^T \Delta \mathbf{u}) \text{Grad} \delta \mathbf{u} dV \\ &= \int_{\mathcal{B}_0} (\text{Grad} \Delta \mathbf{u}) \mathbf{S} : \text{Grad} \delta \mathbf{u} dV. \end{aligned} \quad (3.113)$$

Discretising the domain and the displacement we obtain,

$$\begin{aligned} \int_{\mathcal{B}_0} (\text{Grad} \Delta \mathbf{u}) \mathbf{S} : \text{Grad} \delta \mathbf{u} dV &\approx \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \left(\Delta \mathbf{u} \otimes \frac{\partial N_J}{\partial \mathbf{X}} \right) \mathbf{S} : \left(\delta \mathbf{u} \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) d\Omega \\ &\approx \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \delta \mathbf{u}_I^T \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \Delta \mathbf{u}_J d\Omega \end{aligned} \quad (3.114)$$

where we have utilised the rule $\mathbf{A} : \mathbf{a} \otimes \mathbf{b} = \mathbf{a} \cdot \mathbf{A} \mathbf{b}$ and applied the rule $(\mathbf{a} \otimes \mathbf{b}) \mathbf{c} = (\mathbf{c} \cdot \mathbf{b}) \mathbf{a}$, see Equations (A.17) and (A.18) in Appendix A for more detail. Substituting Equations (3.112) and (3.114) into Equation (3.108) we obtain the variation in the residual as,

$$\Delta G = \delta \mathbf{u}_I^T \left\{ \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \left(\mathbf{F} \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) \mathbf{C} \left(\frac{\partial N_J}{\partial \mathbf{X}} \otimes \mathbf{F}^T \right) + \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \mathbf{I} d\Omega \right\} \Delta \mathbf{u}_J. \quad (3.115)$$

Furthermore, we know from Equation (3.103) that

$$G(\mathbf{u}_k) = \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \mathbf{F} \mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} d\Omega - \bigcup_{r=1}^{n_r} \sum_{I=1}^m \delta \mathbf{u}_I^T \int_{\Gamma_r} N_I \mathbf{t} d\Gamma - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0. \quad (3.116)$$

Finally, combining the above two equations we obtain the final expression for the linearised residual as

$$\begin{aligned} G(\mathbf{u}_{k+1}) &= \underbrace{\delta \mathbf{u}_I^T \left\{ \bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \mathbf{F} \mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} d\Omega \right\}}_{\mathbf{F}_{int}} - \underbrace{\delta \mathbf{u}_I^T \left\{ \bigcup_{r=1}^{n_r} \sum_{I=1}^m \int_{\Gamma_r} N_I \mathbf{t} d\Gamma + \bigcup_{n=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega \right\}}_{\mathbf{F}_{ext}} \\ &+ \underbrace{\delta \mathbf{u}_I^T \left\{ \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \left(\mathbf{F} \otimes \frac{\partial N_I}{\partial \mathbf{X}} \right) \mathbf{C} \left(\frac{\partial N_J}{\partial \mathbf{X}} \otimes \mathbf{F}^T \right) + \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \mathbf{I} d\Omega \right\}}_{\mathbf{K}} \Delta \mathbf{u}_J. \end{aligned} \quad (3.117)$$

In Equation (3.117) we have indicated the internal force vector \mathbf{F}_{int} , the external force vector \mathbf{F}_{ext} and the stiffness matrix \mathbf{K} . It is important to note that it is often more convenient for computational purposes to utilise Voigt notation. The Voigt forms of the stiffness matrix and force vectors are given in Appendix A,

In terms of the stiffness matrix and the force vectors the residual is given as

$$\mathbf{G}(\mathbf{u}_{k+1}) = \delta \mathbf{u}_I^T \left(\mathbf{F}(\mathbf{u}_k)_{ext} - \mathbf{F}(\mathbf{u}_k)_{int} + \mathbf{K}(\mathbf{u}_k) \Delta \mathbf{u}_J \right). \quad (3.118)$$

3.7.4 Newton-Raphson Algorithm

Consider the linearised residual in Equation (3.118). If we assume a solution at \mathbf{u}_{k+1} (i.e. $\mathbf{G}_{k+1} = 0$) we find that,

$$0 = \delta \mathbf{u}_I^T \left(\mathbf{F}_{ext} - \mathbf{F}_{int} + \mathbf{K} \Delta \mathbf{u}_J \right). \quad (3.119)$$

Since the virtual displacement is arbitrary, it allows us to write the following,

$$\mathbf{0} = \left(\mathbf{F}_{ext} - \mathbf{F}_{int} + \mathbf{K} \Delta \mathbf{u}_J \right). \quad (3.120)$$

And we find the displacement increment as

$$\Delta \mathbf{u}_J = -[\mathbf{K}]^{-1} (\mathbf{F}_{ext} - \mathbf{F}_{int}), \quad (3.121)$$

where the force vectors and the stiffness matrix are as defined in Equation (3.117). The displacement is then updated according to

$$\mathbf{u}_{k+1} = \mathbf{u}_k + \Delta \mathbf{u}_k. \quad (3.122)$$

Once the solution is updated, a new stiffness matrix and force vectors can be calculated using the new displacement. Equations (3.121) and (3.122) are then iteratively repeated until the displacement increment is sufficiently small. This iterative process is known as the Newton-Raphson method.

We can summarise the basic approach for the Newton-Raphson method as follows,

1. Assume a solution at \mathbf{u}_{k+1} , leading to $\mathbf{G}(\mathbf{u}_{k+1}) = 0$.
2. Calculate the stiffness matrix $\mathbf{K}(\mathbf{u}_k)$ and the force vectors, $\mathbf{F}(\mathbf{u}_k)_{ext}$ and $\mathbf{F}(\mathbf{u}_k)_{int}$, for the current iteration step using the expressions in Equation (3.117).
3. Calculate the displacement increment with Equation (3.121).
4. Update the displacement with Equation (3.122).
5. Check for convergence:
 - (a) if $\|\Delta \mathbf{u}_k\| < \text{specified tolerance}$, terminate iteration.
 - (b) if $\|\Delta \mathbf{u}_k\| > \text{specified tolerance}$, repeat steps 1 through 5 for new values of \mathbf{u} .

3.8 Brief Introduction to Curvilinear Coordinates

In previous sections we only dealt with Cartesian tensors, i.e. tensors that are defined in a coordinate system with base vectors \mathbf{e}_i that are unit vectors and mutually orthogonal. However, in some cases it might be advantageous to utilise more general bases, also referred to as curvilinear coordinate systems. Curvilinear bases are often used when the problem domain is of a specific shape, such as a domain that resembles a sphere or cylinder [67].

3.8.1 Base Vectors in a General Curvilinear Coordinate System

Figure 3.6 compares the coordinate curves for a Cartesian coordinate system and a curvilinear coordinate system in the two-dimensional space. As shown, the coordinate curves in a curvilinear system need not necessarily be straight or orthogonal as in the case of a Cartesian system.

Let Θ^i be the curvilinear coordinates where $i = 1, 2, 3$ for the three-dimensional space. The curvilinear base vectors \mathbf{g}_i are tangent to the coordinate curves as illustrated in Figure 3.7. That is

$$\mathbf{g}_i = \frac{\partial \mathbf{x}(\Theta^i)}{\partial \Theta^i}, \quad (3.123)$$

where \mathbf{g}_i are the covariant base vectors, denoted with a subscript. The base vectors are non-parallel and are in general neither mutually orthogonal nor of unit length [56]. Furthermore, we introduce \mathbf{g}^i , the dual (or reciprocal) base of \mathbf{g}_j , such that

$$\mathbf{g}^i \cdot \mathbf{g}_j = \delta_j^i = \begin{cases} 1 & \text{if } i = j \\ 0 & \text{if } i \neq j \end{cases}, \quad (3.124)$$

where δ_j^i is the kronecker delta and \mathbf{g}^i are known as the contravariant base vectors.

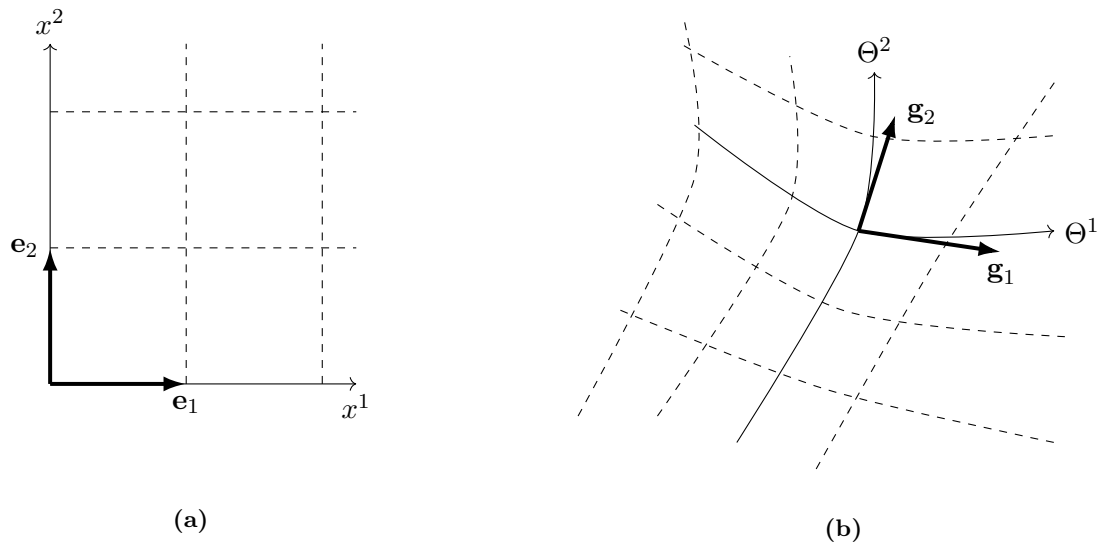


Figure 3.6: Difference between the coordinate curves of (a) a Cartesian coordinate system and (b) a curvilinear coordinate system, adapted from [67].

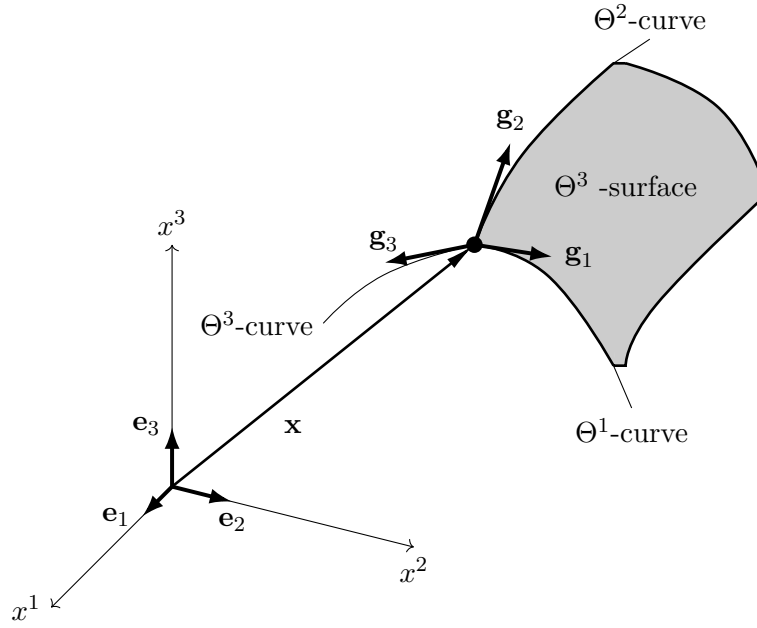


Figure 3.7: The coordinate curve is generated by varying Θ^1 while keeping Θ^2 and Θ^3 constant, modified from [67].

Since the base vectors of a curvilinear system are not necessarily perpendicular, the scalar product of two base vectors is not automatically zero or one. The metric coefficients g_{ij} and g^{ij} describe the geometric properties of the particular basis and is defined as

$$g_{ij} = \mathbf{g}_i \cdot \mathbf{g}_j, \quad g^{ij} = \mathbf{g}^i \cdot \mathbf{g}^j. \quad (3.125)$$

When the indices are the same $i = j$, we obtain the length of the base vector,

$$g_{ii} = |\mathbf{g}_i|^2, \quad g^{ii} = |\mathbf{g}^i|^2. \quad (3.126)$$

For different indices, the metric coefficient represents the angle between the base vectors,

$$g_{ij} = |\mathbf{g}_i| |\mathbf{g}_j| \cos \theta. \quad (3.127)$$

3.8.2 Contravariant and Covariant Components of Vectors and Tensors

A vector \mathbf{a} in the curvilinear space is given as

$$\mathbf{a} = a^k \mathbf{g}_k, \quad (3.128)$$

where a^k are the contravariant components of \mathbf{a} , denoted with a superscript. A vector can also be represented using the contravariant base vectors as

$$\mathbf{a} = a_k \mathbf{g}^k, \quad (3.129)$$

where a_k are the covariant components of \mathbf{a} , denoted with a subscript.

Using Equations (3.124) and (3.125) , the magnitude of \mathbf{a} is given as

$$|\mathbf{a}| = (\mathbf{a} \cdot \mathbf{a})^{\frac{1}{2}} = (g^{ij} a_i a_j)^{\frac{1}{2}} = (g_{ij} a^i a^j)^{\frac{1}{2}} = (a^i a_i)^{\frac{1}{2}} . \quad (3.130)$$

A second-order tensor \mathbf{A} in the curvilinear space is represented as

$$\mathbf{A} = A_{ij} \mathbf{g}^i \otimes \mathbf{g}^j = A^{ij} \mathbf{g}_i \otimes \mathbf{g}_j = A_j^i \mathbf{g}_i \otimes \mathbf{g}^j = A_i^j \mathbf{g}^i \otimes \mathbf{g}_j , \quad (3.131)$$

where A_{ij} , A^{ij} , A_j^i and A_i^j are respectively covariant, contravariant, right-covariant mixed and left-covariant mixed components of \mathbf{A} .

If the base vectors are unit vectors and orthogonal, as in a Cartesian system, then \mathbf{g}_i and \mathbf{g}^i coincide and therefore $\mathbf{g}_i = \mathbf{g}^i$. Cylindrical and spherical coordinate systems are also special cases of curvilinear systems where the base vectors are orthogonal.

3.9 Summary of Continuum Mechanics

In this chapter we introduced the concept of a tensor which was defined as a linear map between two other tensors. The kinematics of a continuum body was reviewed and we presented several tensors commonly used to measure the stress and deformation experienced by a continuum body. We also derived the continuity equation and the equation of motion from the balance of mass and the balance of linear momentum.

We noted that biological tissue is typically modelled as hyperelastic materials. To this end we discussed strain energy functions and presented several examples of hyperelastic material models. These are typically formulated in terms of the scalar invariants of the strain tensor. Additionally, we described the finite element method that is used to obtain approximate solutions to the governing equation. Finally, we gave a brief overview of curvilinear coordinate systems.

Chapter 4

Computational Cardiac Modelling

Computational cardiac modelling concerns the development of computed-based heart models that simulate biological features of the heart. The literature is abundant with research dedicated to computational cardiac models. Each cardiac model has a specific purpose, such as investigating the electro-physiology, analysing the mechanical behaviour or simulating the electro-mechanical coupling in cardiac tissue. The present work focuses on the bio-mechanical behaviour of the heart. This chapter is therefore aimed at providing a literature review of the main components required to create a bio-mechanical model of the heart. The reader is referred to Perez [77] who compiled a list of cardiac models developed over a time period spanning nearly half a century. Apart from discussing a variety of applications of cardiac models, Perez also provides a general overview of whole heart modelling.

The mechanical behaviour of cardiac tissue is often separated into two parts, a passive and an active component. The passive component is related to the elastic behaviour of cardiac tissue, while the active response is a result of the ability of cardiac tissue to generate its own tension when activated. This chapter presents a number of material models from the literature that are used to model passive cardiac tissue. In addition to these material models, methods to simulate the active response are also discussed.

During the last few decades the heart geometry has been approached in a number of ways. Here we highlight some of these approaches that are related to left ventricular and bi-ventricular models. Finally, this chapter concludes with a discussion on the Windkessel model, a popular lumped-parameter model that allows one to model the interaction between the ventricles and the arterial system.

4.1 Modelling the Passive Behaviour of Cardiac Tissue

During diastole, the heart behaves purely passively, i.e cardiac muscles fibres are in a state of rest and are not electrically activated. A classical continuum model is the most popular choice when it comes to modelling the elastic behaviour of biological tissue. In this section, we first consider some experimental findings related to cardiac tissue and expand on the discussions provided in Section 2.4. Secondly, we provide a number of classical continuum material models for passive heart behaviour.

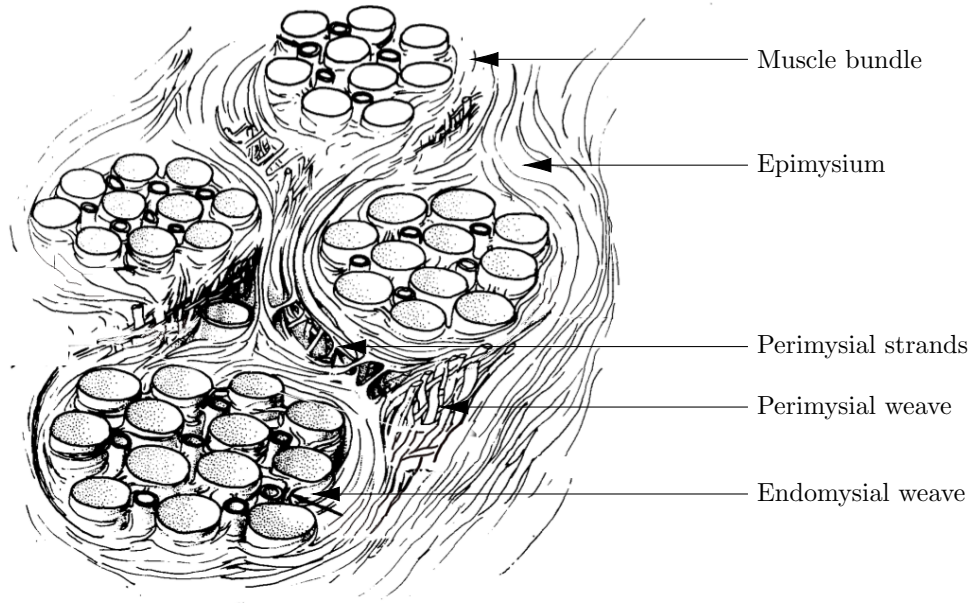


Figure 4.1: The endomysium, perimesium and epimysium form part of of the extracellular matrix that surround cardiac muscle tissue, adapted from [29].

4.1.1 Structure and Experimental Findings

To develop an appropriate constitutive model or to evaluate an existing model, one has to understand the structure of the myocardium. Two thirds of the myocardial volume consist of cardiac muscle tissue. The remaining one third is occupied by other cellular components, the extracellular matrix and also tissue fluid [19, 104].

Figure 4.1 shows bundles of cardiac tissue surrounded by the extracellular matrix. The epimysium, perimysium and endomysium are the main components of the extracellular matrix and contribute to the overall mechanical behaviour of the myocardium [29]. The endomysium surrounds individual heart muscle fibres and laterally joins adjacent fibres. The perimysium is composed of weaves of collagen fibres and surrounds groups of cardiac fibres. Larger structures, also part of the perimysium, join groups of muscle cells. The endocardium and epicardium of the heart wall are surrounded by epimysial collagen fibres, which protect the cardiac fibres from being overstretched and thereby adding to the overall stiffness of the myocardium [29]. For a constitutive law to be a good representation of cardiac muscle tissue, it must feature both the muscle fibres and the extracellular matrix.

Prior to the study by Demer and Yin [26], material models were mainly based on uniaxial material test data. Because of the environment in which the heart operates, cardiac tissue is subjected to multiaxial loading. Therefore, measured data from uniaxial studies are inadequate and cannot be extended to describe the behaviour of the intact heart [26].

Demer and Yin [26] and Yin et al. [23] studied the mechanical properties of the heart by subjecting sheets of cardiac tissue to loading in the fibre and the cross-fibre direction, i.e. biaxial loading. Figure 4.2 provides the stress-strain curve for one of the specimens from [26]. The results show that

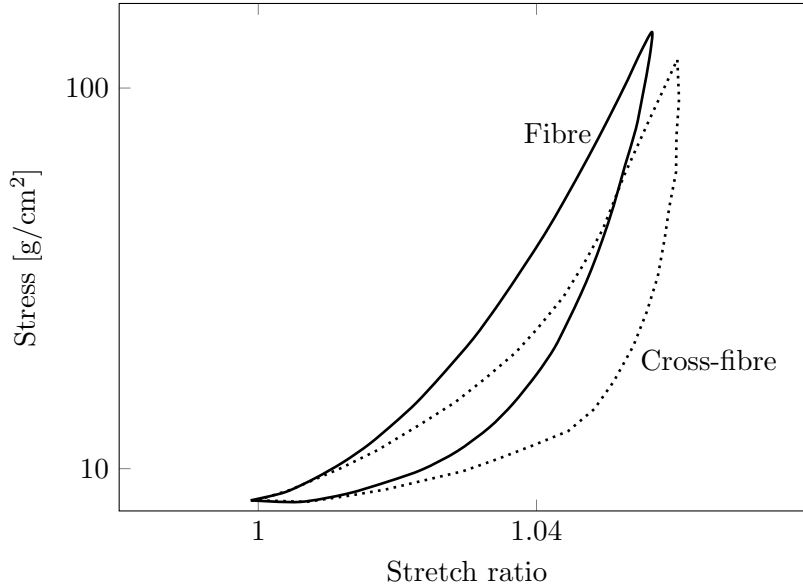


Figure 4.2: Stress-stress curve for cardiac tissue subjected to biaxial loading conditions, reproduced from [26].

the material response in the fibre direction is different to that in the cross-fibre direction. This is indicative of anisotropic behaviour. From biaxial studies we conclude that cardiac tissue exhibits viscoelastic behaviour, is highly non-linear and behaves stiffer when subjected to biaxial loading than uniaxial loading.

Although biaxial material testing highlights some anisotropic features, biaxial test data does not necessarily provide a complete description of the myocardium response. LeGrice et al. [74] used scanning electron microscopy to investigate the arrangement of cardiac muscle fibres in the heart, specifically the architecture of cardiac muscle sheets. The study confirmed the existence of discrete tissue layers in the ventricular wall. Each layer is approximately 4 heart muscle fibres thick and adjacent layers are separated by cleavage planes [75]. A layer contains tightly connected heart muscle fibres and is loosely connected to its adjacent layers [75, 125]. The spacing between layers is typically 1-2 muscle fibres thick.

The study by LeGrice et al. [74] suggests that the heart has an orthotropic material structure. That is, the laminar structure can be characterised locally with three orthogonal directions: the myofibre direction, the cross-fibre or sheet direction and the sheet-normal direction. LeGrice et al. also showed that the myocardium is not a homogeneous structure since the branch density, number of branches and perimysial fibre length vary across the ventricular wall.

Because of the orthotropic nature of cardiac tissue, deformation measured from uniaxial and biaxial testing data are not sufficient to characterise the response of the myocardium [22, 25, 58]. Holzapfel and Ogden [58] propose the use of a combination of measured data obtained from biaxial and shear tests for a more adequate description of cardiac material properties.

Other studies have also verified the orthotropic nature of cardiac tissue. Dokos et al. [28] performed shear tests on samples of cardiac tissue obtained from pig hearts. The samples, consisting of $3 \times 3 \times 3$ mm blocks, were taken from the mid wall of the left ventricle with the edges aligned with

the local material directions. The shear deformation modes are illustrated in Figure 4.3(a). The experimental results, reproduced in Figure 4.3(b), show that the shear properties depend on the loading direction. Note that according to Holzapfel and Ogden [58] the FS and FN labels are not consistent with the other graphs in the paper, therefore these two labels are switched. A similar study conducted by Sommer et al. [116], was performed on human heart specimens to determine tri-axial shear properties with similar findings to that of [28].

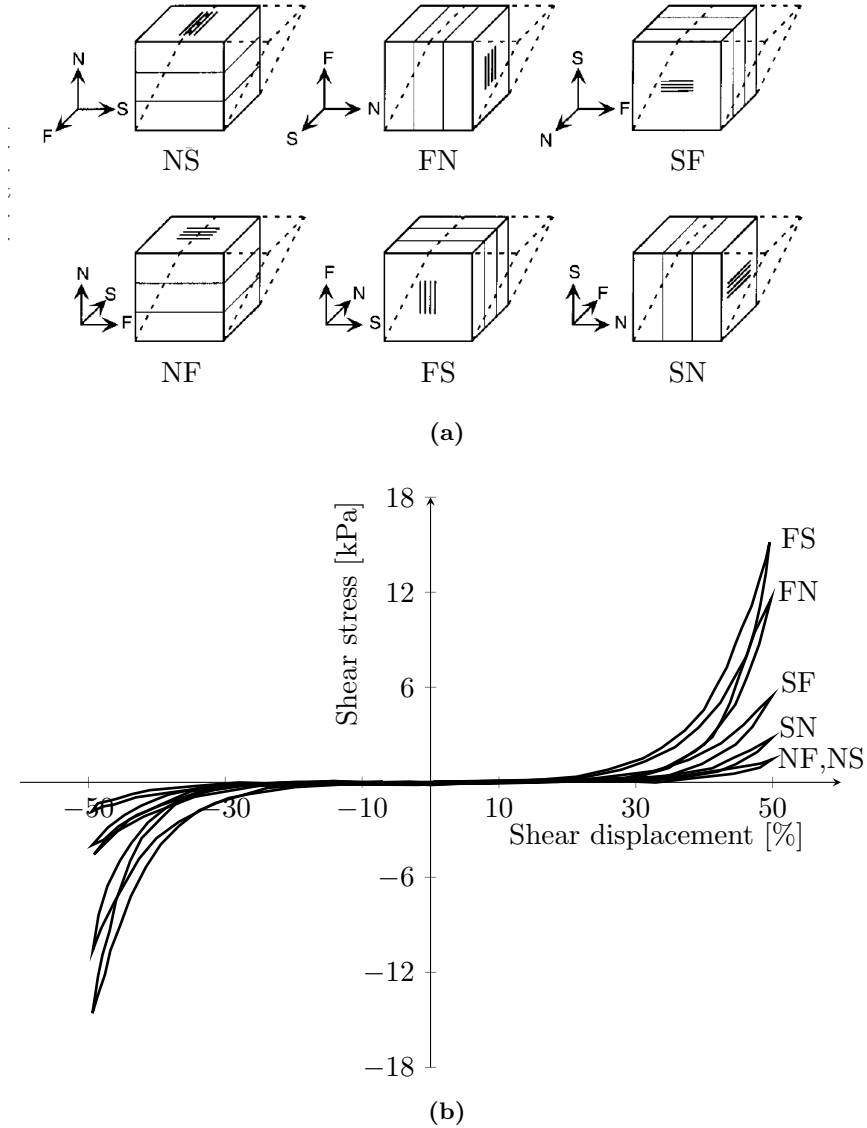


Figure 4.3: Results from tri-axial experiments performed on pig heart, reproduced from [28], including the (a) shear deformation modes where F, S and N are fibre, sheet and normal directions respectively and (b) the non-linear relationship between stress and deformation for the different deformation modes.

4.1.2 Classical Continuum Models

From experimental data and morphological studies we know that cardiac tissue is an anisotropic material. Isotropic models such as those used by Mirsky [82], Demiray [27] and Wong [136], are therefore inappropriate to describe the behaviour of the heart and in this section we only review transversely isotropic and orthotropic models.

Transversely Isotropic Model by Humphrey and Yin [25]

Transversely isotropic models assume that the material behaviour depends only on the fibre direction, but not on the sheet or the sheet-normal direction.

As previously mentioned, a mathematical model of the myocardial material should take into account the contributions from the muscle fibres as well as the extracellular matrix that surrounds the fibres [3, 42]. Humphrey and Yin [25] proposed a phenomenological strain energy function of the form

$$\psi = \psi_m(\mathbf{C}) + \psi_f(\lambda), \quad (4.1)$$

where ψ_m and ψ_f are the energy contributions from the surrounding matrix and muscle fibres respectively. ψ_m is a function of the the right Cauchy-Green deformation tensor \mathbf{C} while ψ_f is taken to be a function of the fibre stretch λ . In this regard the surrounding matrix contribution is assumed to be isotropic while the fibre contribution represents the transversely isotropic component of the tissue. The strain energy function of Humphrey and Yin is given by

$$\psi = \underbrace{c(e^{b(I_1-3)} - 1)}_{\psi_m} + \underbrace{A(e^{a(\lambda^2-1)^2} - 1)}_{\psi_f}, \quad (4.2)$$

where c , b , A and a are material parameters. Because this model is phenomenological, the parameters do not necessarily have any physical meaning.

Transversely Isotropic Model by Guccione et al. [50]

A commonly used strain energy function for soft biological tissue is a Fung-type model [43], which has the form

$$\psi_{fung} = \frac{C}{2}(e^Q - 1), \quad (4.3)$$

where Q is a function of the Green-Lagrange strain tensor \mathbf{E} . To quantify the material properties of the intact myocardium, Guccione et al. [50] based their transversely isotropic material model on the Fung-type strain energy function. The strain energy in [50] is given by Equation (4.3) with Q defined as

$$Q = 2b_1(E_{rr} + E_{ff} + E_{cc}) + b_3(E_{cc}^2 + E_{rr}^2 + E_{cr}^2 + E_{rc}^2) + b_4(E_{rf}^2 + E_{fr}^2 + E_{fc}^2 + E_{cf}^2), \quad (4.4)$$

where b_i are material parameters and E_{ij} are the components of the Green-Lagrange strain tensor. The indices f, r and c refer to the fibre, radial and cross-fibre directions. Guccione et al. also considered an isotropic case of the constitutive law in Equation (4.4). They found that the simulated motion of the heart with the isotropic model did not correspond to clinical observations, thereby highlighting the importance of an anisotropic model.

Orthotropic Model by Usyk et al. [125]

Orthotropic material models for cardiac tissue improve on transversely isotropic models since they capture the distinct responses in the fibre, cross-fibre and normal directions [87]. The study conducted by LeGrice et al. [74] support the view that transversely isotropic material models are not necessarily sufficient to describe the behaviour of the heart.

Usyk et al. [125] modelled the myocardium as an orthotropic and nearly incompressible material using the following strain energy function,

$$\psi = \frac{C}{2}(e^Q - 1) + C_{compr}(J \ln J - J + 1), \quad (4.5)$$

with

$$Q = b_{ff}E_{ff}^2 + b_{ss}E_{ss}^2 + b_{nn}E_{nn}^2 + b_{fs}(E_{fs}^2 + E_{sf}^2) + b_{fn}(E_{fn}^2 + E_{nf}^2) + b_{ns}(E_{ns}^2 + E_{sn}^2), \quad (4.6)$$

where J is the Jacobian and b_{ij} are material constants. The second term in Equation (4.5) is included to ensure nearly-incompressible behaviour. Usyk et al. compared the orthotropic model to the transversely isotropic model described in [50] and found that the orthotropic model produced results that were in better agreement with experimental data.

Orthotropic Model by Holzapfel and Ogden [58]

Holzapfel and Ogden [58] developed a material model based on the orthotropic structure of heart tissue. The strain energy function is given by

$$\psi = \frac{a}{2b}e^{b(I_4-4)} + \sum_{i=f,s} \frac{a_i}{2b_i}(e^{b_i(I_{4i}-1)^2} - 1) + \frac{a_{fs}}{2b_{fs}}(e^{b_{fs}I_{8fs}^2} - 1). \quad (4.7)$$

The invariants I_{4f} and I_{4s} are measures of strain in the fibre direction \mathbf{f}_0 and the sheet direction \mathbf{s}_0 ,

$$I_{4f} = \mathbf{f}_0 \cdot (\mathbf{C}\mathbf{f}_0), \quad I_{4s} = \mathbf{s}_0 \cdot (\mathbf{C}\mathbf{s}_0). \quad (4.8)$$

The invariant I_{8fs} accounts for the shear between the sheet and the fibre direction,

$$I_{8fs} = I_{8sf} = \mathbf{f}_0 \cdot (\mathbf{C}\mathbf{s}_0). \quad (4.9)$$

The first term in Equation (4.7) is associated with the isotropic behaviour of the matrix material. The invariants I_{4f} and I_{4s} account for the transversely isotropic behaviour and I_{8fs} accounts for the orthotropic nature of cardiac tissue. The material parameters a , b , a_f , a_s , b_f , b_s , a_{fs} and b_{fs} are positive constants that can be obtained by fitting the material law to experimental shear data.

The model in Equation (4.7) was used by a number of investigators. Nikou et al. [87] identified the eight material parameters from measured strain and pressure data. Through finite element modelling, they also compared the orthotropic material model in Equation (4.7) to the transversely isotropic model in Equation (4.4). Nikou et al. found that the orthotropic model was in better agreement with experimental data than the transversely isotropic model. Göktepe et al. [47] incorporated Equation (4.7) into a nonlinear finite element framework to model the passive myocardium. Wang et al. [134] modified Equation (4.7) to take into account residual stresses in the left ventricle and found that the addition of residual stresses only had a small effect on the pressure-volume relationship.

4.2 Active Behaviour of Cardiac Tissue

To understand the behaviour of the heart we need to characterise both passive and active properties [23]. As discussed in Chapter 2, an electrical impulse generated by the sinoatrial node travels across the heart resulting in ventricular contraction. The focus of this research is not on modelling the conduction system or the distribution of the impulse, but rather on the mechanical behaviour caused by the contraction, i.e. the resulting stresses and deformation caused by the contracting muscle fibres.

The stress resulting from the contraction of fibres can be included in one of two ways [2, 46]. In the first approach, the *active-stress* method, the passive material behaviour is determined in the absence of the active behaviour. That is, a strain energy function is obtained without including the active component. The active component is then simply included in the balance of linear momentum, such that the total stress $\boldsymbol{\sigma}$ is additively decomposed into an active and a passive stress part,

$$\boldsymbol{\sigma} = \boldsymbol{\sigma}_p + \boldsymbol{\sigma}_a, \quad (4.10)$$

where $\boldsymbol{\sigma}_p$ and $\boldsymbol{\sigma}_a$ are the passive and active stress components respectively. The equilibrium equation in Equation (3.57) for a classical continuum then reads

$$\text{div}(\boldsymbol{\sigma}_p + \boldsymbol{\sigma}_a) + \rho \mathbf{b} = 0. \quad (4.11)$$

A second, less common approach, is to adopt a method of multiplicatively decomposing the deformation gradient,

$$\mathbf{F} = \mathbf{F}_e \mathbf{F}_a, \quad (4.12)$$

where \mathbf{F} is the total deformation gradient and \mathbf{F}_e and \mathbf{F}_a represent the deformations of the passive and active contributions. This is referred to as the *active-strain* approach. A number of active-stress and active-strain models are reviewed by Ambrosi and Pezzuto [2].

Although the active-strain approach is considered to be more robust [2], we only consider active-stress models here. The reader is referred to [46, 84, 94, 120] for examples of active-strain models. The active-stress approach has the advantage that the material parameters for the passive and active models can be determined independently, i.e. from different experimental data sets. Passive material parameters can be obtained from experiments where there is no electrical stimulation of the cardiac muscle fibres and likewise the active parameters can be determined from experiments where the passive parameters are fixed [46].

Hill model

A number of earlier models describing muscle mechanics were based on the well-known Hill model [54]. Hill's equation, which gives a relationship between force and velocity, reads

$$(v + b)(P + a) = b(P_0 + a), \quad (4.13)$$

where v is the contraction velocity and P is the muscle tension. P_0 , a and b are constants. There are several issues with this model as noted by Fung [42]: single muscle twitches, unstimulated muscle behaviour and muscle subjected to time-varying strain cannot be described with Hill's equation.

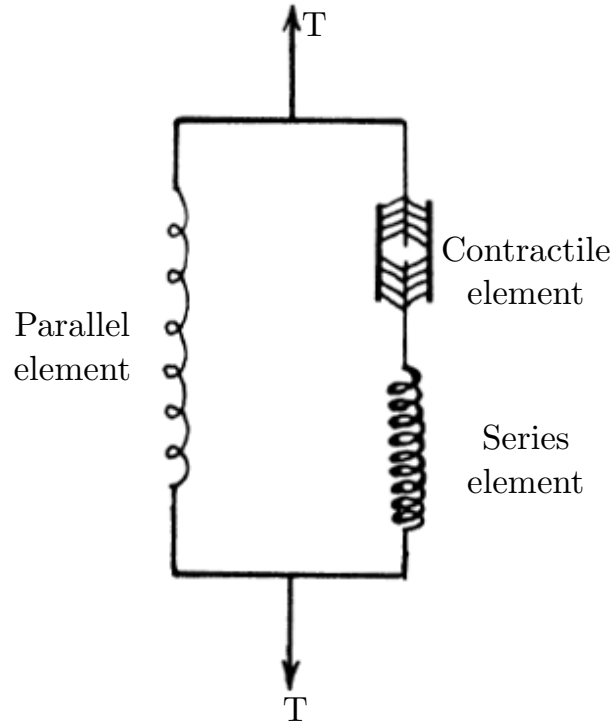


Figure 4.4: Hill's three element model for muscle, from [42].

A modification of Hill's equation is Hill's three-element model, illustrated in Figure 4.4. The three-element model represents muscle with a contractile element connected in series with a non-linear spring component. These two components are arranged in parallel with another non-linear spring element. The parallel element represents the behaviour of the muscle at rest. The contractile element is responsible for muscle shortening, while the muscle stress is determined by the spring elements [41]. This approach allows the total stress to be additively decomposed into active and passive components [46]. A more detailed analysis of the Hill model is found in [42].

The classical Hill model as presented above is unsuitable for modelling the active behaviour of cardiac tissue [42, 49]. Despite its shortcomings, some authors have successfully adapted the Hill model to simulate the contraction of the heart. For example, Göktepe et al. [46] generalised the Hill model to accommodate large strains. Göktepe et al. not only additively decomposed the strain energy function but also multiplicatively decomposed the deformation gradient into passive and active parts. They therefore used a combination of the active-stress and the active-strain approach. Göktepe et al. showed that their approach can be employed successfully to model the excitation-contraction of the heart.

Huxley's model

Huxley's model [62] for muscle contraction is another approach common when simulating the mechanics of cardiac tissue. This model is based on the sliding filament theory discussed in Chapter 2. From the sliding filament theory we know that muscle contraction depends amongst other things on crossbridge attachment. The rate of crossbridge attachment in Huxley's model is determined by

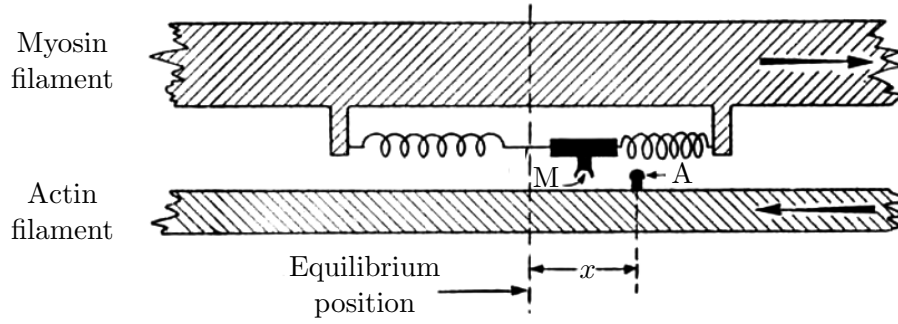


Figure 4.5: Schematic diagram of Huxley's model adapted from [62]. A denotes the binding site and M the myosin head.

the position of the actin-binding site relative to the nearest crossbridge (or myosin head) [62, 101], see x in Figure 4.5. The crossbridge force is taken to be a linear function of displacement and the crossbridges therefore act as springs. Both the Hill model and Huxley model predict a hyperbolic force-velocity relationship.

Elastance Model by Guccione et al. [51]

Guccione and McCulloch [49] derived a constitutive relation for active cardiac tissue that describes isometric¹ and isotonic² contractions as well as the so-called deactivation phenomenon observed in cardiac tissue. Guccione et al. [51] evaluated this deactivation model and also considered two simplifications of the model, namely a Hill and a time varying elastance model. They found that although the complex deactivation model is considered more accurate, the mechanical behaviour of the complex model was similar to that of the less complicated Hill and elastance models.

Because of its simplicity, the elastance model is often the preferred approach [22]. The elastance model in [51] assumes that all the fibres contract simultaneously and that the active stress T_A is a function of the peak intracellular calcium concentration Ca_0 , the maximum peak intracellular calcium concentration $(Ca_0)_{max}$ and the sarcomere length l ,

$$T_A = T_{max} \frac{Ca_0^2}{Ca_0^2 + ECa_{50}^2} C_t \quad \text{with} \quad ECa_{50} = \frac{(Ca_0)_{max}}{\sqrt{e^{B(l-l_0)} - 1}}, \quad (4.14)$$

where B is a constant and C_t is defined as

$$C_t = \frac{1}{2}(1 - \cos \omega) \quad (4.15)$$

$$\omega = \begin{cases} \pi \frac{t}{t_0} & \text{if } 0 \leq t < t_0 \\ \pi \frac{t-t_0+t_r}{t_r} & \text{if } t_0 \leq t < t_0 + t_r \\ 0 & \text{if } t_0 + t_r \leq t \end{cases} \quad (4.16)$$

where t is the time after contraction has started and t_0 is the time to reach the peak tension.

¹Isometric contraction: contraction at a constant length

²Isotonic contraction: contraction at a constant tension

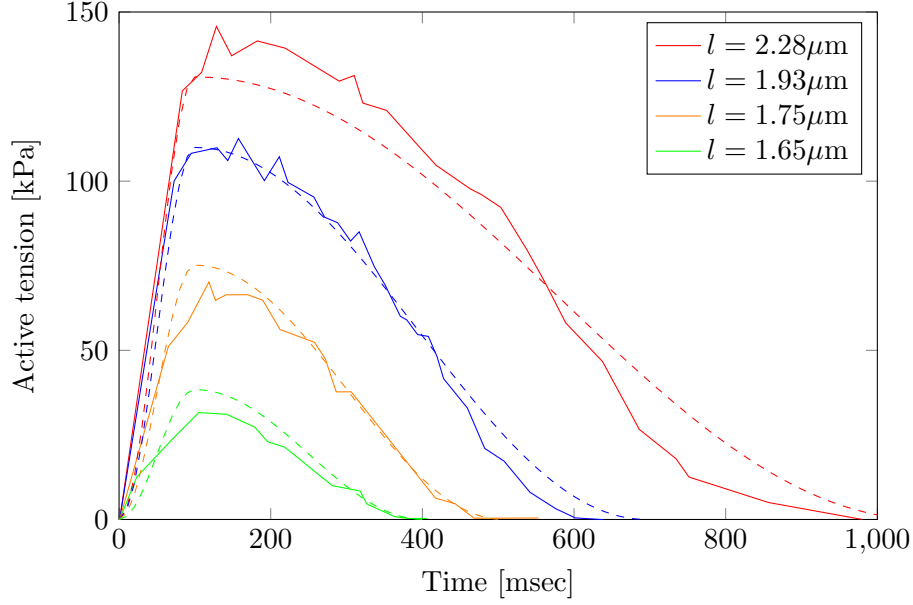


Figure 4.6: Measured active tension (solid lines) from Van Heuningen et al. [129] and the predicted isometric tension (dashed lines) for different sarcomere lengths using the elastance model by Guccione et al. [51].

The relaxation duration t_r is given by $t_r = ml + b$, where m and b are constants. The sarcomere length is given as

$$l = l_R \sqrt{2E_{ff} + 1} \quad \text{with} \quad E_{ff} = \mathbf{E} : \mathbf{f}_0 \otimes \mathbf{f}_0, \quad (4.17)$$

where \mathbf{f}_0 and l_R are the fibre direction and the sarcomere length in the reference configuration. Figure 4.6 compares the elastance model to measured active tension data. The figure shows the active tension during a contraction in which the sarcomere length is kept constant, known as an isometric twitch. The tension development was computed for different sarcomere lengths. The tension was computed using GNU Octave and the material constants were in accordance with [51]: $Ca_0 = 4.35 \mu M$, $(Ca_0)_{max} = 4.35 \mu M$, $B = 4.75 \mu m^{-1}$, $l_0 = 1.58 \mu m$, $m = 1.0489 s \mu m^{-1}$ and $b = -1.429 s$. The curves indicate a good correlation between the model and experimental measurements.

A number of models assume that the active tension develops only in the longitudinal direction of the cardiac fibres [51, 139]. That is,

$$\boldsymbol{\sigma}_{A(local)} = T_A \mathbf{f} \otimes \mathbf{f}, \quad (4.18)$$

where $\boldsymbol{\sigma}_{A(local)}$ is the active stress tensor in the local coordinate system and \mathbf{f} is the fibre direction. However, some studies have shown that significant tension also develops in the transverse direction (cross-fibre direction) [14, 141]. In most cases the transverse stress is specified to be 40% of the tension in the fibre direction such that

$$\mathbf{T}_{A(local)} = T_A \mathbf{f} \otimes \mathbf{f} + 0.4 T_A \mathbf{s} \otimes \mathbf{s}, \quad (4.19)$$

where \mathbf{f} and \mathbf{s} are the fibre direction and the cross-fibre direction respectively.

4.3 Heart Geometry

The relationship between the wall stresses of the ventricles and the blood pressure depends amongst others factors on the chosen geometry [3, 80]. Much of earlier literature is devoted to investigating the behaviour of the left ventricle. It was only after the improvement of computational resources and imaging modalities that the right ventricle started to appear in cardiac models.

4.3.1 Modelling the Left Ventricle

Simple shapes, such as spheres and cylinders, were first used to investigate left ventricular mechanics. Spherical shapes were utilised because of the difficulty to reproduce the fibre arrangement, simulate large deformation and model anisotropic material behaviour. Studies that assume a spherical geometry for the left ventricle include those by Holt et al. [55], Rodbard [105] and Mirsky [82]. Cylindrical models allowed investigators to include the fibre direction and anisotropic material properties. These models also made it easier to study large deformations without the need for complicated computational methods, such as finite element methods [51]. Cylindrical models consist of thick-walled cylinders that represent the equatorial region of the left ventricle. These simple models were fairly popular and used in a number of cardiac models, see for example [3, 9, 50, 51, 60, 64]. Even though the cylindrical model is considered an improvement on the spherical model [9], results obtained from a cylindrical model can still only be used to describe the behaviour of regions close to the equatorial area of the left ventricle [51].

In more recent studies, the left ventricle is often modelled by an ellipse truncated at the top to resemble the base of the heart. Van den Broek and Van den Broek [8] investigated left ventricular contractions with a thick-walled ellipsoidal model. The model consisted of layers and each layer was endowed with a different fibre orientation to simulate the change in fibre angle as one transverse the heart wall. Other studies that used ellipsoidal models include [63, 117, 118, 119, 125, 128, 136]. Because the ellipsoidal geometry approximates the base and equatorial region better than the apical region, the computed deformations near the base and equator are in closer agreement with measured data compared to deformation computed at the apex [125]. Figure 4.7 illustrates examples of cylindrical and ellipsoidal models.

The use of medical imaging modalities allows for accurate reconstruction of the heart geometry. Computed tomography images were utilised by Ordas et al. [90] to construct a statistical shape of the whole human heart. Vinson et al. [130] estimated the three-dimensional geometry of the left ventricle using biplane angiography. Magnetic resonance imaging is another imaging tool widely used to develop three-dimensional models [24]. Figure 4.7(c) shows a finite element mesh that was used in a patient-specific study of the left ventricle. The mesh was created using magnetic resonance images.

4.3.2 Bi-Ventricular Models

A bi-ventricular model includes both the left and the right ventricle. While the left ventricle is often said to have an ellipsoidal shape, the right ventricle forms a crescent-like shape around the

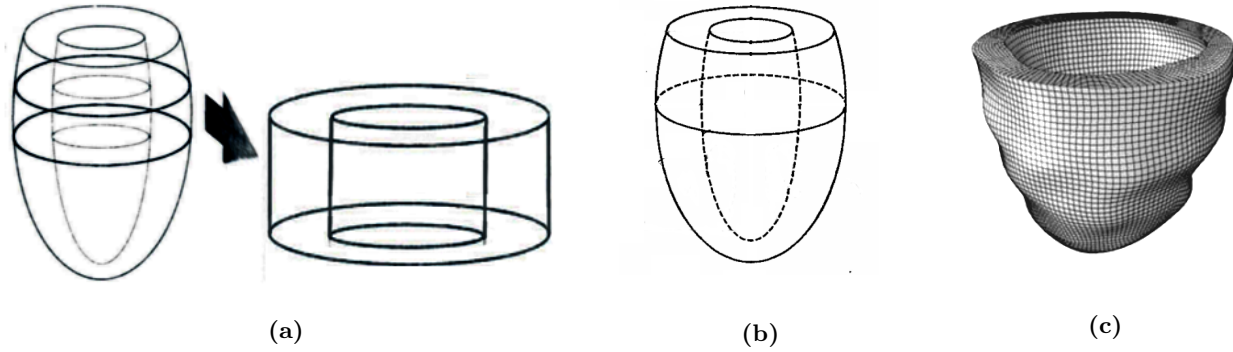


Figure 4.7: Geometries used to represent the left ventricle include (a) thick-walled cylindrical models only valid if restricted to the equatorial region, from [50] (b) ellipsoidal models, figure adapted [6] and (c) image-based geometries, from [134].

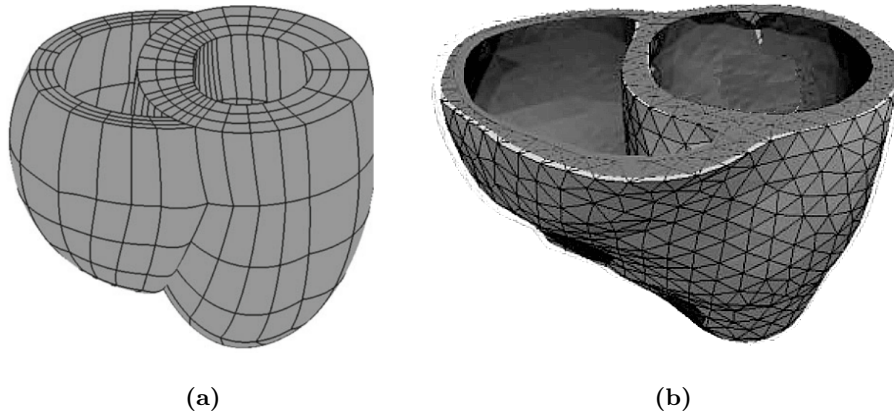


Figure 4.8: Different geometries for bi-ventricular heart models, including (a) a generic bi-ventricular model that approximates both ventricles as two truncated ellipsoids, from [95] and (b) an image based model used for patient specific simulations, from [36].

left ventricle and overall has a more complicated shape [71]. One of the most popular methods to create bi-ventricular geometries is using ellipsoids. An idealised bi-ventricular model consists of two ellipsoids truncated at their tops as shown in Figure 4.8(a). Some of the studies that employ this method include [47, 95, 114]. As in the case of the left ventricle, medical imaging modalities can also be used to create bi-ventricular models. Figure 4.8(b) presents a bi-ventricular model that was reconstructed from magnetic resonance images.

4.3.3 Myocardial Fibre Orientation

One of the major goals in computational modelling of the heart is to gain insight into how the tissue structure affects the pumping performance of the heart [85]. When modelling the mechanical action of the heart, the two important quantities are deformation and stress [126]. Non-linear finite element modelling of the left ventricle performed by Van Campen et al. [126] and Bovendeerd et al. [6] showed that the spatial variation of the wall stress depends on the orientation of the cardiac muscle fibres in the ventricle.

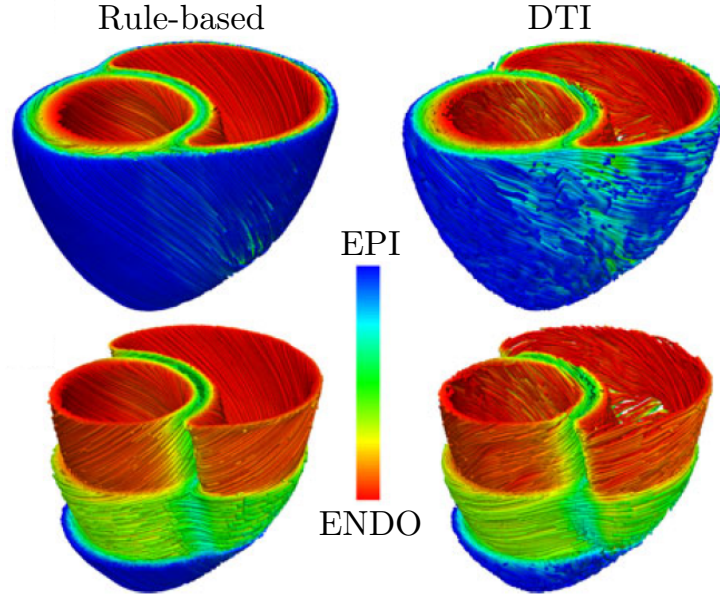


Figure 4.9: Longitudinal fibre direction obtain from diffusion tensor imaging and a rule-based method of a canine ventricle, from [4].

Since the contraction depends on the fibre arrangement, it is important to represent the fibre orientation accurately in the computational model [4]. The fibre orientation is assigned either through rule-based or image-based approaches [4]. Rule-based methods are based on mathematical expressions that describe the fibre direction, while imaging techniques typically use diffusion tensor imaging (DTI), a non-invasive technique, to obtain the fibre arrangement. DTI is based on measurements of the diffusion of water molecules in biological tissues [88, 115]. Because diffusion in tissues depends on the fibre direction, DTI gives information about the degree of anisotropy and the structural layout of the fibres [115]. DTI is often applied in studies to characterise brain tissue [1] but can also be used to obtain the fibre arrangement in the heart, see for example [40, 44, 93].

Bayer et al. [4] developed a Laplace-Dirichlet rule-based algorithm that was used to define the fibre arrangement in a cardiac model. The method is based on solving the Laplace equation with Dirichlet boundaries. Electrical activation of the heart was simulated using this model and compared to a DTI-based model with fairly similar results. Figure 4.9 compares the DTI-derived longitudinal fibre orientation with the rule-based method developed by Bayer et al. Wong and Kuhl [138] provides another example of the rule-based method. They utilised Poisson interpolation to generate a vector field that described the fibre direction. Wong and Kuhl successfully applied this method to a bi-ventricular heart as well as a patient-specific heart, thereby demonstrating the flexibility and robustness of their method.

4.4 Hemodynamics

Although diastolic filling, isovolumetric contraction and isovolumetric relaxation can easily be modelled using pressure loads and deformation constraints, see for example [114], there are a number of possibilities to simulate blood ejection. These typically include models that describe the interaction of the ventricles with the arterial system. The objective for this study is not to model the wave

Table 4.1: Hydraulic and electrical representations of the two and three element windkessel models (figures adapted from [135]).

	2 element Windkessel	3 element Windkessel
Hydraulic analogue		
Electrical analogue		

propagation, pressure or flow within the arterial tree, but rather on the outflow of blood from the ventricles and the corresponding effect on the ventricular pressure and volume.

Among the simpler models that are used to incorporate the arterial system are tube and lumped parameter models [72]. In the former an elastic or visco-elastic tube is used to model the arterial tree [15]. The tube has an effective length and is either straight or tapered. Tube models can account for the elasticity of the arterial walls as well as its geometry [127].

A widely used lumped parameter model is the Windkessel model. The arterial Windkessel model draws an analogy between the flow of blood from the heart and an electrical circuit (see Table 4.1). A Windkessel model represents the arterial system with resistances, capacitors and inductors, with each of these components representing a different feature of the arterial system [72]. Since we are only interested in the overall effect of the arterial system, the Windkessel model is sufficient [127] and is therefore discussed in more detail below.

The two-element Windkessel or Frank model formulated by Frank Otto [39], is the simplest form of the Windkessel model. The two-element circuit consists of a resistance and a compliant element. As blood is ejected, small arterioles and capillaries resist the flow of blood [127]. The peripheral resistance R in the circuit represents the sum of all the individual resistances. The large compliant arteries are thought of as chambers and are represented by the capacitor C in the electrical circuit. The current $I(t)$ and potential difference $P(t)$ of the circuit are analogous to blood flow and the aortic pressure respectively. The two-element model predicts the aortic pressure decay during diastole as

$$P(t) = P_0 e^{-\frac{t}{RC}}, \quad (4.20)$$

where P_0 is the end-systolic pressure (or pressure at the start of diastole).

To improve the prediction of the relation between pressure and blood flow, a third element, the characteristic impedance Z_a , is added. This component represents the impedance of the aorta, but is also sometimes thought of as the resistance that the flow experiences as it moves through the

semilunar valve [135]. The expression that governs the flow and pressure is given by

$$\left(1 + \frac{Z_c}{R}\right)I(t) + CZ_c \frac{dI(t)}{dt} = \frac{P(t)}{R} + C \frac{dP(t)}{dt} . \quad (4.21)$$

4.5 Summary of Computational Cardiac Modelling

From material testing and morphological studies we can conclude that the myocardium is a non-homogeneous structure that exhibits non-linear and anisotropic material behaviour. This chapter listed a number of constitutive laws to describe the behaviour of cardiac tissue, including the transversely isotropic model by Humphrey and Yin [25], the Fung-type model by Guccione et al. [50] and the orthotropic models by Usyk et al. [125] and Holzapfel and Ogden [58].

There exists a variety of mathematical models that can be employed to simulate the active contraction of cardiac fibres. We noted that the active tension is typically included using either an active-stress or an active-strain approach. In Section 4.2 we presented three models to simulate the active behaviour, namely the Hill model, Huxley's model and the elastance model by Guccione et al [51].

In this chapter we also discussed the improvement of the heart geometries used in cardiac models. Initially the left ventricle was approximated using simple geometric shapes, however recent studies employ imaging techniques to obtain the heart geometry.

It is up to the investigator to choose an appropriate material model and three-dimensional geometry that would produce sufficiently accurate results. But it is also important to avoid over complication. A complex model might be too computationally expensive, while a too simplistic model could lead to unsatisfactory results.

Chapter 5

Introduction to Micromorphic Modelling

Classical continuum mechanics can be applied to describe a variety of macroscopic phenomena. However, classical theories are often inadequate when dealing with phenomena that occur on lower scales. To address this issue, one can utilise so-called generalised continuum theories. Generalised continuum theories are well suited to describe materials with granular or fibrous structures [140]. According to Madeo [78] there are two methods with which classical theories may be generalised. In the first the continuum kinematics remains the same but the chosen constitutive law depends on higher gradients of the displacement field. In the second method the kinematic expressions are extended by adding additional degrees of freedom to the existing displacement degrees of freedom. In this chapter we consider the microcontinuum, a specific case of the latter approach.

Microcontinuum theories enhance the classical approach by attaching additional vectors (called directors) to the continuum particle. The directors allow one to describe the micro-deformation of the particle. Therefore, a microcontinuum is often described as a body that consists of a continuous set of deformable particles [31]. This approach makes it possible to account for the deformation of micro-constituents since the continuum particles themselves are now also considered to be deformable.

A simple method to determine whether the microcontinuum theory is appropriate, is to consider the ratio of the characteristic length L of the continuum and its intrinsic length λ [31]. If the characteristic length is much greater than intrinsic length, i.e. $\frac{L}{\lambda} > 1$, then a classical continuum formulation is appropriate and will give reasonable results. If however the two lengths are comparable, i.e. $\frac{L}{\lambda} \approx 1$, then a classical formulation will give erroneous results since the micro-constituents in this case will have a significant effect on the macroscopic behaviour of the body.

We propose the use of a micromorphic model, a subclass of the microcontinuum, to describe the passive response of cardiac tissue. The goal of this chapter is to give a brief introduction to the micromorphic theory and to outline the kinematics and strain measures. The theory discussed in the remainder of this chapter is heavily based on the works of Sansour et al. [112] and Von Hoegen et al. [131].

5.1 The 3M Continua

The term *3M continua* refers to the three subclasses of the microcontinuum namely micropolar, microstretch and micromorphic. As mentioned before, to include the micro-deformation of a continuum particle, we attach directors to the particle and as a consequence introduce additional degrees of freedom. The microcontinuum subclass depends on how these directors are allowed to deform [31].

A microcontinuum whose material particles are allowed to experience micro-stretch, micro-rotation and micro-shear is referred to as a micromorphic continuum. This is the most general case of the microcontinuum. Within first-grade theories, a particle is endowed with three directors. Therefore a first-grade micromorphic continuum has a total of 12 degrees of freedom. Apart from the 3 classical displacement degrees of freedom, 9 additional degrees of freedom are introduced to describe the deformation of the particle (3 micro-shear + 3 micro-rotation + 3 micro-stretch).

When a micromorphic continuum is constrained such that its particles experience only micro-rotation and micro-stretch it is referred to as a microstretch continuum. A first-grade microstretch continuum has a total of 7 degrees of freedom (3 classical displacement + 3 micro-rotation + 1 micro-stretch). If the microcontinuum is further constrained to only allow rigid rotation of the particle, the number of degrees of freedom reduces to 6 (3 classical displacement + 3 micro-rotation) and is known as a micropolar or Cosserat continuum.

The idea of adding additional degrees of freedom was first presented in the seminal work of the Cosserat brothers [21]. Their approach was extended by Eringen [31, 33] and Mindlin [81] to the micromorphic theory.

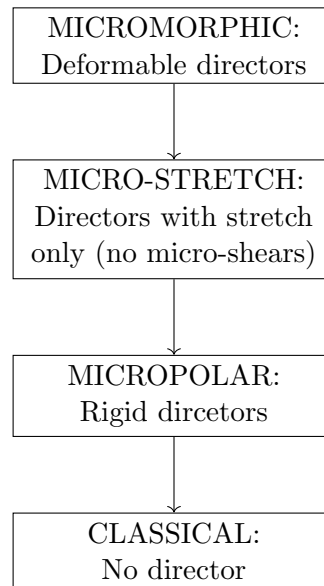


Figure 5.1: Diagram illustrating the different subclasses of micromorphic continua, from [31]. The directors of a micropolar (or Cosserat) continuum are rigid. A particle is only allowed to rotate. A microstretch continuum has directors that can experience micro-stretch and micro-rotation. A particle in a micromorphic continuum is allowed to rotate, stretch and experience micro-shear.

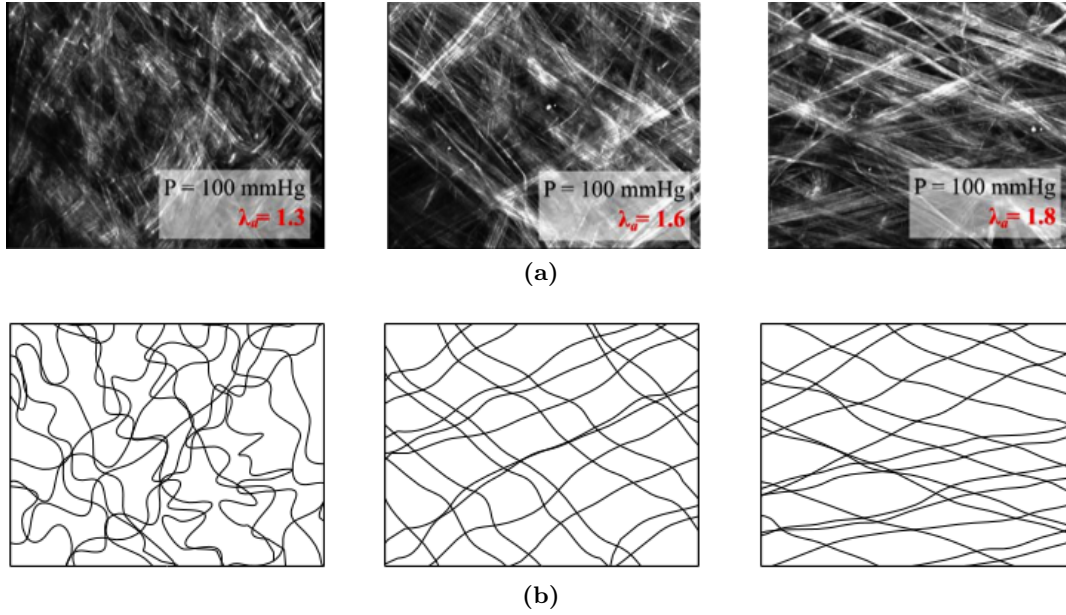


Figure 5.2: (a) Multiphoton microscopy imaging of collagen fibres experiencing reorientation under axial load, adapted from [70] and (b) a schematic illustration of fibre kinematics.

5.2 Microcontinuum Modelling of Biological Tissue

Classical continuum models assume that a continuum body experiences affine deformation [70]. The material models discussed in Chapter 4 therefore predict that cardiac muscle fibres orientate and deform along with the matrix material. This implies that there is no relative deformation between these two constituents. However, several authors have investigated the kinematics of collagen fibres in biological material and observed a realignment of the fibres under certain loading conditions [5, 53, 70]. Krasny et al. [70] performed experimental testing on carotid arteries and found that fibres realigned when the arteries were tensioned axially, see Figure 5.2. Little reorientation was observed during inflation of the arteries. Billiar and Sacks [5] subjected collagenous fibres to biaxial stretching in order to investigate the fibre kinematics of planar tissue and also observed reorientation of the fibres.

The microcontinuum theory has been applied in a number of studies related to the biomechanics of bone, i.e. hard biological tissue. Yang and Lakes [140] experimentally and analytically analysed the behaviour of bone in bending. Experimental testing was conducted on compact bone and the analytical analysis was based on the couple stress and the micropolar theory. The results obtained by Yang and Lakes showed that the behaviour of bone is more accurately described with a microcontinuum theory than a classical or a viscoelastic theory. Park and Lakes [92] performed experimental testing on wet and dry bone specimens to investigate the deformation along the lateral surface of the specimen and also to quantify the displacement of micro-constituents. The behaviour of wet bone was well predicted by a micropolar model while dry bone behaved classically. The micropolar behaviour was attributed to the osteonal microstructure of bone. Other studies related to bone mechanics and microcontinuum modelling include those by Fatemi et al. [35], Buechner and Lakes [13], Eringen [32] and Rosenberg and Cimrman [107].

Even though the microcontinuum modelling of hard biological tissue has proven to be successful, it has only been extended to soft tissues fairly recently. The lack of microcontinuum modelling in biomechanical problems may be attributed to the large number of material parameters one has to deal with [131].

Sack et al. [111] modelled the behaviour of the myocardium as a Cosserat continuum which allowed non-local material responses to be included in the model. Thurieau et al. [121] investigated the deformation of a healthy and a diseased left ventricle using a microstretch formulation. Heart tissue was modelled as a microstretch medium and the infarcted areas were modelled as a microdilation medium. The microdilation medium is a constrained version of the microstretch medium which only allows for breathing type motion. The left ventricle was represented with a hollow cylinder and the investigation focussed on small strains only. When compared to a healthy heart, Thurieau et al. observed reduced axial deformation for the infarcted regions, which is in line with clinical studies.

For a background on the micromorphic theory as well as academic examples illustrating the use of micromorphic modelling, the reader is referred to [131].

5.3 Kinematics

Figure 5.3 depicts the deformation of a micromorphic continuum which is also viewed as a two-level continuum [73]. In the reference configuration the micromorphic continuum occupies the space \mathcal{G}_0 which consists of a macro-space \mathcal{B}_0 and a micro-space \mathcal{S}_0 . Similarly, in the current configuration, the macro-space \mathcal{B} and the micro-space \mathcal{S} compose the micromorphic body \mathcal{G} . The macro-space is described by the curvilinear coordinates ϑ^i and the micro-space by the curvilinear coordinates ζ^α . The number of $\alpha = 1, 2, 3 \dots M$ is chosen based on the internal structure of the material [112].

The position of a point in \mathcal{G}_0 is expressed as the sum of the macro-placement $\mathbf{X} \in \mathcal{B}_0$ and the micro-placement $\mathbf{\Xi} \in \mathcal{S}_0$. Accordingly, we define the position $\tilde{\mathbf{X}} \in \mathcal{G}_0$ as

$$\tilde{\mathbf{X}} = \mathbf{X}(\vartheta^i) + \mathbf{\Xi}(\vartheta^i, \zeta^\alpha). \quad (5.1)$$

Since the micro-placement is relative to the macro-placement it also depends on the macro-coordinates ϑ^i . Similarly, the position in the deformed configuration \mathcal{G} is given by

$$\tilde{\mathbf{x}} = \mathbf{x}(\vartheta^i, t) + \boldsymbol{\xi}(\vartheta^i, \zeta^\alpha, t), \quad (5.2)$$

where $\mathbf{x} \in \mathcal{B}$ and $\boldsymbol{\xi} \in \mathcal{S}$ are the positions in the macro-space and micro-space respectively. The motion $\tilde{\varphi}$ maps each point in the reference configuration to its current position,

$$\tilde{\varphi} : \mathcal{G}_0 \rightarrow \mathcal{G}. \quad (5.3)$$

In Section 3.8 we introduced the covariant base vectors, which are tangent to the curvilinear coordinates. The covariant base vectors in the reference configuration are given as

$$\tilde{\mathbf{G}}_i = \frac{\partial \tilde{\mathbf{X}}}{\partial \vartheta^i}, \quad \tilde{\mathbf{I}}_\alpha = \frac{\partial \tilde{\mathbf{X}}}{\partial \zeta^\alpha}, \quad (5.4)$$

where $\tilde{\mathbf{G}}_i$ and $\tilde{\mathbf{I}}_\alpha$ are the covariant base vectors of the macro-scale and the micro-scale respectively.

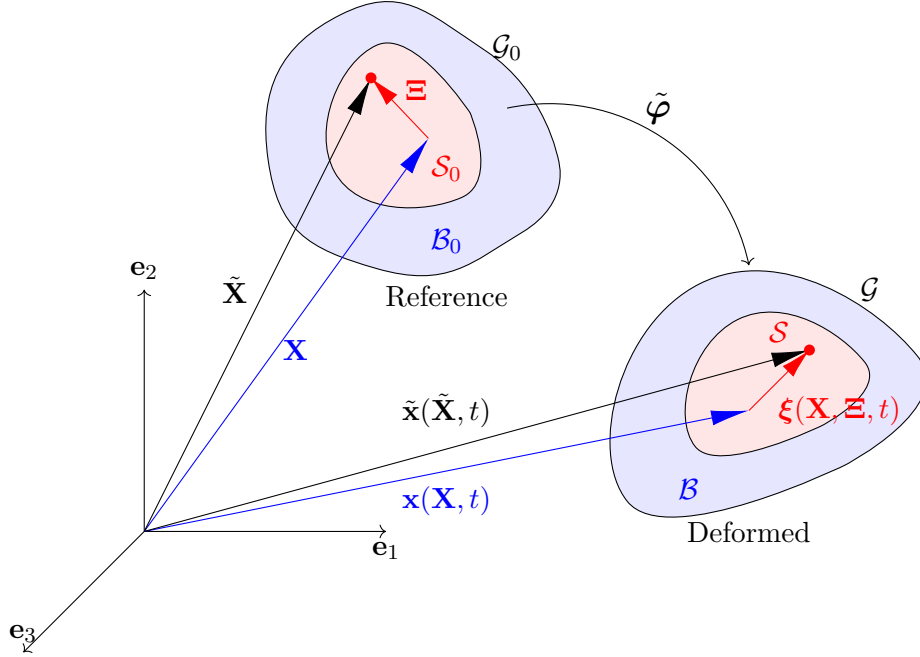


Figure 5.3: Motion of a micromorphic continuum. The continuum is composed of a macro- and a micro-space. The micro-space represents the deformable particle.

Similarly we define the base vectors in the current configuration as

$$\tilde{\mathbf{g}}_i = \frac{\partial \tilde{\mathbf{x}}}{\partial \vartheta^i}, \quad \tilde{\mathbf{i}}_\alpha = \frac{\partial \tilde{\mathbf{x}}}{\partial \zeta^\alpha}. \quad (5.5)$$

The form of the micro-space placement has to be chosen a priori. The choice defines the type of micro-continua, as discussed in Section 5.1. A popular choice for a micromorphic formulation is a linear approximation [31],

$$\Xi = \zeta^\alpha \mathbf{A}_\alpha, \quad \xi = \zeta^\alpha \mathbf{a}_\alpha, \quad (5.6)$$

where the micro-directors \mathbf{A}_α and \mathbf{a}_α are vector functions that orientate the micro-placement in reference and current configurations. As shown by Sansour et al. [112], a linear ansatz does not produce strain measures of full rank. In this work we follow Sansour et al. and introduce scalar fields χ_β to obtain a micro-motion with a quadratic ansatz,

$$\tilde{\mathbf{x}} = \mathbf{x} + \zeta^\alpha (1 + \zeta^\beta \chi_\beta) \mathbf{a}_\alpha. \quad (5.7)$$

The introduction of the directors \mathbf{a}_α and scalar fields χ_β add extra degrees of freedom to the problem (on top of the classical displacement degrees of freedom) that need be solved. The macroscopic displacement \mathbf{u} and the micro-deformation \mathbf{w}_α are defined as

$$\mathbf{u} = \mathbf{X} - \mathbf{x}, \quad \mathbf{w}_\alpha = \mathbf{a}_\alpha - \mathbf{A}_\alpha. \quad (5.8)$$

In this work we take α to be one and the director is chosen to align with the longitudinal direction of the cardiac fibre. For $\alpha = 1$, we need to solve a total of 7 degrees of freedom: 3 classical displacement (u_1, u_2, u_3) , 3 micro-deformation (w_1, w_2, w_3) and 1 scalar field (χ) .

5.4 Strain Measures

The generalised deformation gradient is given as

$$\tilde{\mathbf{F}} = \frac{\partial \tilde{\mathbf{x}}}{\partial \tilde{\mathbf{X}}} . \quad (5.9)$$

Since $\tilde{\mathbf{x}}$ is a function of the curvilinear coordinates ν^i and ζ^α , we can expand the deformation gradient as follows,

$$\begin{aligned} \tilde{\mathbf{F}} &= \frac{\partial \tilde{\mathbf{x}}}{\partial \nu^i} \otimes \frac{\partial \nu^i}{\partial \tilde{\mathbf{X}}} + \frac{\partial \tilde{\mathbf{x}}}{\partial \zeta^\gamma} \otimes \frac{\partial \zeta^\gamma}{\partial \tilde{\mathbf{X}}} \\ &= \frac{\partial \tilde{\mathbf{x}}}{\partial \nu^i} \otimes \mathbf{G}^i + \frac{\partial \tilde{\mathbf{x}}}{\partial \zeta^\gamma} \otimes \mathbf{I}^\gamma , \end{aligned} \quad (5.10)$$

where \mathbf{G}^i and \mathbf{I}^γ are the contravariant base vectors for the macro-and micro-space respectively. Furthermore, utilising the Equation (5.7) and taking $\alpha = 1$, we obtain the deformation gradient as

$$\begin{aligned} \tilde{\mathbf{F}} &= \left(\frac{\partial \mathbf{x}}{\partial \nu^i} + \zeta \zeta \frac{\partial \chi}{\partial \nu^i} \mathbf{a} + \zeta (1 + \zeta \chi) \frac{\partial \mathbf{a}}{\partial \nu^i} \right) \otimes \mathbf{G}^i \\ &\quad + \left(\frac{\partial \zeta}{\partial \zeta} \mathbf{a} + \zeta \frac{\partial \mathbf{a}}{\partial \zeta} + \frac{\partial \zeta}{\partial \zeta} \zeta \chi \mathbf{a} + \zeta \frac{\partial \zeta}{\partial \zeta} \chi \mathbf{a} + \zeta \chi \frac{\partial \mathbf{a}}{\partial \zeta} \right) \otimes \mathbf{I} \\ &= \underbrace{\left(\frac{\partial \mathbf{x}}{\partial \nu^i} + \zeta^2 \frac{\partial \chi}{\partial \nu^i} \mathbf{a} + \zeta (1 + \zeta \chi) \frac{\partial \mathbf{a}}{\partial \nu^i} \right)}_{\mathbf{g}_i} \otimes \mathbf{G}^i \\ &\quad + \underbrace{\left(\mathbf{a} + \zeta \frac{\partial \mathbf{a}}{\partial \zeta} + 2\zeta \chi \mathbf{a} + \zeta^2 \chi \frac{\partial \mathbf{a}}{\partial \zeta} \right)}_{\mathbf{i}} \otimes \mathbf{I} \\ &= \mathbf{g}_i \otimes \mathbf{G}^i + \mathbf{i}_\gamma \otimes \mathbf{I} , \end{aligned} \quad (5.11)$$

where \mathbf{g}_i and \mathbf{i}_γ are the covariant base vectors for the macro-space and micro-space in the current configuration. The generalised right Cauchy-Green tensor is expressed as

$$\begin{aligned} \tilde{\mathbf{C}} &= \tilde{\mathbf{F}}^T \tilde{\mathbf{F}} \\ &= (\mathbf{G}^k \otimes \mathbf{g}_k + \mathbf{I} \otimes \mathbf{i})(\mathbf{g}_i \otimes \mathbf{G}^i + \mathbf{i} \otimes \mathbf{I}) \\ &= (\mathbf{G}^k \otimes \mathbf{g}_k)(\mathbf{g}_i \otimes \mathbf{G}^i) + (\mathbf{G}^k \otimes \mathbf{g}_k)(\mathbf{i} \otimes \mathbf{I}) + (\mathbf{I} \otimes \mathbf{i})(\mathbf{g}_i \otimes \mathbf{G}^i) + (\mathbf{I} \otimes \mathbf{i})(\mathbf{i} \otimes \mathbf{I}) \\ &= \underbrace{\mathbf{g}_k \cdot \mathbf{g}_i (\mathbf{G}^k \otimes \mathbf{G}^i)}_{\tilde{\mathbf{C}}^{(0)}} + \underbrace{\mathbf{g}_k \cdot \mathbf{i} (\mathbf{G}^k \otimes \mathbf{I}) + \mathbf{i} \cdot \mathbf{g}_i (\mathbf{I} \otimes \mathbf{G}^i)}_{\tilde{\mathbf{C}}^{(1)}} + \underbrace{\mathbf{i} \cdot \mathbf{i} (\mathbf{I} \otimes \mathbf{I})}_{\tilde{\mathbf{C}}^{(2)}} . \end{aligned} \quad (5.12)$$

Through inspection of the basis vectors, we see that $\mathbf{C}^{(0)}$ is defined purely in the macro-space, $\mathbf{C}^{(1)}$ is defined in both the macro-and micro-space while $\mathbf{C}^{(2)}$ is defined in the micro-space only. We define the generalised Green-Lagrange strain tensor $\tilde{\mathbf{E}}$ as,

$$\tilde{\mathbf{E}} = \frac{1}{2}(\tilde{\mathbf{C}} - \tilde{\mathbf{C}}_0) = \tilde{\mathbf{E}}^{(0)} + \tilde{\mathbf{E}}^{(1)} + \tilde{\mathbf{E}}^{(2)} , \quad (5.13)$$

where $\tilde{\mathbf{C}}_0$ is the right Cauchy-Green deformation tensor at time t_0 .

5.5 Weak Formulation

We assume that the generalised second Piola-Kirchhoff stress tensor $\tilde{\mathbf{S}}$ takes the form

$$\tilde{\mathbf{S}} = \tilde{\mathbf{S}}^{(0)}(\tilde{\mathbf{C}}^{(0)}) + \tilde{\mathbf{S}}^{(1)}(\tilde{\mathbf{C}}^{(1)}) + \tilde{\mathbf{S}}^{(2)}(\tilde{\mathbf{C}}^{(2)}) . \quad (5.14)$$

Furthermore we extend the principle of virtual work (weak formulation) in Equation (3.97) to include the generalised stress and strain measures,

$$\frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} \tilde{\mathbf{S}} : \delta \tilde{\mathbf{C}} \, dS dV - W_{ext} = 0 . \quad (5.15)$$

The external work W_{ext} is a function of the body and traction forces as defined in [112],

$$W_{ext} = \int_{\mathcal{B}} \mathbf{b} \cdot \delta \mathbf{u} dV + \int_{\mathcal{B}} \mathbf{l}_\alpha \cdot \delta \mathbf{w}_\alpha dV + \int_{\partial \mathcal{B}} \mathbf{t} \cdot \delta \mathbf{u} dA + \int_{\mathcal{B}} \mathbf{q}_\alpha \cdot \delta \mathbf{w}_\alpha dA , \quad (5.16)$$

where \mathbf{b} and \mathbf{t} are the body and the traction forces, and \mathbf{l} and \mathbf{q} are the higher-order body and traction forces. In the above $\delta \mathbf{u}$ is the virtual displacement and $\delta \mathbf{w}$ is the virtual micro-deformation.

Substituting Equations (5.12) and (5.14) into the weak formulation, we arrive at

$$\frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} (\tilde{\mathbf{S}}^{(0)} + \tilde{\mathbf{S}}^{(1)} + \tilde{\mathbf{S}}^{(2)}) : \delta(\tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{C}}^{(2)}) \, dS dV - W_{ext} = 0 . \quad (5.17)$$

Since the dot product satisfies the distributive law,

$$\begin{aligned} \frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} \left\{ \tilde{\mathbf{S}}^{(0)} : \delta(\tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{C}}^{(2)}) + \tilde{\mathbf{S}}^{(1)} : \delta(\tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{C}}^{(2)}) \right. \\ \left. + \tilde{\mathbf{S}}^{(2)} : \delta(\tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{C}}^{(2)}) \right\} dS dV - W_{ext} = 0 . \end{aligned} \quad (5.18)$$

Because $\tilde{\mathbf{S}}^{(0)}$, $\tilde{\mathbf{S}}^{(1)}$ and $\tilde{\mathbf{S}}^{(2)}$ are functions only of $\tilde{\mathbf{C}}^{(0)}$, $\tilde{\mathbf{C}}^{(1)}$ and $\tilde{\mathbf{C}}^{(2)}$ respectively, the expression above simplifies to

$$\frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} (\tilde{\mathbf{S}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{S}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)}) \, dS dV - W_{ext} = 0 . \quad (5.19)$$

The above expression is non-linear in \mathbf{u} and \mathbf{w} and therefore needs to be linearised. As before the aim is to find the macroscopic displacement \mathbf{u} as well as the micro-deformation \mathbf{w} such that the residual G is minimised,

$$G(\mathbf{u}, \mathbf{a}) = \frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} (\tilde{\mathbf{S}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{S}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)}) \, dS dV - W_{ext} = 0 . \quad (5.20)$$

To linearise the residual we consider a first-order Taylor expansion,

$$G(\mathbf{u}_{k+1}, \mathbf{a}_{k+1}) = G(\mathbf{u}_k, \mathbf{a}_k) + \Delta G_k . \quad (5.21)$$

where ΔG_k is the variation in the residual.

Using Equation (5.20), we obtain the linearised residual as

$$\begin{aligned}
G(\mathbf{u}_{k+1}, \mathbf{a}_{k+1}) &= \frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} (\tilde{\mathbf{S}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{S}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)}) dS dV - W_{ext} \\
&+ \frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} \left(\frac{\partial \tilde{\mathbf{S}}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} \Delta \tilde{\mathbf{C}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \frac{\partial \tilde{\mathbf{S}}^{(1)}}{\partial \tilde{\mathbf{C}}^{(1)}} \Delta \tilde{\mathbf{C}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \frac{\partial \tilde{\mathbf{S}}^{(2)}}{\partial \tilde{\mathbf{C}}^{(2)}} \Delta \tilde{\mathbf{C}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)} \right) dS dV \\
&+ \frac{1}{2} \int_{\mathcal{B}_0} \int_{\mathcal{S}_0} (\tilde{\mathbf{S}}^{(0)} : \Delta \delta \tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{S}}^{(1)} : \Delta \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \Delta \delta \tilde{\mathbf{C}}^{(2)}) dS dV, \tag{5.22}
\end{aligned}$$

where $\Delta \tilde{\mathbf{C}}^{(i)}$ are the variations in the right Cauchy-Green tensor and $\frac{\partial \tilde{\mathbf{S}}^{(i)}}{\partial \tilde{\mathbf{C}}^{(i)}}$ are the elasticity tensors. In the above we assumed that the external forces are conservative. Instead of having just one elasticity tensor, as in the classical continuum case, we now have three elasticity tensors. The elasticity tensors $\mathbf{C}^{(i)}$ can also be defined in terms of the strain energy function,

$$\mathbf{C}^{(i)} = 2 \frac{\partial^2 \tilde{\psi}}{\partial \tilde{\mathbf{C}}^{(i)} \otimes \partial \tilde{\mathbf{C}}^{(i)}}. \tag{5.23}$$

A similar process to that outlined in Chapter 3 is followed to obtain the stiffness matrix and the internal and external force vectors, albeit a more complicated process.

5.6 Material Model for Passive Cardiac Tissue

In this section we present a micromorphic material model that can describe the passive behaviour of cardiac tissue. The proposed strain energy is separated into three components. The first describes the deformation of the bulk material while the second and third components are related to the deformation experienced by the cardiac fibres. The second component describes the relative deformation between the bulk material and the cardiac fibres and the third describes the pure micro-motion of the cardiac fibres. To model the behaviour of the bulk material we utilise the strain energy function by Usyk et al. [125]. The surrounding tissue is therefore modelled as an orthotropic material and the material model is formulated in terms of the macroscopic deformation tensor $\tilde{\mathbf{E}}^{(0)}$. To second and third components of the strain energy function make use of the following invariants,

$$I_{8ff} = \tilde{\mathbf{C}}^{(1)} : \mathbf{V}^f \otimes \mathbf{V}^f, \tag{5.24}$$

$$I_{8fs} = \tilde{\mathbf{C}}^{(1)} : \mathbf{V}^f \otimes \mathbf{V}^s, \tag{5.25}$$

$$I_{8fn} = \tilde{\mathbf{C}}^{(1)} : \mathbf{V}^f \otimes \mathbf{V}^n, \tag{5.26}$$

$$I_{4f} = \tilde{\mathbf{C}}^{(2)} : \mathbf{V}^f \otimes \mathbf{V}^f, \tag{5.27}$$

where \mathbf{V}^f , \mathbf{V}^s and \mathbf{V}^n are the initial fibre, sheet and normal directions respectively. The invariants I_{8ff} , I_{8fs} and I_{8fn} give information about the relative deformation between the bulk material and the cardiac fibres, while I_{4f} is a measure of the micro-deformation experienced by the cardiac fibres.

Considering the above, we can express the strain energy as

$$\tilde{\psi} = \tilde{\psi}(\tilde{\mathbf{E}}^{(0)}, I_{8ff}, I_{8fs}, I_{8fn}, I_{4f}). \tag{5.28}$$

We utilise Fung-type functions to define the contributions for the relative deformation between the cardiac fibres and the bulk material as well as the micro-deformation of the fibres,

$$\begin{aligned}
\tilde{\psi} &= \underbrace{\frac{A_0}{2B_0}(e^{B_0 Q_m} - 1)}_{\text{bulk deformation}} + \underbrace{A_{comp} \left(\ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} \right)^2}_{\text{compressibility}} \\
&\quad + \underbrace{\frac{A_{8ff}}{2B_{8ff}}(e^{B_{8ff} Q_{8ff}} - 1) + \frac{A_{8fs}}{2B_{8fs}}(e^{B_{8fs} Q_{8fs}} - 1) + \frac{A_{8fn}}{2B_{8fn}}(e^{B_{8fn} Q_{8fn}} - 1)}_{\text{relative deformation}} \\
&\quad + \underbrace{\frac{A_{4f}}{2B_{4f}}(e^{B_{4f} Q_{4f}} - 1)}_{\text{micro-deformation}} \\
&= \tilde{\psi}_{bulk}^{(0)} + \tilde{\psi}_{comp}^{(0)} + \tilde{\psi}^{(1)} + \tilde{\psi}^{(2)}
\end{aligned} \tag{5.29}$$

with

$$\begin{aligned}
Q_m &= b_{ff}(\tilde{E}_{ff}^{(0)})^2 + b_{ss}(\tilde{E}_{ss}^{(0)})^2 + b_{nn}(\tilde{E}_{nn}^{(0)})^2 + b_{fs}((\tilde{E}_{fs}^{(0)})^2 + (\tilde{E}_{sf}^{(0)})^2) \\
&\quad + b_{fn}((\tilde{E}_{fn}^{(0)})^2 + (\tilde{E}_{nf}^{(0)})^2) + b_{sn}((\tilde{E}_{sn}^{(0)})^2 + (\tilde{E}_{ns}^{(0)})^2)
\end{aligned} \tag{5.30}$$

$$Q_{8ff} = (I_{8ff} - 2)^2 \tag{5.31}$$

$$Q_{8fs} = I_{8fs}^2 \tag{5.32}$$

$$Q_{fn} = I_{8fn}^2 \tag{5.33}$$

$$Q_{4f} = (I_{8ff} - 1)^2 \tag{5.34}$$

where A_0 , B_0 and b_{ij} are material parameters related to the bulk material and A_{8ff} , B_{8ff} , A_{8fn} , B_{8fn} , A_{8fs} , B_{8fs} , A_{4f} and B_{4f} are material parameters related to the cardiac fibres. The scaling parameter A_{comp} is used to ensure the incompressibility condition is satisfied. With Equation (3.67) we find the generalised Second Piola-Kirchhoff stress tensor to be

$$\begin{aligned}
\tilde{\mathbf{S}} = \frac{\partial \tilde{\psi}}{\partial \tilde{\mathbf{E}}} &= 2 \frac{\partial \tilde{\psi}}{\partial \tilde{\mathbf{C}}} \\
&= 2 \frac{\partial \tilde{\psi}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} + 2 \frac{\partial \tilde{\psi}^{(1)}}{\partial \tilde{\mathbf{C}}^{(1)}} + 2 \frac{\partial \tilde{\psi}^{(2)}}{\partial \tilde{\mathbf{C}}^{(2)}} \\
&= \tilde{\mathbf{S}}^{(0)} + \tilde{\mathbf{S}}^{(1)} + \tilde{\mathbf{S}}^{(2)}.
\end{aligned} \tag{5.35}$$

Furthermore we substitute in the proposed strain energy function to obtain the generalised stress tensors $\tilde{\mathbf{S}}^{(0)}$, $\tilde{\mathbf{S}}^{(1)}$ and $\tilde{\mathbf{S}}^{(2)}$.

$$\begin{aligned}
\tilde{\mathbf{S}}^{(0)} = 2 \frac{\partial \tilde{\psi}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} &= 2 \frac{\partial \tilde{\psi}_{bulk}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} + 2 \frac{\partial \tilde{\psi}_{comp}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} \\
&= \frac{\partial \tilde{\psi}_{bulk}^{(0)}}{\partial \tilde{\mathbf{E}}^{(0)}} + 2 \frac{\partial \tilde{\psi}_{comp}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} \\
&= \tilde{\mathbf{S}}_{bulk}^{(0)} + \tilde{\mathbf{S}}_{comp}^{(0)}
\end{aligned} \tag{5.36}$$

The two stress components in the previous expression are given by

$$\tilde{\mathbf{S}}_{bulk}^{(0)} = \frac{\partial \tilde{\psi}_{bulk}^{(0)}}{\partial \tilde{\mathbf{E}}^{(0)}} = \frac{1}{2} A_0 e^{B_0 Q_m} \frac{\partial Q_m}{\partial \tilde{\mathbf{E}}^{(0)}}, \quad (5.37)$$

$$\begin{aligned} \tilde{\mathbf{S}}_{comp}^{(0)} &= 2 \frac{\partial \tilde{\psi}_{comp}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} = 2 \frac{\partial A_{comp} \left(\ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} \right)^2}{\partial \tilde{\mathbf{C}}^{(0)}} \\ &= 4 A_{comp} \ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} \frac{\partial \ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}}}{\partial \tilde{\mathbf{C}}^{(0)}} \\ &= 4 A_{comp} \ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} \frac{1}{2} \frac{1}{\sqrt{\det \tilde{\mathbf{C}}^{(0)}}} \frac{1}{\sqrt{\det \tilde{\mathbf{C}}^{(0)}}} \det \tilde{\mathbf{C}}^{(0)} (\tilde{\mathbf{C}}^{(0)})^{-1} \\ &= 2 A_{comp} \ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} (\tilde{\mathbf{C}}^{(0)})^{-1}, \end{aligned} \quad (5.38)$$

where we have used the rule $\frac{\partial \det \mathbf{A}}{\partial \mathbf{A}} = (\det \mathbf{A}) \mathbf{A}^{-T}$. Similarly we find that

$$\begin{aligned} \tilde{\mathbf{S}}^{(1)} &= 2 \frac{\partial \tilde{\psi}^{(1)}}{\partial \tilde{\mathbf{C}}^{(1)}} \\ &= A_{8ff} e^{B_{8ff} Q_{8ff}} \frac{\partial Q_{ff}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fs} e^{B_{8fs} Q_{8fs}} \frac{\partial Q_{fs}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fn} e^{B_{8fn} Q_{8fn}} \frac{\partial Q_{fn}}{\partial \tilde{\mathbf{C}}^{(1)}}, \end{aligned} \quad (5.39)$$

$$\begin{aligned} \tilde{\mathbf{S}}^{(2)} &= 2 \frac{\partial \tilde{\psi}^{(2)}}{\partial \tilde{\mathbf{C}}^{(2)}} \\ &= A_{4f} e^{B_{4f} Q_{8ff}} \frac{\partial Q_{4f}}{\partial \tilde{\mathbf{C}}^{(2)}}. \end{aligned} \quad (5.40)$$

The complete expression for the generalised stress tensor therefore reads

$$\begin{aligned} \tilde{\mathbf{S}} &= \frac{1}{2} A_0 e^{B_0 Q_m} \frac{\partial Q_m}{\partial \tilde{\mathbf{E}}^{(0)}} + 2 A_{comp} \ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} (\tilde{\mathbf{C}}^{(0)})^{-1} + \\ &A_{8ff} e^{B_{8ff} Q_{8ff}} \frac{\partial Q_{ff}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fs} e^{B_{8fs} Q_{8fs}} \frac{\partial Q_{fs}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fn} e^{B_{8fn} Q_{8fn}} \frac{\partial Q_{fn}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{4f} e^{B_{4f} Q_{8ff}} \frac{\partial Q_{4f}}{\partial \tilde{\mathbf{C}}^{(2)}}. \end{aligned} \quad (5.41)$$

The strain energy components $\tilde{\psi}^{(1)}$ and $\tilde{\psi}^{(2)}$ are related to the deformation experienced by the cardiac fibres. With the introduction of the director \mathbf{a} , we have automatically achieved a transversely isotropic material for the fibre component of the strain energy function. To model a fully orthotropic material one would need to include two more directors that align with the sheet and sheet-normal directions.

It is important to identify the elasticity tensors since they provide a relationship between the strain and stress in a material. Using the expression for the stress, the three passive elasticity tensors are found to be

$$\begin{aligned}\mathbf{C}^{(i)} &= \frac{\partial \tilde{\mathbf{S}}^{(i)}}{\partial \tilde{\mathbf{C}}^{(i)}} = 2 \frac{\partial^2 \tilde{\psi}}{\partial \tilde{\mathbf{C}}^{(i)} \otimes \partial \tilde{\mathbf{C}}^{(i)}} \\ &= \frac{1}{2} \frac{\partial \tilde{\mathbf{S}}^{(i)}}{\partial \tilde{\mathbf{E}}^{(i)}} = \frac{1}{2} \frac{\partial^2 \tilde{\psi}}{\partial \tilde{\mathbf{E}}^{(i)} \otimes \partial \tilde{\mathbf{E}}^{(i)}}\end{aligned}\quad (5.42)$$

Therefore,

$$\begin{aligned}\mathbf{C}^{(0)} &= \frac{1}{2} \frac{\partial \tilde{\mathbf{S}}_{bulk}^{(0)}}{\partial \tilde{\mathbf{E}}^{(0)}} + \frac{\partial \tilde{\mathbf{S}}_{comp}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} \\ &= \frac{1}{4} A_0 B_0 e^{B_0 Q_m} \frac{\partial Q_m}{\partial \tilde{\mathbf{E}}^{(0)}} \otimes \frac{\partial Q_m}{\partial \tilde{\mathbf{E}}^{(0)}} + \frac{1}{4} A_0 e^{B_0 Q_m} \frac{\partial^2 Q_m}{\partial \tilde{\mathbf{E}}^{(0)} \otimes \partial \tilde{\mathbf{E}}^{(0)}} \\ &\quad + 2A_{comp}(\tilde{\mathbf{C}}^{(0)})^{-1} \otimes (\tilde{\mathbf{C}}^{(0)})^{-1} - 2A_{comp} \left(\ln \sqrt{\det \tilde{\mathbf{C}}^{(0)}} (\tilde{\mathbf{C}}^{(0)})^{-T} \otimes (\tilde{\mathbf{C}}^{(0)})^{-1} \right)\end{aligned}\quad (5.43)$$

$$\begin{aligned}\mathbf{C}^{(1)} &= \frac{\partial \tilde{\mathbf{S}}^{(1)}}{\partial \tilde{\mathbf{C}}^{(1)}} \\ &= A_{8ff} B_{8ff} e^{B_{8ff} Q_{8ff}} \frac{\partial Q_{ff}}{\partial \tilde{\mathbf{C}}^{(1)}} \otimes \frac{\partial Q_{ff}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8ff} e^{B_{8ff} Q_{8ff}} \frac{\partial Q_{ff}}{\partial \tilde{\mathbf{C}}^{(1)} \otimes \tilde{\mathbf{C}}^{(1)}} \\ &\quad + A_{8fs} B_{8fs} e^{B_{8fs} Q_{8fs}} \frac{\partial Q_{fs}}{\partial \tilde{\mathbf{C}}^{(1)}} \otimes \frac{\partial Q_{fs}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fs} e^{B_{8fs} Q_{8fs}} \frac{\partial Q_{fs}}{\partial \tilde{\mathbf{C}}^{(1)} \otimes \tilde{\mathbf{C}}^{(1)}} \\ &\quad + A_{8fn} B_{8fn} e^{B_{8fn} Q_{8fn}} \frac{\partial Q_{fn}}{\partial \tilde{\mathbf{C}}^{(1)}} \otimes \frac{\partial Q_{fn}}{\partial \tilde{\mathbf{C}}^{(1)}} + A_{8fn} e^{B_{8fn} Q_{8fn}} \frac{\partial Q_{fn}}{\partial \tilde{\mathbf{C}}^{(1)} \otimes \tilde{\mathbf{C}}^{(1)}}\end{aligned}\quad (5.44)$$

$$\begin{aligned}\mathbf{C}^{(2)} &= \frac{\partial \tilde{\mathbf{S}}^{(1)}}{\partial \tilde{\mathbf{C}}^{(2)}} \\ &= A_{4f} B_{4f} e^{B_{4f} Q_{8ff}} \frac{\partial Q_{4f}}{\partial \tilde{\mathbf{C}}^{(2)}} \otimes \frac{\partial Q_{4f}}{\partial \tilde{\mathbf{C}}^{(2)}} + A_{4f} e^{B_{4f} Q_{8ff}} \frac{\partial Q_{4f}}{\partial \tilde{\mathbf{C}}^{(2)} \otimes \partial \tilde{\mathbf{C}}^{(2)}}\end{aligned}\quad (5.45)$$

5.7 Summary of Micromorphic Modeling

Most continuum models of the heart are based on classical continuum mechanics. Unfortunately, a classical continuum formulation lacks the ability to describe non-affine deformation, i.e. it cannot describe relative deformation of the micro-constituents to the bulk material. In this chapter we proposed the use of a micromorphic formulation to obtain a more complete description of the mechanical behaviour of cardiac tissue.

This chapter introduced the kinematics of a micromorphic continuum as well as the generalised strain measures used to formulate constitutive models. Finally, a material model for passive cardiac tissue

was presented. An orthotropic material model is used to describe the behaviour of the surrounding tissue and Fung-type strain energy functions are used to model the relative and micro-deformation of the cardiac muscle fibres.

Chapter 6

Modelling the Active Stresses in the Myocardium

For a complete description of the mechanical behaviour of the heart we not only need to include the passive response but also the active response of cardiac tissue, i.e. the tension generated when the tissue is electrically stimulated. In this chapter we present a model for the active behaviour of cardiac tissue and discuss how the model is incorporated into the weak formulation presented in Chapter 5.

6.1 Mathematical Model for Active Cardiac Tissue

To include the active behaviour we follow an active-stress approach. That is, the total wall stress $\tilde{\mathbf{S}}_{total}$ is additively decomposed into a passive and an active contribution,

$$\tilde{\mathbf{S}}_{total} = \tilde{\mathbf{S}}_P + \tilde{\mathbf{S}}_A, \quad (6.1)$$

where $\tilde{\mathbf{S}}_A$ is the active stress tensor. The passive stress $\tilde{\mathbf{S}}_P$ is given in Equation (5.41). To obtain the active stress we utilise the elastance model that was presented in Chapter 4. We repeat the model here for convenience. All the cardiac fibres are assumed to contract simultaneously and the active tension is taken to act only in the fibre direction. As proposed by Guccione et al. [51], the active tension T_A has the following form,

$$T_A = T_{max} \frac{Ca_0^2}{Ca_0^2 + ECa_{50}^2} C_t, \quad (6.2)$$

where T_{max} is the tension at the longest sarcomere length and Ca_0 is the peak intracellular calcium concentration. The calcium sensitivity ECa_{50} is defined as

$$ECa_{50} = \frac{(Ca_0)_{max}}{\sqrt{e^{B(l-l_0)} - 1}}, \quad (6.3)$$

where B , l_0 and $(Ca_0)_{max}$ are model parameters. B determines the shape of the peak tension-sarcomere length curve, $(Ca_0)_{max}$ is the maximum peak intracellular calcium concentration and l_0

is the sarcomere length at which no active tension is generated. The internal variable C_t represents the extend of activation [122]. The internal variable varies between 0 and 1 and is defined as

$$C_t = \frac{1}{2}(1 - \cos\omega). \quad (6.4)$$

The parameter ω has different values during activation and relaxation,

$$\omega = \begin{cases} \pi \frac{t}{t_0} & \text{if } 0 \leq t < t_0 \quad (\text{activation}) \\ \pi \frac{t-t_0+t_r}{t_r} & \text{if } t_0 \leq t < t_0 + t_r \quad (\text{relaxation}) \\ 0 & \text{if } t_0 + t_r \leq t \end{cases} \quad (6.5)$$

where t is the time after contraction has started and t_0 is the time needed to reach the peak tension. The duration of relaxation t_r is a function of the sarcomere length l ,

$$t_r = ml + b, \quad (6.6)$$

where the slope m and time-intercept b are model parameters. Guccione et al. [51] defined the sarcomere length as a function of the stretch along the initial fibre direction, see Equation (4.17). In the present analysis the sarcomere length is defined in terms of the micro-director \mathbf{a} . As mentioned in Chapter 5, the micro-director lies along the cardiac fibre. As the body is loaded, the direction of the director changes and therefore also the direction in which contraction takes place. In order to be consistent with the micromorphic theory, the active stress and the sarcomere length should be defined in terms of the director. Consider the following,

$$\begin{aligned} (d\mathbf{X})^2 &= d\mathbf{X} \cdot d\mathbf{X} \\ &= \mathbf{F}^{-1}d\mathbf{x} \cdot \mathbf{F}^{-1}d\mathbf{x} \\ &= d\mathbf{x} \cdot (\mathbf{F}\mathbf{F}^T)^{-1}d\mathbf{x}, \end{aligned} \quad (6.7)$$

where $d\mathbf{X}$ and $d\mathbf{x}$ are line elements in the reference and current configurations. If we let $d\mathbf{X}$ and $d\mathbf{x}$ represent a sarcomere in the reference and the current state respectively, then we can rewrite the above as

$$l_R^2 = l^2 \hat{\mathbf{a}} \cdot (\mathbf{F}\mathbf{F}^T)^{-1} \hat{\mathbf{a}} \quad \text{with} \quad \hat{\mathbf{a}} = \frac{\mathbf{a}}{\|\mathbf{a}\|}, \quad (6.8)$$

where l_R is the sarcomere length at rest and l is the current sarcomere length. The unit vector $\hat{\mathbf{a}}$ represents the direction of the director and therefore also the direction of a sarcomere in the current configuration. Finally, using Equation (6.8) the sarcomere length is given as

$$l = \frac{l_R}{\sqrt{\hat{\mathbf{a}} \cdot (\mathbf{F}\mathbf{F}^T)^{-1} \hat{\mathbf{a}}}} = \frac{l_R}{\sqrt{\mathbf{F}^{-T} \mathbf{F}^{-1} : \hat{\mathbf{a}} \otimes \hat{\mathbf{a}}}}. \quad (6.9)$$

Since we assume that the active stress develops in the fibre direction only, the generalised Cauchy stress tensor is given by

$$\tilde{\boldsymbol{\sigma}}_A = T_A(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}}), \quad (6.10)$$

where $\hat{\mathbf{a}}$ represents the fibre direction.

Using Equation (3.44), the Second Piola-Kirchhoff stress is obtained as

$$\tilde{\mathbf{S}}_A^{(0)} = J\mathbf{F}^{-1}T_A(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}})\mathbf{F}^{-T}, \quad (6.11)$$

where \mathbf{F} is the deformation gradient associated with the macro-deformation.

6.2 Weak Formulation

The active-stress approach allows us to simply add the active stress component to the governing equation. With the active stress component, the weak formulation stated in Equation (5.19) now reads

$$\frac{1}{2} \int_{\mathcal{B}} \int_{\mathcal{S}} (\tilde{\mathbf{S}}^{(0)} + \tilde{\mathbf{S}}_A^{(0)}) : \delta \tilde{\mathbf{C}}^{(0)} + \tilde{\mathbf{S}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)} \, dS dV - W_{ext} = 0, \quad (6.12)$$

where we have assumed that the active stress acts on the macro-scale only and is therefore only included in the first term of the integrand.

6.2.1 Linearisation of the Weak Form

As before we consider the residual G and its linearisation, but now we also include the active stress. We extend the linearised residual in Equation (5.22) to include the active stress component,

$$\begin{aligned} G(\mathbf{u}_{k+1}, \mathbf{a}_{k+1}) = & \frac{1}{2} \int_{\mathcal{B}} \int_{\mathcal{S}} \left(\tilde{\mathbf{S}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \underbrace{\tilde{\mathbf{S}}_A^{(0)} : \delta \tilde{\mathbf{C}}^{(0)}}_{\text{active stress}} + \tilde{\mathbf{S}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)} \right) dS dV - W_{ext} \\ & + \frac{1}{2} \int_{\mathcal{B}} \int_{\mathcal{S}} \left(\frac{\partial \tilde{\mathbf{S}}^{(0)}}{\partial \tilde{\mathbf{C}}^{(0)}} \Delta \tilde{\mathbf{C}}^{(0)} : \delta \tilde{\mathbf{C}}^{(0)} + \underbrace{\frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{F}} \Delta \mathbf{F} : \delta \tilde{\mathbf{C}}^{(0)} + \frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{a}} \Delta \mathbf{a} : \delta \tilde{\mathbf{C}}^{(0)}}_{\text{active stress linearisation}} \right. \\ & + \frac{\partial \tilde{\mathbf{S}}^{(1)}}{\partial \tilde{\mathbf{C}}^{(1)}} \Delta \tilde{\mathbf{C}}^{(1)} : \delta \tilde{\mathbf{C}}^{(1)} + \frac{\partial \tilde{\mathbf{S}}^{(2)}}{\partial \tilde{\mathbf{C}}^{(2)}} \Delta \tilde{\mathbf{C}}^{(2)} : \delta \tilde{\mathbf{C}}^{(2)} \left. \right) dS dV \\ & + \frac{1}{2} \int_{\mathcal{B}} \int_{\mathcal{S}} \left(\tilde{\mathbf{S}}^{(0)} : \Delta \delta \tilde{\mathbf{C}}^{(0)} + \underbrace{\tilde{\mathbf{S}}_A^{(0)} : \Delta \delta \tilde{\mathbf{C}}^{(0)}}_{\text{active stress linearisation}} + \tilde{\mathbf{S}}^{(1)} : \Delta \delta \tilde{\mathbf{C}}^{(1)} + \tilde{\mathbf{S}}^{(2)} : \Delta \delta \tilde{\mathbf{C}}^{(2)} \right) dS dV = 0, \quad (6.13) \end{aligned}$$

where we have underlined the additions made to the original weak formulation. Because the active stress is given in terms of the deformation gradient and the micro-director, the stress is derived with respect to \mathbf{F} and \mathbf{a} . Apart from the three passive elasticity tensors defined in Equations (5.43) through (5.45), we now also have two additional constitutive tensors, $\frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{F}}$ and $\frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{a}}$.

6.2.2 Evaluation of the Active Constitutive Tensors

In this section we derive the fourth and fifth constitutive tensor. Using Equation (6.11) and applying the product rule, we find the fourth constitutive tensor to be,

$$\begin{aligned} \mathbf{C}^{(3)} = \frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{F}} &= \frac{\partial J}{\partial \mathbf{F}} \mathbf{F}^{-1} T_A(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}}) \mathbf{F}^{-T} + J \frac{\partial \mathbf{F}^{-1}}{\partial \mathbf{F}} T_A(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}}) \mathbf{F}^{-T} + \\ & J \mathbf{F}^{-1} \frac{\partial T_A}{\partial \mathbf{F}}(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}}) \mathbf{F}^{-T} + J \mathbf{F}^{-1} T_A(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}}) \frac{\partial \mathbf{F}^{-T}}{\partial \mathbf{F}}. \quad (6.14) \end{aligned}$$

Similarly the fifth constitutive tensor is given as

$$\mathbf{C}^{(4)} = \frac{\partial \tilde{\mathbf{S}}_A^{(0)}}{\partial \mathbf{a}} = J\mathbf{F}^{-1}T_A\left(\frac{\partial \hat{\mathbf{a}}}{\partial \mathbf{a}} \otimes \hat{\mathbf{a}}\right)\mathbf{F}^{-T} + J\mathbf{F}^{-1}T_A(\hat{\mathbf{a}} \otimes \frac{\partial \hat{\mathbf{a}}}{\partial \mathbf{a}})\mathbf{F}^{-T} + J\mathbf{F}^{-1}\frac{\partial T_A}{\partial \mathbf{a}}(\hat{\mathbf{a}} \otimes \hat{\mathbf{a}})\mathbf{F}^{-T}. \quad (6.15)$$

The constitutive tensors in Equations (6.14) and (6.15) can be expanded with the following useful identities,

$$\frac{\partial J}{\partial \mathbf{F}} = \frac{\partial \det \mathbf{F}}{\partial \mathbf{F}} = J\mathbf{F}^{-T}, \quad (6.16)$$

$$\frac{\partial \mathbf{F}^{-1}}{\partial \mathbf{F}} = -\mathbf{F}^{-T} \otimes \mathbf{F}^{-1}. \quad (6.17)$$

Since the active tension T_A is a function of the sarcomere length, its derivative with respect to the deformation gradient is given as follows,

$$\frac{\partial T_A}{\partial \mathbf{F}} = \frac{\partial T_A}{\partial l} \frac{\partial l}{\partial \mathbf{F}}. \quad (6.18)$$

Using the definition of the active tension in Equation (6.2), we find that

$$\frac{\partial T_A}{\partial l} = T_{max} \frac{ECa_{50}^2 e^{B(l-l_0)} B}{(e^{B(l-l_0)} - 1)(Ca_0^2 + ECa_{50}^2)^2} C_t - T_{max} \frac{Ca_0^2}{Ca_0^2 + ECa_{50}^2} \frac{1}{2} \sin \omega \frac{\partial \omega}{\partial l}. \quad (6.19)$$

The derivative of the sarcomere length with respect to the deformation gradient in Equation (6.18) is easiest to evaluate in index notation. From Equation (6.9) the sarcomere length in index notation is

$$l = \frac{l_R}{\sqrt{F_{kj}^{-1} F_{kl}^{-1} \hat{a}_j \hat{a}_l}}. \quad (6.20)$$

Therefore,

$$\begin{aligned} \frac{\partial l}{\partial F_{mn}} &= \frac{\partial}{\partial F_{mn}} \left(\frac{l_R}{\sqrt{F_{sp}^{-1} F_{st}^{-1} \hat{a}_p \hat{a}_t}} \right) \\ &= -\frac{1}{2} \frac{l_R}{(F_{sp}^{-1} F_{st}^{-1} \hat{a}_p \hat{a}_t)^{\frac{3}{2}}} \left(\frac{\partial F_{sp}^{-1}}{\partial F_{mn}} F_{st}^{-1} \hat{a}_p \hat{a}_t + F_{sp}^{-1} \frac{\partial F_{st}^{-1}}{\partial F_{mn}} \hat{a}_p \hat{a}_t \right) \\ &= -\frac{1}{2} \frac{l_R}{(F_{sp}^{-1} F_{st}^{-1} \hat{a}_p \hat{a}_t)^{\frac{3}{2}}} \left(-F_{sm}^{-1} F_{np}^{-1} F_{st}^{-1} \hat{a}_p \hat{a}_t - F_{sp}^{-1} F_{sm}^{-1} F_{nt}^{-1} \hat{a}_p \hat{a}_t \right), \end{aligned} \quad (6.21)$$

where we have used the rule $\frac{\partial \mathbf{F}^{-1}}{\partial \mathbf{F}} = -\mathbf{F}^{-T} \otimes \mathbf{F}^{-1}$. Similarly the derivative of the active tension with respect to the director is given as

$$\frac{\partial T_A}{\partial \mathbf{a}} = \frac{\partial T_A}{\partial l} \frac{\partial l}{\partial \hat{\mathbf{a}}} \frac{\partial \hat{\mathbf{a}}}{\partial \mathbf{a}}. \quad (6.22)$$

Again we evaluate derivative of the sarcomere length with respect to the unit vector $\hat{\mathbf{a}}$ in index notation,

$$\begin{aligned}
\frac{\partial l}{\partial \hat{a}_r} &= \frac{\partial}{\partial \hat{a}_r} \left(\frac{l_R}{\sqrt{F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \hat{a}_w}} \right) \\
&= -\frac{1}{2} \frac{l_R}{(F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \hat{a}_w)^{\frac{3}{2}}} \frac{\partial}{\partial \hat{a}_r} (F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \hat{a}_w) \\
&= -\frac{1}{2} \frac{l_R}{(F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \hat{a}_w)^{\frac{3}{2}}} (F_{qu}^{-1} F_{qw}^{-1} \delta_{ur} \hat{a}_w + F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \delta_{wr}) \\
&= -\frac{1}{2} \frac{l_R}{(F_{qu}^{-1} F_{qw}^{-1} \hat{a}_u \hat{a}_w)^{\frac{3}{2}}} (F_{qr}^{-1} F_{qw}^{-1} \hat{a}_w + F_{qu}^{-1} F_{qr}^{-1} \hat{a}_u). \tag{6.23}
\end{aligned}$$

Lastly, the derivative of the unit vector with respect to the director is given as,

$$\begin{aligned}
\frac{\partial \hat{\mathbf{a}}}{\partial \mathbf{a}} &= \frac{\partial}{\partial \mathbf{a}} \left(\frac{\mathbf{a}}{\|\mathbf{a}\|} \right) \\
&= \frac{\partial}{\partial \mathbf{a}} \left(\frac{\mathbf{a}}{\sqrt{\mathbf{a} \cdot \mathbf{a}}} \right). \tag{6.24}
\end{aligned}$$

In index form we have,

$$\begin{aligned}
\frac{\partial \hat{a}_k}{\partial a_s} &= \frac{\partial}{\partial a_s} \left(\frac{a_k}{\sqrt{a_m a_m}} \right) \\
&= \frac{\delta_{ks} \sqrt{a_m a_m} - a_k \frac{1}{2} \frac{1}{\sqrt{a_m a_m}} (\delta_{sm} a_m + a_m \delta_{sm})}{a_m a_m} \\
&= \frac{\delta_{ks} \sqrt{a_m a_m} - a_k \frac{1}{2} \frac{1}{\sqrt{a_m a_m}} (2a_s)}{a_m a_m} \\
&= \frac{\delta_{ks} \sqrt{a_m a_m} - \frac{1}{\sqrt{a_m a_m}} a_k a_s}{a_m a_m}. \tag{6.25}
\end{aligned}$$

The expressions in Equations (6.16) through (6.25) are substituted back into Equations (6.14) and (6.15) to obtain the full expressions for the active constitutive tensors. As discussed in Chapters 3 and 5, to obtain the macro-displacement and the micro-deformation we discretise the weak formulation and solve it using an iterative scheme such as the Newton Raphson method.

6.3 Summary

In this chapter we described an active-stress model for the tension generated when cardiac tissue contracts. The proposed model is based on the elastance model presented in [51]. The biggest difference between the elastance model in [51] and the model presented in Section 6.1 is the formulation of the sarcomere length. Here the sarcomere length is expressed in terms of the director and not in terms of the initial fibre direction. We also extended the weak formulation to include the active stress and derived the the active constitutive tensors.

Chapter 7

Simulation Procedure for a Full Heartbeat

In this chapter we first discuss the reconstruction of a patient-specific left ventricle from magnetic resonance images. The remainder of the chapter aims to present a methodology of how each of the four phases in a heartbeat can be simulated with the use of appropriate boundary and loading conditions.

7.1 Three-Dimensional Heart Anatomy and Tissue Structure

This study utilises a patient-specific geometry that was generated by the Computational Continuum Mechanics Group at the University of Cape Town, see [59]. The geometry was reconstructed using magnetic resonance imaging (MRI) scans of the heart provided by the Cape University Body Imaging Centre (CUBIC). The use of the scans was approved by the Ethics Committee of the Faculty of Health Sciences, see Appendix B. Below we summarise the general approach that is used to generate a realistic three-dimensional shape of the left ventricle.

MRI scans provide both short- and long-axis views of the heart. The stack of scans used in this study contains short-axis slices spaced 10 mm apart and long-axis slices that are spaced at 8.3 mm intervals. The first short-axis slice is located just below the mitral valve and the last slice is positioned near the apex.

To extract the geometric data from the scans, a process known as image segmentation is performed. In this research manual image segmentation of the scans is carried out using Synopsys' Simpleware software together with the image processing tool ScanIP. Because continuum balance laws are often formulated with respect to the reference configuration it is important to identify a reference (undeformed) state. In our model the reference configuration is taken to be at the start of diastolic filling since early diastole is regarded as being relatively stress-free [87]. The images corresponding to the start of diastolic filling are imported into Simpleware, cropped and segmented as illustrated in Figures 7.1 and 7.2. After completing the segmentation process, a three-dimensional geometry is created based on the segmented regions and smoothing algorithms are applied to the surface. The +NURBS module in Simpleware is utilised to generate a NURBS (non-uniform rational B-spline) surface, which is suitable for finite element analysis.

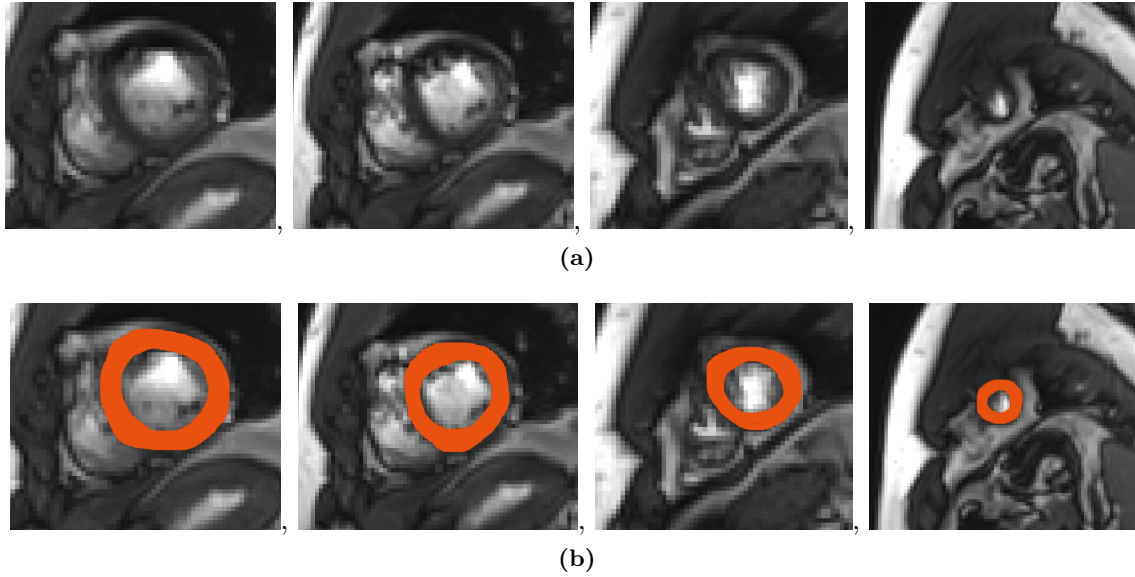


Figure 7.1: Examples of (a) magnetic resonance images from the CUBIC database and (b) MRI segmentation of the left ventricle.

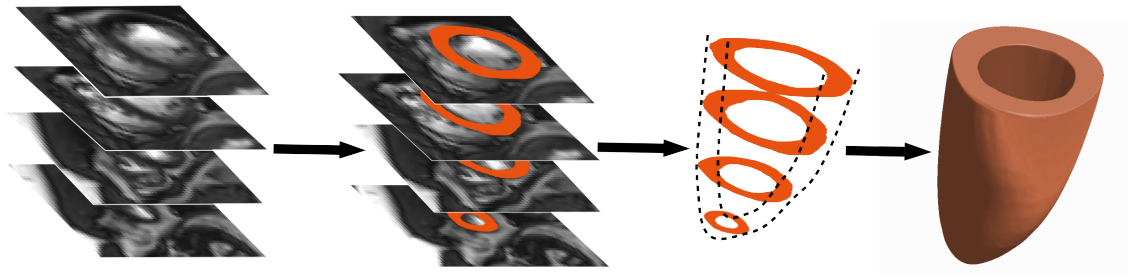


Figure 7.2: Diagram illustrating the process of creating a three-dimensional geometry of the left ventricle. The stack of magnetic resonance images is imported into the image processing software and the anatomical structure of the left ventricle is segmented. A surface is created that fit the segmented regions.

The NURBS surface is imported into the pre-and post processing software GiD. The geometry is discretised into four-node tetrahedral elements and a mesh convergence study is performed to determine an appropriate mesh density. The material properties, boundary conditions, fibre directions and sarcomere lengths are also assigned in GiD.

Boundary conditions should reflect the physiological conditions in which the heart operates [42]. Because the fibrous structure surrounding the valves is much stiffer than the ventricles, the vertical displacement of the base is taken to be zero. The outer edge of the base is constrained in the x-and y-directions using an elastic line force with a stiffness of 50 nN/mm. This constraint prevents excessive movement of the ventricle in the x-y plane. A pressure load is applied to the endocardial surface of the ventricle to simulate the pressure exerted by the blood. The pressure load is determined for each phase of the cardiac cycle, as discussed in the following sections. The load exerted by the pericardium and surrounding tissue is assumed to be negligible and therefore the pressure on the epicardial surface is taken to be zero.

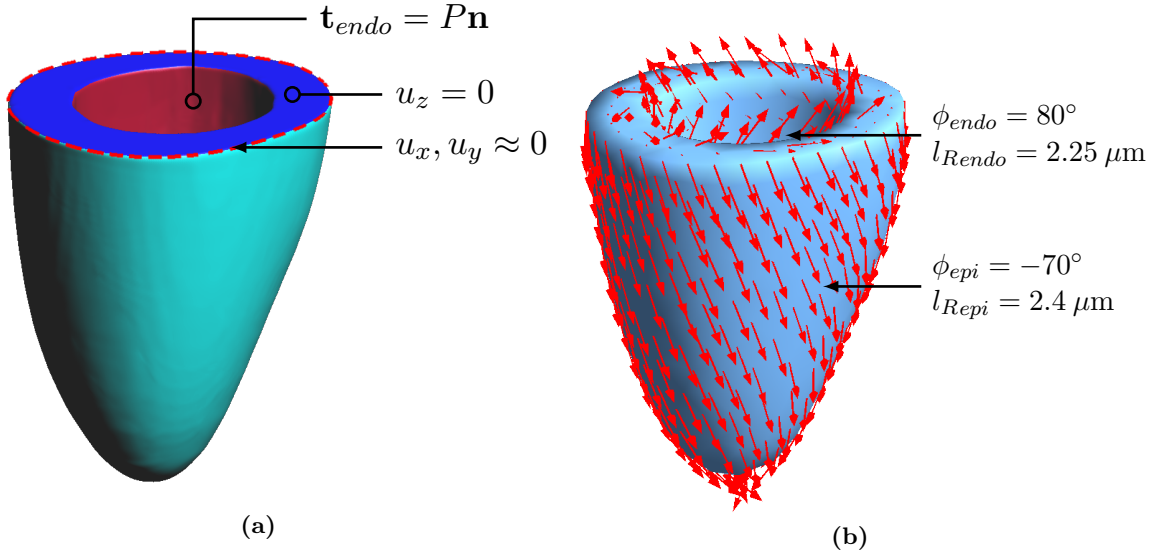


Figure 7.3: Ventricular geometry with (a) the Dirichlet and Neumann boundary conditions and (b) the fibre direction and sarcomere lengths. The vectors indicate the fibre direction in the reference configuration.

The direction of the cardiac fibres in the ventricle is chosen in accordance with [138]. The fibre angles at the epicardial and endocardial surfaces are taken to be -70° and 80° respectively, as depicted in Figure 7.3(b). The transmural variation of the fibre direction is obtained by interpolating the fibre angle through the ventricular wall using the algorithm developed by Wong and Kuhl [138]. Similarly, the resting sarcomere lengths are assigned such that it varies from $2.25 \mu\text{m}$ at the endocardium to $2.4 \mu\text{m}$ at the epicardium.

7.2 Modelling the Cardiac Cycle

As discussed in Chapter 2, the cardiac cycle consists of four phases. Accordingly, the simulation is also divided into four different phases. We follow a similar approach to that used by Skatulla and Sansour [114] of which an outline is given below.

7.2.1 Diastolic Filling

During diastolic filling, blood flows into the left ventricle from its adjoining atrium. As the ventricle fills with blood, its volume as well as the pressure acting on the endocardial surface increase. The ventricle is in a relaxed state since the cardiac fibres are not electrically activated. As a result, the only stresses present in the ventricular wall are the passive stresses,

$$\tilde{\mathbf{S}}_{total} = \tilde{\mathbf{S}}_P. \quad (7.1)$$

The passive stress component $\tilde{\mathbf{S}}_P$ is obtained from the micromorphic model discussed in Chapter 5. A Neumann boundary condition in the form of a pressure load P^{ED} is applied to the endocardial surface of the left ventricle. This pressure load simulates the inflow of blood and is taken to be 1.6 kPa [69]. Because of the high degree of non-linearity, the external load, in this case the end-diastolic

pressure P^{ED} , is applied incrementally,

$$P_n = \zeta_n P^{ED} \quad \text{with} \quad 0 \leq \zeta_n \leq 1, \quad (7.2)$$

where ζ_n is the loading factor and n is the current loading step. The current loading factor is determined with a known loading factor increment,

$$\zeta_n = \zeta_{n-1} + \Delta\zeta_n, \quad (7.3)$$

where $\Delta\zeta_n$ is the loading factor increment. The displacement field and the micro-deformation are determined for each loading step. Using the displacement field, we determine the cavity volume for each loading step. In this way we obtain a set of pressure-volume data points that represents the end-diastolic pressure-volume relationship.

7.2.2 Isovolumetric Contraction

Once the cavity pressure in the ventricle reaches the end-diastolic pressure, the mitral closes and no inflow or outflow of blood is permitted. Since blood is a nearly-incompressible fluid, the cavity volume of the ventricle remains relatively constant. An electrical impulse results in ventricular contraction which in turn causes a significant rise in the cavity pressure. In addition to the passive stress, cardiac fibres now also generate their own tension. Therefore, the total ventricular wall stress is given by

$$\tilde{\mathbf{S}}_{total} = \tilde{\mathbf{S}}_P + \tilde{\mathbf{S}}_A, \quad (7.4)$$

where $\tilde{\mathbf{S}}_A$ is the active stress based on the active model discussed in Chapter 6. During isovolumetric contraction, the displacement field should be such that the cavity volume of the left ventricle remains constant ($\Delta V \approx 0$). The applied pressure load is determined such that the volume constraint is satisfied. Instead of having a known loading factor increment $\Delta\zeta_n$ (as in the case of diastolic filling), the increment is now unknown and needs to be solved. A displacement control approach is used to determine the loading factor increment and consequently also the pressure load. The displacement control approach and its extension to structures with cavities are discussed in the more detail by Skatulla and Sansour [114].

The end of isovolumetric contraction is reached when the left ventricular pressure is equal to the pressure in the aorta which is taken to be 10 kPa [68, 69].

7.2.3 Ejection

Because of the high ventricular pressure at the end of isovolumetric contraction, the aortic valve opens and blood is allowed to flow into the aorta. As blood is ejected, the cavity volume of the ventricle decreases. We use a three-element Windkessel model to obtain the change in volume. The three-element model, described in Chapter 4, is repeated below,

$$\left(1 + \frac{R_a}{R_p}\right)I(t) + CR_a \frac{dI(t)}{dt} = \frac{P(t)}{R_p} + C \frac{dP(t)}{dt}, \quad (7.5)$$

where R_p and R_a are the peripheral resistance of the blood vessels and the resistance of the aorta. C represents the compliance of the elastic arteries.

The initial flow rate is taken to be zero, while the initial pressure is the same as the pressure at the end of isovolumetric contraction. If assuming a small time increment, the three-element Windkessel equation can be written as

$$\left(1 + \frac{R_a}{R_p}\right)I_n + CR_a \frac{I_n - I_{n-1}}{\Delta t} = \frac{P_n}{R_p} + C \frac{P_n - P_{n-1}}{\Delta t}, \quad (7.6)$$

where n is the current time step. The volume increment in terms of the flow rate is given by

$$\Delta V_n = -I_n \Delta t. \quad (7.7)$$

The negative sign is to ensure a negative volume change during the ejection phase [99]. Using the above equation and Equation (7.6), the volume increment in terms of the Windkessel parameters is found to be

$$\Delta V = \left(\frac{CR_0}{\Delta t} + \left(1 + \frac{R_0}{R}\right) \right)^{-1} \left(C \frac{P_n - P_{n-1}}{\Delta t} + \frac{P_n}{R} + CR_0 \frac{I_{n-1}}{\Delta t} \right). \quad (7.8)$$

Again the cavity control approach outlined in [114] is used to determine the pressure load on the endocardial surface. The ejection loading phase ends when the flow rate nears zero.

7.2.4 Isovolumetric Relaxation

After the blood has been ejected the aortic valve closes and the cavity pressure starts reducing while the cavity volume remains constant. Both passive and active stresses are still present, but the latter gradually decreases to zero. Similar to isovolumetric contraction, the cavity-volume change ΔV is set to zero. As soon as the ventricular pressure falls below the atrium pressure the mitral valve opens which signals the end of relaxation. Therefore the fourth and final phase ends when the ventricular pressure reaches that of the atrium which is taken to be 0.2 kPa.

7.3 Classical Continuum Model

To verify the micromorphic results, the model is compared to an existing classical model. Cardiac modelling performed in the Computational Continuum Mechanics Research Group led to the development of a fully-calibrated classical continuum model of the heart, see [99]. The material model used to describe the passive behaviour is given by

$$\psi = \frac{A}{2}(e^Q - 1) + A_{compr}(J \ln J - J + 1), \quad (7.9)$$

where

$$Q = b_{ff}E_{ff}^2 + b_{ss}E_{ss}^2 + b_{nn}E_{nn}^2 + b_{fs}(E_{fs}^2 + E_{sf}^2) + b_{fn}(E_{fn}^2 + E_{nf}^2) + b_{ns}(E_{ns}^2 + E_{sn}^2), \quad (7.10)$$

which is based on the model of Usyk et al [125]. As in the micromorphic case, the active behaviour of the cardiac tissue is described using the elastance model by Guccione et al. [51] and blood ejection with the three-element Windkessel model.

Chapter 8

Results and Discussions

Previous chapters provided the theoretical background needed to create a computational model of the left ventricle and outlined how the mechanical behaviour of the ventricle may be simulated. Here we discuss the results obtained from the modelling procedure detailed in Chapter 7. We first present the results of a mesh convergence study that was performed to determine an appropriate mesh density for the finite element model. Next we summarise the passive, active and Windkessel material parameters determined from shear experiments and pressure-volume data. Finally, the simulated ventricular mechanics is presented and we discuss how the results compare to clinical observations. Additionally, we compare the micromorphic formulation with a classical formulation and show that a micromorphic modelling approach can be employed successfully to model the behaviour of the ventricle.

8.1 Mesh Convergence Study

A mesh convergence study was conducted to determine an appropriate element size for the finite element model of the left ventricle. The ventricular geometry was first discretised into 479 four-node tetrahedral elements. With each subsequent analysis the mesh was refined to determine the effect of the element size. The most refined mesh consisted of 13 552 tetrahedral elements. Figure 8.1 shows examples of the different mesh densities used in the mesh convergence study.

In this initial analysis, the base of the left ventricle was constrained as discussed in Section 7.1 and a pressure load of 2.0 kPa was applied incrementally to the endocardial surface of the ventricle. The passive material parameters that describe the bulk material in Equation (5.41) were chosen based on a study performed by Usyk et al. [125] and are summarised in Table 8.1. Unfortunately, without additional material testing that report on the non-affine deformation of cardiac tissue, it will not be possible to uniquely identify the material parameters related to fibre material. Therefore these parameters, reported in Table 8.2, were chosen to ensure that a solution was obtained. For simplicity, the exponential scaling parameters (B_{8ff} , B_{8fs} , B_{8fn} and B_{4f}) were taken to be equal to 1.00, while values of the scaling parameters (A_{8ff} , A_{8fs} , A_{8fn} and A_{4f}) were assumed to be small to limit reorientation of the fibres.

The results from the mesh convergence study are depicted in Figures 8.2 and 8.3. For each simulation, the applied pressure is plotted against the simulated cavity volume in Figure 8.2. We see that the difference between successive simulations reduces with each refinement.

The relationship between the number of tetrahedral elements in the finite element mesh and the final chamber volume at 2.0 kPa is provided in Figure 8.3. The different geometry approximation of each mesh results in the slightly different starting volumes. We see that there is a significant difference between the chamber volumes of the first three simulations, while the difference between the results of the last three mesh refinements is much smaller. The curve clearly shows that the simulated volume converges to a solution as the mesh is refined.

Even though a denser mesh usually provides results that are closer to the actual solution, one has to also take into account the simulation run time when choosing an appropriate mesh. To ensure a balance between computation time and accuracy, a final mesh of 2017 finite elements was chosen for subsequent analyses.

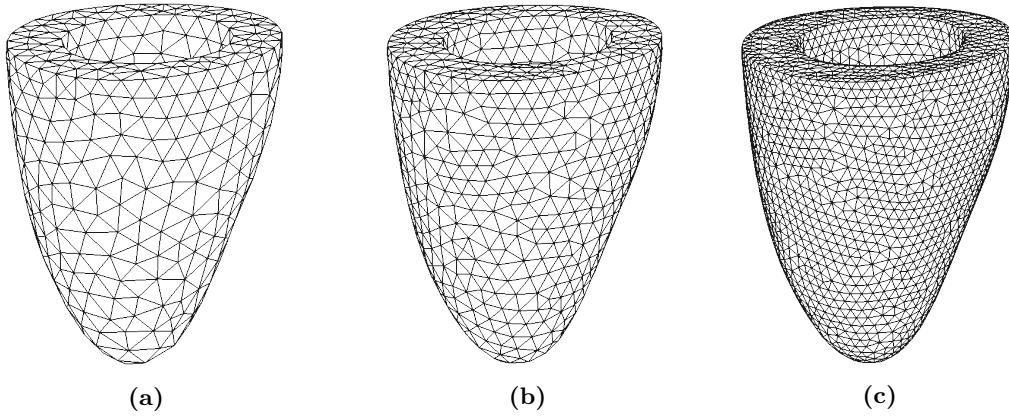


Figure 8.1: Examples of finite element geometries with different mesh densities consisting of (a) 836 elements, (b) 2017 elements and (c) 7246 elements.

Table 8.1: Values of the passive material parameters to describe the bulk material in the mesh convergence analysis, obtained from [125].

Parameter	Units	Initial value
A_0	kPa	0.88
B_0	—	1.0
A_{comp}	kPa	3.0
b_{ff}	—	6.0
b_{ss}	—	7.0
b_{nn}	—	3.0
b_{fs}	—	12.0
b_{fn}	—	3.0
b_{sn}	—	3.0

Table 8.2: Values of the passive material parameters to describe the fibre material in the mesh convergence analysis.

Parameter	Units	Initial value
A_{8ff}	kPa	0.25
B_{8ff}	—	1.0
A_{8fs}	kPa	0.10
B_{8fs}	—	1.0
A_{8fn}	kPa	0.10
B_{8fn}	—	1.0
A_{4f}	kPa	0.10
B_{4f}	—	1.0

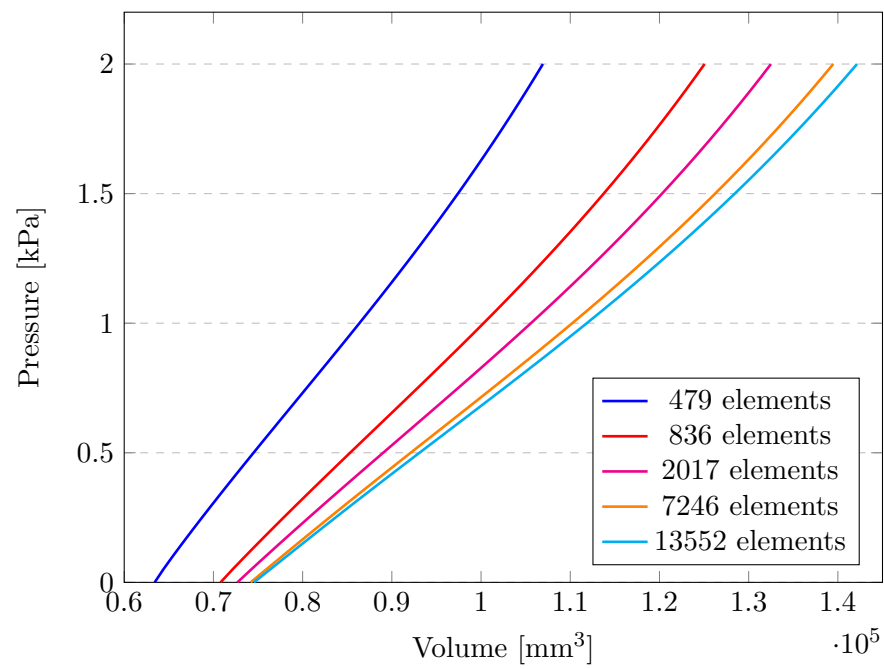


Figure 8.2: Simulated pressure-volume relationship using different mesh densities.

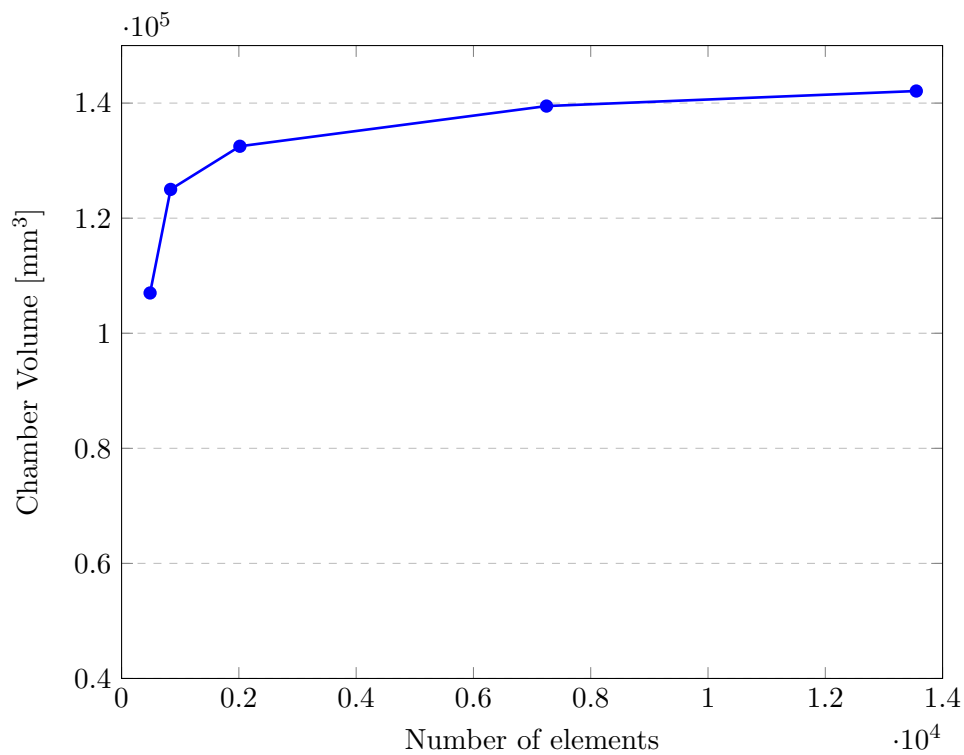


Figure 8.3: Relationship between the element density and the final volume illustrating the convergence of the finite element solution.

8.2 Material Parameters

Material parameters found in the passive, active and Windkessel models have to be chosen such that the models replicate the observed behaviour of cardiac tissue. To determine suitable values for the parameters, the mathematical models are fit to experimental data. The objective during the calibration process is to minimise the difference between the experimental results and the corresponding simulation results. The additive nature of the stress tensor allowed us to calibrate the passive and active models separately.

8.2.1 Parameter Estimation

To identify material parameters, the Computational Continuum Mechanics Research Group implemented a parameter estimation scheme. The scheme is based on the least squares curve fitting problem,

$$\min_{\alpha} J(\alpha) = \frac{1}{2} \sum_{i=1}^m (f_i(\alpha, x_i) - y_i(x_i))^2 \quad (8.1)$$

where J is the sum of the squared differences. In Equation (8.1), x_i is the independent variable, y_i is the experimental data point and m is the number of data points. The simulation result f_i depends on the material parameter(s) α and the independent variable x_i . The objective is to minimise sum of the square of the differences between the experimental data and the computational model. The parameter estimation scheme developed by the research group solves the curve fitting problem using the Levenberg-Marquardt algorithm, which is a combination of the steepest decent and the Gauss-Newton method. More detail regarding the implementation of this algorithm is found in [34, 99].

8.2.2 Passive Material Parameters

The micromorphic model in Equation (5.29) has 17 material parameters, 9 related to the bulk material and 8 related to the fibre material. The bulk material parameters, except for the scaling parameter A_0 and the incompressibility parameter A_{comp} , were calibrated with the use of shear data. The set of passive material parameters were previously determined by the Computational Continuum Mechanics Group using the Levenberg-Marquardt algorithm, and it is therefore not necessary to recalibrate the passive model. Below we still outline how the parameters were obtained.

The passive model was calibrated to fit the experimental shear data obtained by Sommer et al. [116]. Sommer et al. performed tri-axial experiments on cubes of human cardiac tissue. Cubes with dimensions $4 \times 4 \times 4$ mm and sides aligned with the material directions of cardiac tissue were taken from the ventricular wall. The cubes of tissue were deformed by constraining one side of the sample and translating the opposite surface with a specially designed shear testing device. A shear of 50% was obtained in this way. The shear tests were performed in different orientations to obtain the different shear deformation modes, see Figure 4.3 in Chapter 4.

To determine the material parameters, the shear experiments were reproduced computationally. The simulation consisted of a $4 \times 4 \times 4$ mm cube. The degrees of freedom in all three directions on the bottom face were fixed. To ensure pure shear, the opposite face was prevented from moving in the vertical direction while an in-plane displacement of 2 mm was prescribed to achieve a strain of 50%, as shown in Figure 8.4.

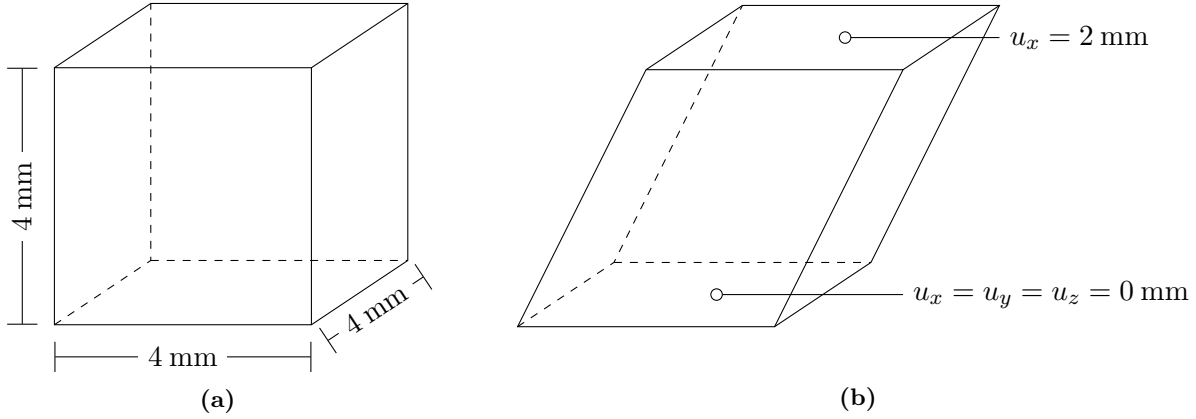


Figure 8.4: Replication of the shear experiments showing (a) the dimensions of the undeformed cube and (b) the boundary conditions to produce shear deformation.

Table 8.3: Material parameters for the bulk material in the passive model. The parameters were determined by fitting the model to data from shear experiments and the calibration was performed using the Levenberg-Marquardt method.

Model Parameter		Initial value and source	Value after calibration
Bulk	B_0	1.0 [125]	1.0
	b_{ff}	6.0 [125]	21.03
	b_{ss}	7.0 [125]	8.02
	b_{nn}	3.0 [125]	0.22
	b_{fs}	12.0 [125]	10.68
	b_{fn}	3.0 [125]	8.12
	b_{sn}	3.0 [125]	9.71

As noted in Chapter 2, the direction of the cardiac fibres changes gradually across the heart wall. To include this variation in the simulation, the fibre direction was specified such that it varied across the cube. The one side was chosen to have a fibre angle of $+30^\circ$ which varied to -30° at the opposite face [99]. All together 6 computational experiments were required to represent the 6 deformation modes. This was achieved by changing the planes on which the displacement boundary conditions were applied.

The final values for the bulk material parameters are reported in Table 8.3. Only after more detailed experimental data of cardiac tissue deformation becomes available will it be possible to determine the parameters of the fibre material. In the following sections, the same material parameters for the fibre contribution were used as in the mesh convergence study, see Table 8.2.

We used the end-diastolic pressure-volume relation to obtain suitable values for the incompressibility parameter A_{comp} and the scaling parameter A_0 . To determine these values, the diastolic filling phase of the cardiac cycle was reproduced as discussed in Section 7.2.1. The end-diastolic pressure of 1.6 kPa was applied to the endocardial surface of the left ventricle and from the magnetic resonance images we know that the end-diastolic volume is equal to 146 ml. Since the starting volume is 73 ml, the ejection fraction is equal to 50%, which is consistent with clinical measurements.

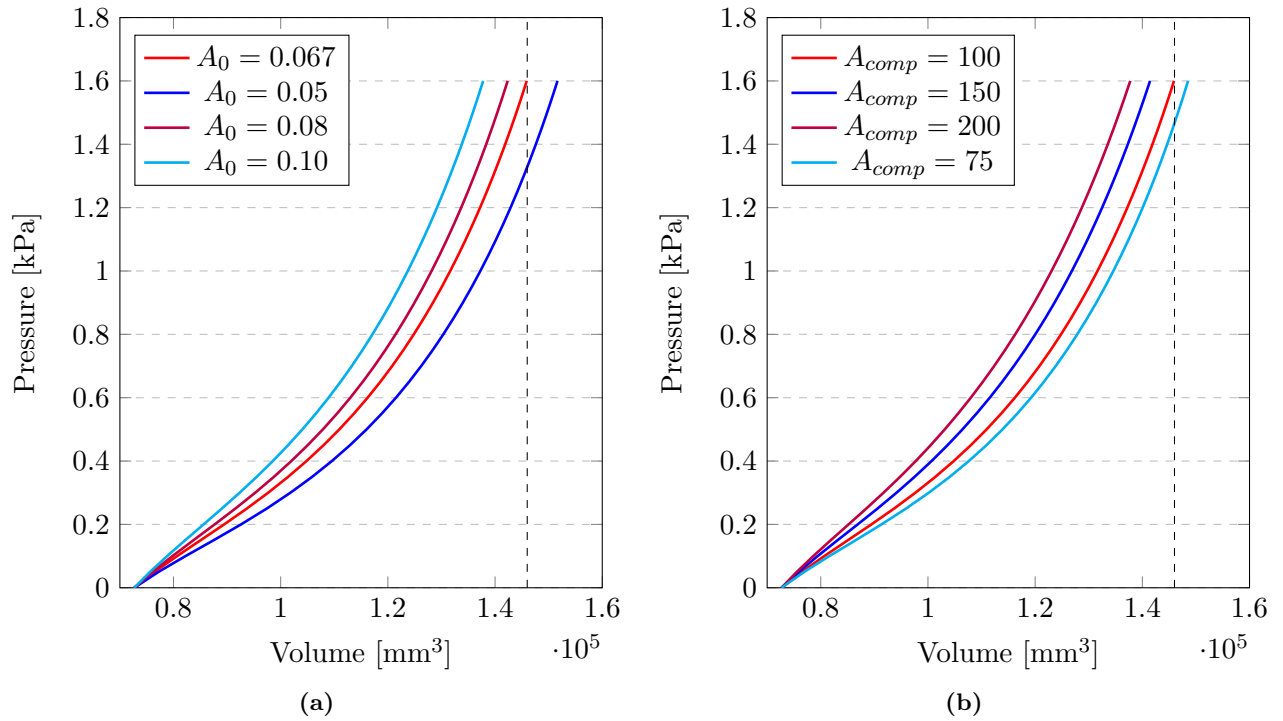


Figure 8.5: The effect on the material response when using different values for (a) the scaling parameter A_0 and (b) the incompressibility parameter A_{comp} . The pressure-volume curves show that increasing these parameter makes the material behave stiffer. The dashed line indicates the target volume.

The parameters A_{comp} and A_0 were chosen in order to obtain the target volume of 146 ml at the end of diastolic filling. Figure 8.5 shows the effect of these two parameters on the end-diastolic pressure-volume relation. The values of $A_{comp} = 0.067$ kPa and $A_0 = 100$ kPa were found to be most appropriate.

8.2.3 Active and Windkessel Material Parameters

Calibration of the passive model was followed by the calibration of the systolic parameters. The active and Windkessel model parameters, see Equations (6.2) and (7.8), are presented in Table 8.4. The initial material parameters for the active model were chosen in accordance with [51], while the initial Windkessel parameters were based on the study of [68].

The pressure-volume data point at the end of ejection was used to obtain suitable values for the active and Windkessel parameters. The model parameters were chosen such that the simulated volume at the end of ejection matched the clinically measured end-ejection volume. Since the ventricle relaxes at a constant volume, the volume at the start of diastolic filling corresponds to the volume at the end of ejection. From magnetic resonance imaging data we know the initial volume and therefore also the end-systolic volume is 73 ml. A set of suitable material parameters was determined such that an end-ejection volume of 73 ml was obtained. The final values are summarised in Table 8.4.

Table 8.4: Active and Windkessel material constants determined such that the end-systolic volume is equal to 72 ml.

Model Parameter		Units	Initial value and source	Value after calibration
active	T_{max}	kPa	135.7 [51]	195
	C_{a0}	μM	4.35 [51]	6.35
	C_{max}	μM	4.35 [51]	6.35
	b	s	-1.429 [51]	-1.429
	B	μm^{-1}	4.75 [51]	4.75
	l_0	μm	1.58 [51]	1.45
	t_0	s	0.1 [51]	0.25
Windkessel	R_a	$\text{kPa}\cdot\text{s}/\text{mm}^3$	1.50×10^{-5} [68]	1.50×10^{-4}
	R_p	$\text{kPa}\cdot\text{s}/\text{mm}^3$	1.20×10^{-6} [68]	1.47×10^{-5}
	C	mm^3/kPa	1000 [68]	4500

Table 8.5: Material parameters of the classical formulation, from [99].

Parameter	Units	Initial value
A_0	kPa	0.108
A_{comp}	kPa	100
b_{ff}	—	23.92
b_{ss}	—	5.89
b_{nn}	—	0.59
b_{fs}	—	12.74
b_{fn}	—	10.19
b_{sn}	—	11.7

Table 8.6: Active and Windkessel material parameters used in the micromorphic and classical model.

Model Parameter		Units	Micromorphic	Classic
active	T_{max}	kPa	195	140
	C_{a0}	μM	6.35	4.35
	C_{max}	μM	6.35	4.35
	b	s	-1.429	-1.429
	B	μm^{-1}	4.75	4.75
	l_0	μm	1.45	1.58
	t_0	s	0.25	0.22
Windk	R_a	$\text{kPa}\cdot\text{s}/\text{mm}^3$	1.50×10^{-4}	1.08×10^{-4}
	R_p	$\text{kPa}\cdot\text{s}/\text{mm}^3$	1.47×10^{-5}	1.47×10^{-5}
	C	mm^3/kPa	4500	4500

8.3 Left Ventricular Mechanics

In this section we present the main findings of the computational study of the left ventricle. We compare the results obtained from the analysis that uses the micromorphic continuum description to that using a classical continuum formulation.

8.3.1 Simulated Pressure-Volume Relationship

The pressure-volume curves of the two continuum descriptions are presented in Figure 8.6. The material parameters of the classical model in Equation (7.9) are given in Table 8.5. The orthotropic material parameters, b_{ij} were obtained during a previous study conducted by the Computational Continuum Mechanics Group, see [99]. As in the micromorphic case, the scaling parameter A and the incompressibility parameter A_{comp} were chosen to obtain an end-diastolic volume of 146 ml. Table 8.6 compares the Windkessel and active model parameters of the classical and micromorphic formulations. These parameters were chosen to ensure the end-ejection volume was equal to 73 ml.

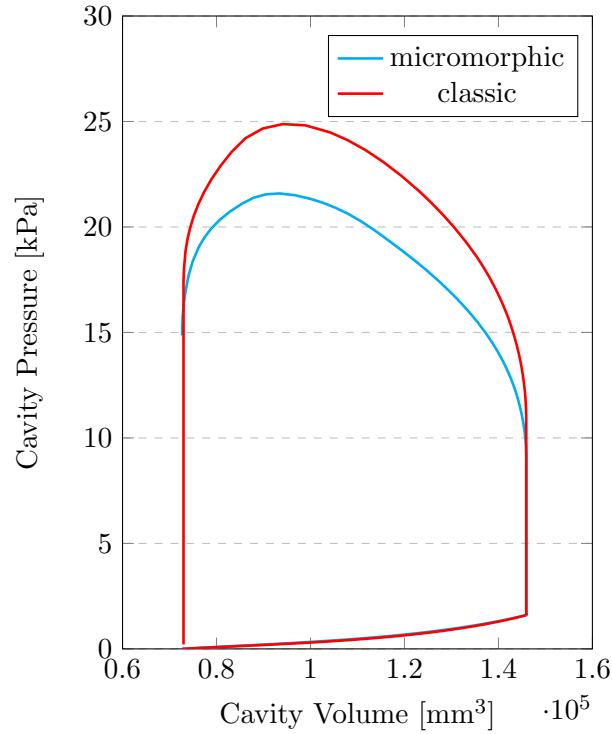


Figure 8.6: Simulated pressure volume curves using a micromorphic formulation and a classical continuum formulation.

Even though the two curves do not match exactly, they are similar in shape. The discrepancy between the curves is expected since the micromorphic formulation includes additional features of the material microstructure. Figure 8.6 shows that the overall passive behaviour during diastolic filling is very similar for the two formulations. However, comparison of the scaling parameters A and A_0 , reveals that the micromorphic material behaves stiffer than the classical material. A smaller scaling parameter had to be used in the micromorphic case to achieve the same end-diastolic volume of 146 ml. A similar observation was made in a previous study where the Cosserat continuum was used to model cardiac tissue [110].

During ejection the cavity pressure increases, reaches a peak value and subsequently decreases. The maximum pressure in the micromorphic case is 21.6 kPa, which is lower than the peak pressure observed in the classical case. The pressure values at the end of ejection are 15 kPa and 17 kPa for the micromorphic and the classical model respectively. Clearly, the inclusion of the micro-space in the micromorphic model has a greater effect on the ventricular behaviour during systole. One reason is the complex deformation experienced by the left ventricle during contraction and ejection, as discussed in Section 8.3.3.

It is important to note that neither of these models were calibrated to patient-specific pressure data, but rather to average values reported in the literature. This is because of the invasive nature of the procedure associated with obtaining pressure data. Even though the model is not truly patient-specific, it still provides qualitative information that allows us to investigate the mechanics of the ventricle. Furthermore, the micromorphic model deals well with irregular geometries and also captures clinically observed behaviour as discussed the following sections.

8.3.2 Diastolic expansion

Figure 8.7 presents the development of the effective strain during the cardiac cycle, while Figure 8.8 shows the spatial variation of the strain on cross-sectional views of the ventricle. In both figures, the effective strain, defined as

$$\tilde{E}_{eff} = \sqrt{\frac{3}{2} \tilde{\mathbf{E}} : \tilde{\mathbf{E}}}, \quad (8.2)$$

is superimposed on the deformed geometry of the left ventricle.

Because the start of diastolic filling is assumed to be the reference configuration, the ventricle has yet to experience any deformation and the spatial distribution of the strain is zero across the ventricle. Comparison of the deformed geometries in first two panels of Figure 8.7 shows that the left ventricle expands during diastolic filling. Additionally, the apex moves down causing the ventricle to experience lengthening along its long axis. The cross sectional views show that the myocardial wall of the ventricle changes during a heartbeat. The relative sliding of myocardial sheets allows the changes in the ventricular geometry as well as changes in the wall thickness to take place [74].

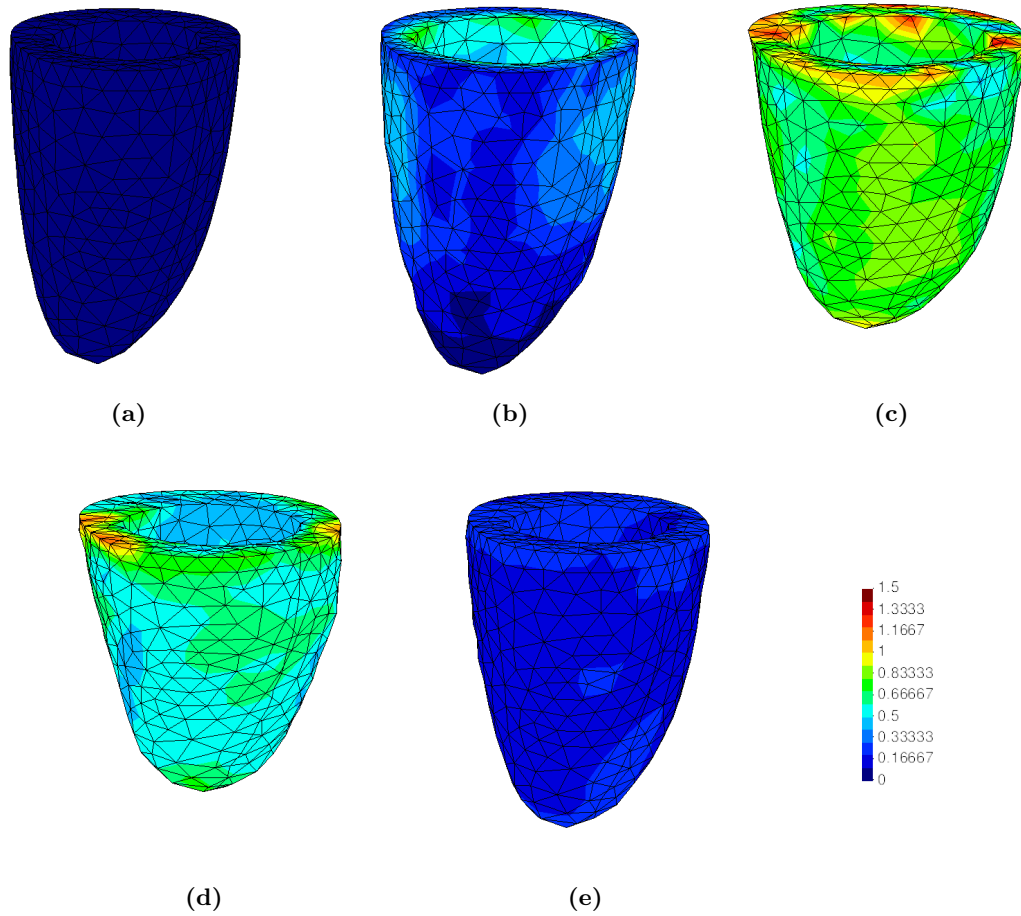


Figure 8.7: The computed effective strain using a micromorphic model, superimposed on the deformed ventricle at the (a) start of diastolic filling, (b) end of diastole, (c) end of isovolumetric contraction, (d) end of ejection and (e) end of isovolumetric relaxation.

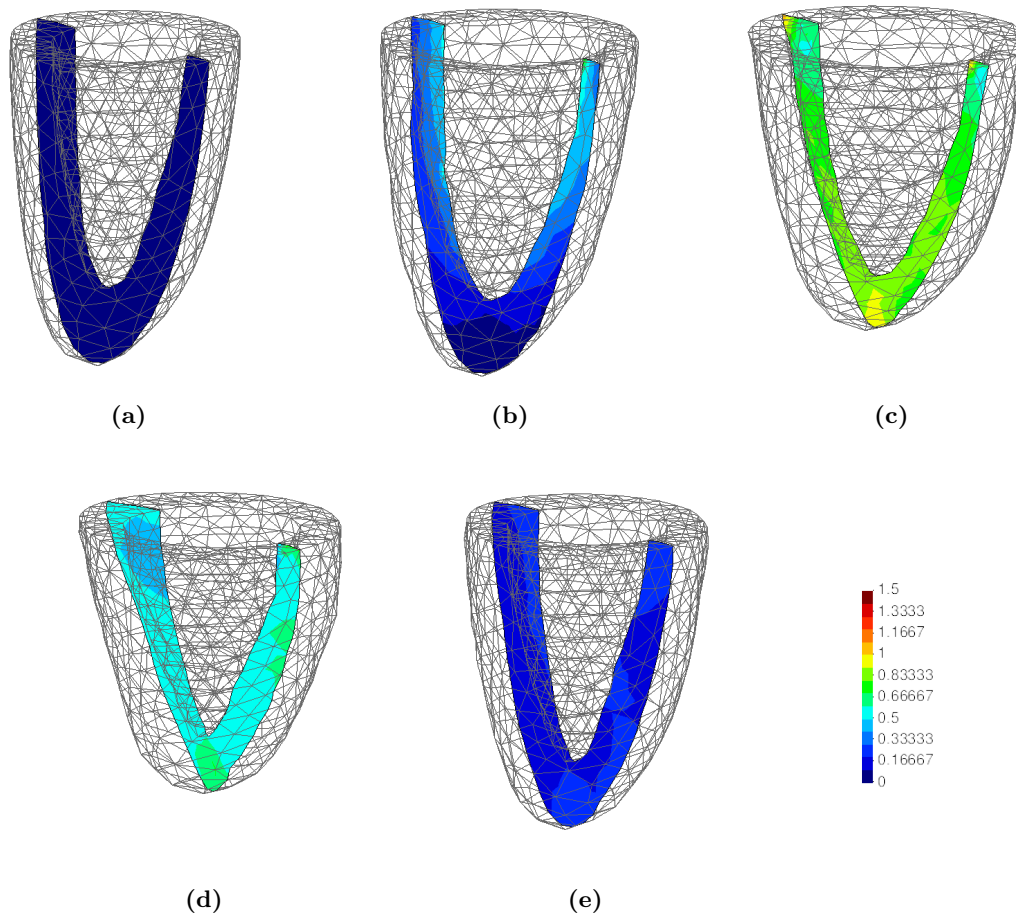


Figure 8.8: The computed strain contours on cross-sectional views of the deformed ventricle at the (a) start of diastolic filling, (b) end of diastole, (c) end of isovolumetric contraction, (d) end of ejection and (e) end of isovolumetric relaxation.

8.3.3 Ventricular Twist

During contraction, we observe a shortening of the ventricle along its centroidal axis. This results in a slightly spherical shape of the ventricle at the end of isovolumetric contraction, as seen in Figure 8.7(c). The apex does not experience significant motion during ejection in spite of the rise in the chamber pressure, compare for example Figures 8.7 (c) and (d) . This was also observed by Göktepe et al. [47] and the behaviour is expected since biological tissue behaves stiffer at large strains as illustrated by the exponential Fung-type strain energy function.

As the ventricle contracts, it undergoes a twisting motion in order to raise the blood pressure. Figure 8.9 provides short-axis views of the ventricle at the end of diastolic filling and mid-systole. The figure depicts the displacement of the ventricle as viewed from the apex. It clearly illustrates the overall radial motion during diastolic filling as well as the twisting motion during contraction. The radial motion continues into the ejection phase. At the beginning of relaxation, the ventricle starts to untwist, i.e. it starts to rotate in the opposite direction.

The observed twisting is in accordance with clinical findings. The twisting motion of the ventricle

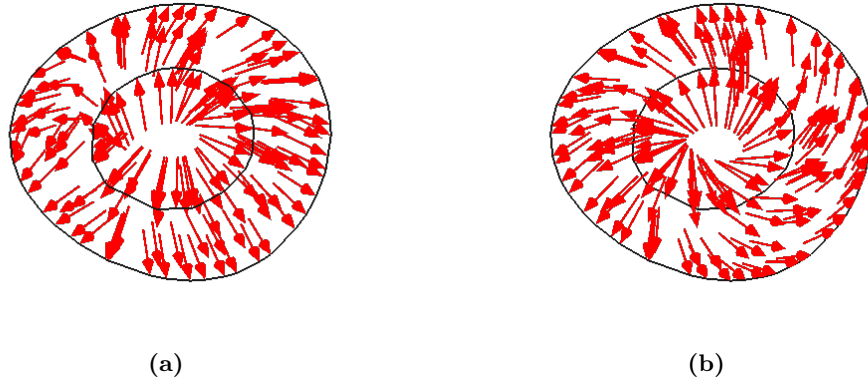


Figure 8.9: The computed displacement of the ventricle as viewed from the base during (a) filling and (b) contraction. The displacement is illustrated as vectors to visualise the rotational motion of the ventricle.

occurs as a result of the tissue layout in the myocardium [58, 83]. Fibres in the subepicardial region form a left handed helix while fibres in the subendocardium form a left handed helix. Without the twisting motion, the ejection fraction would be much less, around 15-20% instead of the actual 50-70% [83].

At the end of isovolumetric relaxation, the ventricle is in a state similar to the initial configuration but with some residual stains. Residual strains are in line with clinical observations since the heart is always in a state of stress. The reason is that pressure exerted by blood during the heart cycle is always larger than zero. In this study we treated the start of diastolic filling as the unloaded (or stress-free) state. However, this is somewhat inconsistent with clinical observations. The model may be improved by using the final state as the starting point for a new simulation. This will allow for a more accurate simulation, since residual stresses will then automatically be included.

8.3.4 Stresses in the Left Ventricle

Figure 8.10 provides the spatial variation of the effective stress computed at different points in time in the cardiac cycle. Presented in Figure 8.11 are the contour plots of the stress on cross-sectional views of the left ventricle. The effective stress is given by,

$$\tilde{\sigma}_{eff} = \sqrt{\frac{3}{2} \tilde{\sigma} : \tilde{\sigma}}, \quad (8.3)$$

where $\tilde{\sigma}$ is the generalised Cauchy stress. As seen in Figure 8.10 (a), stress values at the end of diastolic filling are quite low and the distribution is fairly uniform. The reason is that the pressure at the end of filling is still quite low compared to the pressures throughout the rest of the cardiac cycle.

The wall stresses are highest during contraction and ejection, see Figures 8.10 (b) and (c). The stresses tend to be higher near the endocardium. This corresponds with the study performed by Humphrey and Yin [60], who found that the peak stresses occur in the inner third of the ventricular wall.

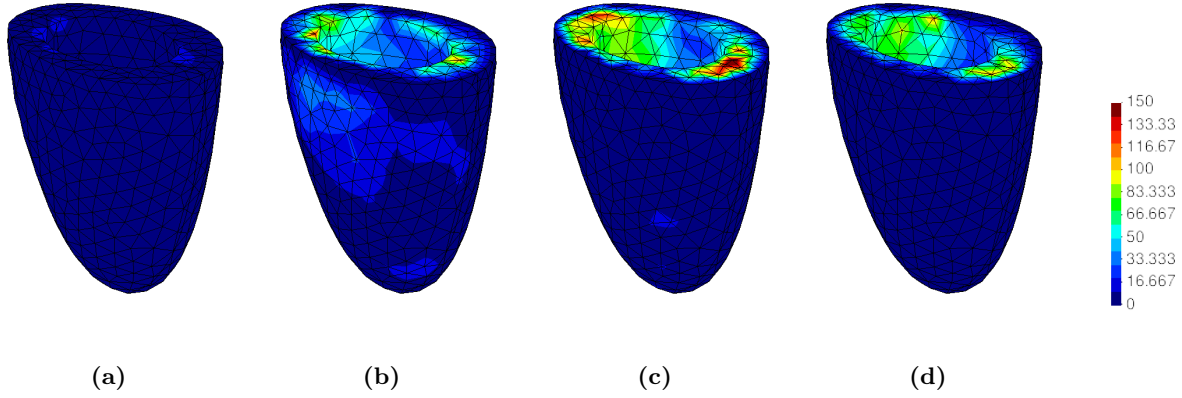


Figure 8.10: The computed effective stress using a micromorphic approach, superimposed on the deformed ventricle at (a) end of diastolic filling, (b) end of isovolumetric contraction, (c) mid ejection and (e) end of ejection.

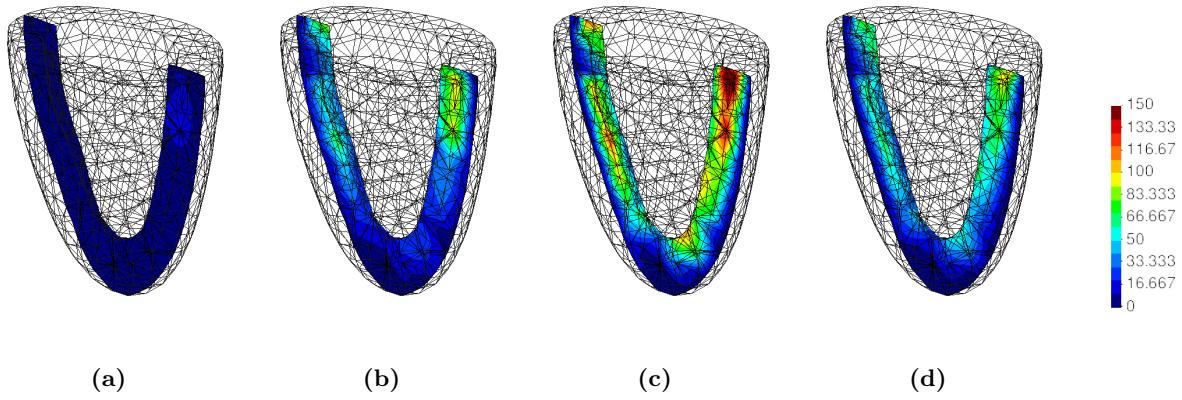


Figure 8.11: The computed effective stress contours on cross-sections, superimposed on the deformed ventricle at (a) end of diastolic filling, (b) end of isovolumetric contraction, (c) mid ejection and (e) end of ejection.

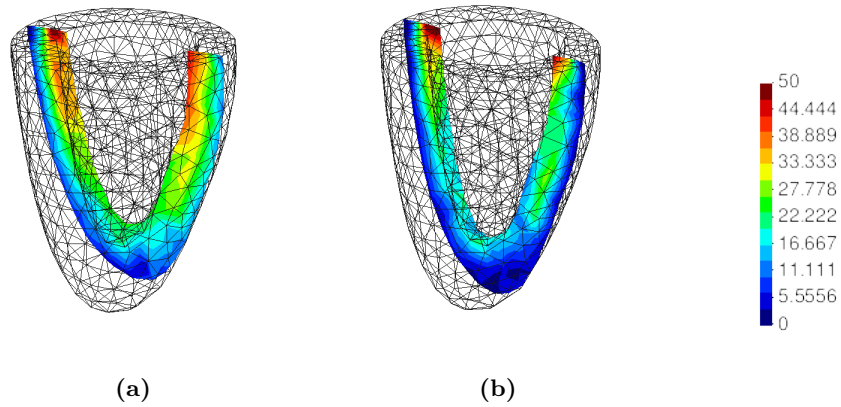


Figure 8.12: The computed active stress at the end of isovolumetric contraction using (a) a classic formulation and (b) a micromorphic formulation.

During systole the active tension is the largest component of the total stress. Figure 8.12 compares the active tension distributions obtained using the micromorphic and classical continuum formulations. The disparity between the two formulations is a result of the different approaches used to calculate the sarcomere length, see Equations (4.17) and (6.9). The sarcomere length in the micromorphic continuum is generally lower and as a consequence the active tension in the micromorphic case is lower than in the classic case. Figure 8.13 presents the evolution of the active tension during the cardiac cycle. In general the active tension is highest at the base of the ventricle and lowest near the apex. It is important to note that we have assumed that all fibres contract at the same time. This assumption is likely to reduce the twisting and shape change of the ventricle [8].

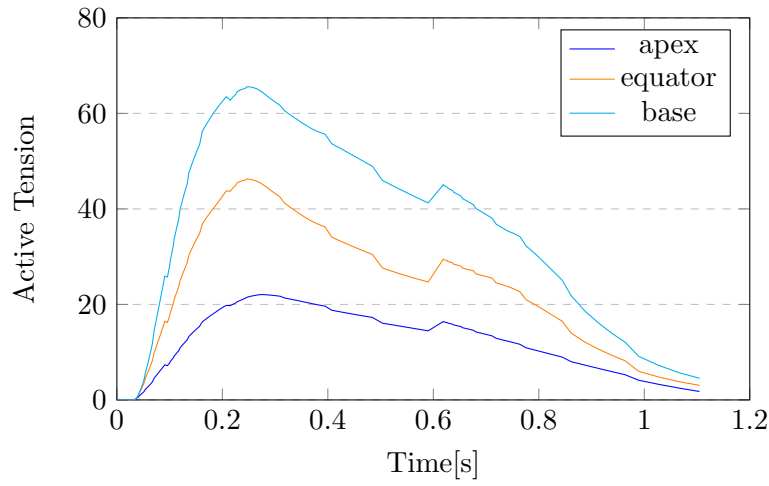


Figure 8.13: The change in active stress over time near the apex, equatorial region and base of the ventricle.

8.4 Summary of Computational Results

In this chapter we presented a mesh convergence study and discussed how suitable material parameters were obtained for the passive, active and Windkessel models. Furthermore, we presented the results from the computational study of the left ventricle where the myocardium was modelled as a micromorphic medium. The results compared favourably to a classical continuum description and also to clinical observations. We therefore conclude that the micromorphic model has potential to simulate the passive and active behaviour of cardiac tissue.

Chapter 9

Concluding Remarks

In this study we investigated the suitability of a micromorphic formulation to model the behaviour of cardiac tissue. The aim of a micromorphic continuum description is to incorporate features of the cardiac tissue microstructure that cannot be included with a classical continuum formulation. Together with an active-stress model and Windkessel model, the micromorphic model was used to simulate the behaviour of the left ventricle during the cardiac cycle. Below we present the main findings and also provide suggestions for future work.

9.1 Summary

Experimental studies show that cardiac tissue is an anisotropic material that exhibits non-linear viscoelastic behaviour. The tissue forms a helical arrangement in the heart with distinct behaviour in three mutually orthogonal directions: the fibre, the sheet and the sheet-normal direction. Classical continuum mechanics is one of the most popular tools to describe and predict the mechanical behaviour of the intact heart and several material models from the literature were presented in Chapter 4. Recent material models simulate passive cardiac tissue as either a transversely isotropic or an orthotropic material.

Literature indicates that some biological materials exhibit non-affine deformation, that is, tissue fibres deform relative to the surrounding matrix material. One disadvantage of classical continuum models is that they lack the ability to describe this type of relative deformation. To overcome some of the limitations associated with classical continuum modelling, we chose a micromorphic continuum description for cardiac tissue. In the formulation presented, micro-directors were introduced to represent the deformation experienced by the cardiac muscle fibres. In this way, the micromorphic approach allowed us to better capture and define the tissue microstructure and to also describe the relative deformation between the cardiac fibres and the bulk material.

The constitutive model used to approximate the passive response of cardiac tissue was separated into three components. The first described the behaviour of the bulk material and was modelled with the orthotropic model by Usyk et al. [125]. The second and third characterised the relative deformation between the cardiac fibres and the bulk material and the pure micro-deformation of the fibres. These components were modelled using Fung-type strain energy functions.

As a working micromorphic model already existed, the biggest part of this study focussed on de-

veloping and implementing a model to describe the active stresses in cardiac tissue. For this, the elastance model proposed by Guccione et al. [51] was modified and coupled with the existing micromorphic model. All implementations were performed in the in-house finite element software, Seska.

The research utilised a patient-specific geometry of a human left ventricle segmented from magnetic resonance images. The pre- and post-processor GiD was used to discretise the geometry and also to apply boundary and loading conditions. The boundary and loading conditions were chosen so as to replicate the different phases of the cardiac cycle, as presented in Chapter 7.

Apart from the micromorphic continuum acting slightly stiffer than the classical continuum, both models produced similar behaviour for passive cardiac tissue. During systole, the difference between the two formulations were more noticeable. One of the main reasons is the approach used to determine the active tension. The active tension is highly dependant on the sarcomere length. The micromorphic and the classical model use different approaches to calculate the sarcomere length, as shown in Equations (4.17) and (6.9). In general, the micromorphic model produced a smaller sarcomere length which resulted in a lower active tension. Nevertheless, both approaches produced ventricular motions that are consistent with clinical observations, such as twisting and longitudinal motion.

One disadvantage of a micromorphic formulation is the running time. Since a larger number of degrees of freedom need to be solve, the micromorphic simulation required much more computation time in comparison with the classical simulation. In this sense, if the simulation time is of great importance the classical continuum modelling is the more viable approach. Additionally, the micromorphic model was also very sensitive to changes in the material parameters during systole.

9.2 Recommendations and Future Work

9.2.1 Geometry

To model the full heart cycle, we assumed that the left ventricle is unloaded at the beginning of diastolic filling, i.e. the start of diastole was chosen as the reference configuration. However, the working heart is never truly unloaded as it is always subjected to some form of passive or active loading. One possibility for future simulations is to explore the inclusion of residual stresses in the continuum model.

With the current model we investigated the mechanics of the left ventricle without taking into account the right ventricle or the two atria. Inclusion of these chambers is expected to add to the overall stiffness of the body and will therefore affect the ventricular mechanics. Therefore, further studies are required whereby the other heart chambers are included in the cardiac model. A model such as this will provide a better understanding of the effects of these chambers.

The effects of the pericardium and surrounding tissue were also excluded from the presented model. These effects may be modelled with the use of spring elements, as was done in [47]. However it might be more accurate (and more interesting) to model the pericardium as a separate continuum body. This will require reconstructing the geometry of the pericardium and modelling its interaction with the ventricle.

9.2.2 Passive Model

Micromorphic modelling of cardiac tissue is unfortunately limited by the lack of experimental data. Further experimental studies that elucidate deformations experienced by the heart fibres are required. More specifically, to improve the calibration of the micromorphic model, we need experimental data that quantify the non-affine deformation of cardiac tissue during a heartbeat.

In the micromorphic formulation presented, one micro-director that aligned with the fibre direction was defined for each point particle. In this way, the fibre component of the strain energy function was modelled as a transversely isotropic material. The passive model may be further improved by introducing two additional micro-directions to describe the sheet and sheet-normal directions. This will allow us to achieve a fully orthotropic material.

9.2.3 Active Model

The elastance model remains a fairly simple model. To model a more complete response, it is recommended that an active model that incorporates crossbridge interactions be used. Furthermore, experimental studies indicate that active tension is present in the sheet (cross-fibre) directions. In our analysis the active tension was assumed to act only in the longitudinal direction of the fibre. Therefore, to enhance the active stress model, the active stress tensor should also have components in the sheet direction, see Equation (4.19).

In our model we assumed that cardiac fibres contract simultaneously. One way to account for the difference in activation time, is to use an equation that describes the spatial distribution of electrical impulse. With such an approach, the fibres near the base will contract before the fibres located in the apical region.

The elastance model was calibrated to fit the pressure-volume curve of the left ventricle. However it will be more appropriate to determine the material parameters based on active-tension experiments performed on isolated cardiac tissue.

9.2.4 Heart Diseases

Generalised continuum approaches perform well when dealing with localised deformation. Research conducted by the Computational Continuum Mechanics Group and their medical collaborators is currently under way to better understand rheumatic heart diseases. One goal is to use the micromorphic model to study the behaviour of diseased hearts. For instance, a micromorphic formulation will allow us to study stress and strain localisation that occur as a result of heart diseases. The model presented in this thesis is of a healthy heart and is therefore useful as a benchmark model. In this way, a diseased model can be compared to our healthy heart model to investigate the effect of the disease.

Furthermore, tissue continuously adapts because of changes to the environment in which the tissue works. Another application of the model is simulating the process of tissue remodelling and investigating how the microstructure changes over time.

In conclusion, the micromorphic model has great potential to provide insight and understanding of the microstructure of the human heart. However, in order to use of the micromorphic model to its fullest potential, more detailed experimental investigations are required, specifically experiments

focusing on the non-affine deformation of heart fibres. Such experimental studies will provide accurate data for calibration purposes and also motivation for microcontinuum models.

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Appendix A

Finite Element Method in Voigt Notation

In Chapter 3, we discussed the finite element methods as it relates to non-linear elastic problems. In this chapter we derive the finite element stiffness matrix and the force vectors in Voigt notation. Voigt notation exploits the symmetry of tensors and offers a more compact form of writing tensors.

A.1 Discretisation of the Weak Form

Consider the weak formulation,

$$\int_{\mathcal{B}_0} \mathbf{S} : \delta \mathbf{E} dV - \int_{\partial \mathcal{B}_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\mathcal{B}_0} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV = 0. \quad (\text{A.1})$$

Since \mathbf{S} and \mathbf{E} are symmetric tensors, both of these tensors contain only six independent components. Using Voigt notation, the second-Piola Kirchhoff stress in vector form is $\bar{\mathbf{S}} = [S_{11}, S_{22}, S_{33}, S_{12}, S_{23}, S_{13}]^T$ and similarly $\delta \bar{\mathbf{E}} = [E_{11}, E_{22}, E_{33}, 2E_{12}, 2E_{23}, 2E_{13}]^T$.

In index notation, the virtual strain $\delta \mathbf{E}$ is written as,

$$\delta E_{ik} = \frac{1}{2} \left(F_{ji} \frac{\partial \delta u_j}{\partial X_k} + \frac{\partial \delta u_j}{\partial X_i} F_{jk} \right). \quad (\text{A.2})$$

Furthermore, if we introduce the shape functions discussed in Chapter 3 we obtain the following approximation of the strain tensor,

$$\delta E_{ik} \approx \sum_{I=1}^n \frac{1}{2} \left(F_{ji} \frac{\partial N_I}{\partial X_k} + \frac{\partial N_I}{\partial X_i} F_{jk} \right) \delta u_{Ij}. \quad (\text{A.3})$$

We use the previous expression to write the virtual strain in Voigt notation as

$$\delta \bar{\mathbf{E}} = \begin{bmatrix} \delta E_{11} \\ \delta E_{22} \\ \delta E_{33} \\ 2\delta E_{12} \\ 2\delta E_{23} \\ 2\delta E_{13} \end{bmatrix} = \sum_{I=1}^n \mathbf{B}_I \delta \mathbf{u}_I, \quad (\text{A.4})$$

where \mathbf{B}_I is given by

$$\mathbf{B}_I = \begin{bmatrix} F_{11} \frac{\partial N_I}{\partial X_1} & F_{21} \frac{\partial N_I}{\partial X_1} & F_{31} \frac{\partial N_I}{\partial X_1} \\ F_{12} \frac{\partial N_I}{\partial X_2} & F_{22} \frac{\partial N_I}{\partial X_2} & F_{32} \frac{\partial N_I}{\partial X_2} \\ F_{13} \frac{\partial N_I}{\partial X_3} & F_{22} \frac{\partial N_I}{\partial X_3} & F_{32} \frac{\partial N_I}{\partial X_3} \\ F_{11} \frac{\partial N_I}{\partial X_2} + F_{12} \frac{\partial N_I}{\partial X_1} & F_{21} \frac{\partial N_I}{\partial X_2} + F_{22} \frac{\partial N_I}{\partial X_1} & F_{31} \frac{\partial N_I}{\partial X_2} + F_{32} \frac{\partial N_I}{\partial X_1} \\ F_{12} \frac{\partial N_I}{\partial X_3} + F_{13} \frac{\partial N_I}{\partial X_2} & F_{22} \frac{\partial N_I}{\partial X_3} + F_{23} \frac{\partial N_I}{\partial X_2} & F_{32} \frac{\partial N_I}{\partial X_3} + F_{33} \frac{\partial N_I}{\partial X_2} \\ F_{11} \frac{\partial N_I}{\partial X_3} + F_{13} \frac{\partial N_I}{\partial X_1} & F_{21} \frac{\partial N_I}{\partial X_3} + F_{23} \frac{\partial N_I}{\partial X_1} & F_{31} \frac{\partial N_I}{\partial X_3} + F_{33} \frac{\partial N_I}{\partial X_1} \end{bmatrix}. \quad (\text{A.5})$$

Using Voigt notation and the property of the scalar product $\mathbf{a} \cdot \mathbf{b} = \mathbf{b}^T \mathbf{a}$, we can rewrite the weak form in Equation (A.1) as

$$\int_{\mathcal{B}_0} \delta \bar{\mathbf{E}}^T \bar{\mathbf{S}} dV - \int_{\partial \mathcal{B}_0} \delta \mathbf{u}^T \mathbf{t} dS - \int_{\mathcal{B}_0} \rho_0 \delta \mathbf{u}^T \mathbf{b} dV = 0, \quad (\text{A.6})$$

where the bar indicates the Voigt form of the stress and strain tensors. Discretising the above expression, we arrive at

$$\bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \mathbf{B}_I^T \bar{\mathbf{S}} d\Omega - \bigcup_{r=1}^{n_r} \sum_{I=1}^m \delta \mathbf{u}_I^T \int_{\Gamma_r} N_I \mathbf{t} d\Gamma - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0, \quad (\text{A.7})$$

where n_r is the number of boundaries with traction loads and m is the number of nodes on the traction surface Γ_r . As in Chapter 3, \bigcup denotes an assembly process.

Because the virtual displacement is arbitrary we can write

$$\bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \mathbf{B}_I^T \bar{\mathbf{S}} d\Omega - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Gamma_e} N_I \mathbf{t} d\Gamma - \bigcup_{r=1}^{r_e} \sum_{I=1}^n \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0. \quad (\text{A.8})$$

The discretised weak formulation in Equation (A.8) is a set of non-linear algebraic equations that needs to be solved using an iterative scheme.

A.2 Linearisation of the Weak Formulation

To find an approximate solution to the weak formulation we first consider the residual G ,

$$G(\mathbf{u}) = \int_{\Omega_0} \mathbf{S} : \delta \mathbf{E} dV - \int_{\partial \Omega_0} \mathbf{t} \cdot \delta \mathbf{u} dS - \int_{\Omega_0} \rho_0 \mathbf{b} \cdot \delta \mathbf{u} dV = 0. \quad (\text{A.9})$$

The objective is to find \mathbf{u} such that the residual is minimised. To find a solution that satisfies the above, we consider a first-order Taylor expansion of the residual,

$$G(\mathbf{u}_{k+1}) = G(\mathbf{u}_k) + \frac{\partial G(\mathbf{u}_k)}{\partial \mathbf{u}} \cdot \Delta \mathbf{u}_k = G(\mathbf{u}_k) + \Delta G_k, \quad (\text{A.10})$$

where $\Delta \mathbf{u}$ is the displacement increment, k the current iteration step and ΔG is the variation in the residual. The updated displacement is the sum of the current displacement and the displacement increment, i.e.

$$\mathbf{u}_{k+1} = \mathbf{u}_k + \Delta \mathbf{u}_k. \quad (\text{A.11})$$

If we assume that the traction and body forces are conservative, i.e. do not depend on the displacement, then the last two terms in $G(\mathbf{u})$ are independent of $\Delta \mathbf{u}$ and the variation in the residual is given as

$$\Delta G = \frac{\partial G(\mathbf{u})}{\partial \mathbf{u}} \cdot \Delta \mathbf{u} = \int_{\Omega_0} \Delta \mathbf{S} : \delta \mathbf{E} dV + \int_{\Omega_0} \mathbf{S} : \Delta \delta \mathbf{E} dV, \quad (\text{A.12})$$

where $\Delta \delta \mathbf{E} = \frac{\partial \delta \mathbf{E}}{\partial \mathbf{u}} \Delta \mathbf{u}$ and $\Delta \mathbf{S} = \frac{\partial \mathbf{S}}{\partial \mathbf{u}} \Delta \mathbf{u}$ are the variations in the virtual strain and the stress. Again using Voigt notation, we write the integrand of the first term in Equation (A.12) as

$$\Delta \mathbf{S} : \delta \mathbf{E} = \Delta \bar{\mathbf{S}} \cdot \delta \bar{\mathbf{E}} = \bar{\mathbf{D}} \Delta \bar{\mathbf{E}} \cdot \delta \bar{\mathbf{E}} = \delta \bar{\mathbf{E}}^T \bar{\mathbf{D}} \Delta \bar{\mathbf{E}}, \quad (\text{A.13})$$

where the bar denotes Voigt notation and $\bar{\mathbf{D}}$ is the matrix form of the fourth-order constitutive tensor $\mathbf{D} = \frac{\partial \mathbf{S}}{\partial \mathbf{E}}$. Discretising of the above term and using Equation (A.5), we have that

$$\Delta \bar{\mathbf{S}} \cdot \delta \bar{\mathbf{E}} \approx \sum_{I=1}^n \sum_{J=1}^n \bar{\mathbf{D}} \mathbf{B}_I \Delta \mathbf{u}_I \cdot \mathbf{B}_J \delta \mathbf{u}_J = \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I^T \mathbf{B}_I^T \bar{\mathbf{D}} \mathbf{B}_J \Delta \mathbf{u}_J. \quad (\text{A.14})$$

The integrand of the second term in Equation (A.12) can be written as

$$\mathbf{S} : \Delta \delta \mathbf{E} = \mathbf{S} : \frac{1}{2} \left(\frac{\partial \Delta \mathbf{u}}{\partial \mathbf{X}}^T \frac{\partial \delta \mathbf{u}}{\partial \mathbf{X}} + \frac{\partial \Delta \mathbf{u}}{\partial \mathbf{X}} \frac{\partial \delta \mathbf{u}^T}{\partial \mathbf{X}} \right) = \mathbf{S} : \left(\frac{\partial \Delta \mathbf{u}}{\partial \mathbf{X}}^T \frac{\partial \delta \mathbf{u}}{\partial \mathbf{X}} \right), \quad (\text{A.15})$$

where we have utilised definition of the Green-Lagrange strain tensor and the fact that the scalar product of a symmetric and a skew-symmetric tensor is zero. Using the definition of the gradient, we write the above expression as

$$\mathbf{S} : \Delta \delta \mathbf{E} = \frac{\partial \Delta \mathbf{u}}{\partial \mathbf{X}} \mathbf{S} : \frac{\partial \delta \mathbf{u}}{\partial \mathbf{X}} = \underbrace{(\Delta \mathbf{u} \otimes \nabla_X)}_{\mathbf{A}} \mathbf{S} : \underbrace{(\delta \mathbf{u})}_{\mathbf{a}} \otimes \underbrace{(\nabla_X)}_{\mathbf{b}}. \quad (\text{A.16})$$

Utilising the rule $\mathbf{A} : \mathbf{a} \otimes \mathbf{b} = \mathbf{a} \cdot \mathbf{A} \mathbf{b}$ and the shape function approximations we obtain

$$\begin{aligned} \mathbf{S} : \Delta \delta \mathbf{E} &= (\Delta \mathbf{u} \otimes \nabla_X) \mathbf{S} : (\delta \mathbf{u} \otimes \nabla_X) = \delta \mathbf{u} \cdot (\Delta \mathbf{u} \otimes \nabla_X) \mathbf{S} \nabla_X \\ &\approx \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I \cdot \left(\Delta \mathbf{u}_J \otimes \frac{\partial N_J}{\partial \mathbf{X}} \right) \mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} \\ &= \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I^T \underbrace{\left(\underbrace{\Delta \mathbf{u}_J}_{\mathbf{a}} \otimes \underbrace{\frac{\partial N_J}{\partial \mathbf{X}}}_{\mathbf{b}} \right)}_{\mathbf{c}} \underbrace{\mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}}}_{\mathbf{c}}. \end{aligned} \quad (\text{A.17})$$

Furthermore we apply the rule $(\mathbf{a} \otimes \mathbf{b}) \mathbf{c} = (\mathbf{c} \cdot \mathbf{b}) \mathbf{a}$ to obtain the following,

$$\mathbf{S} : \Delta \delta \mathbf{E} \approx \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I^T \left(\mathbf{S} \frac{\partial N_I}{\partial \mathbf{X}} \right) \cdot \frac{\partial N_J}{\partial \mathbf{X}} \Delta \mathbf{u}_J = \sum_{I=1}^n \sum_{J=1}^n \delta \mathbf{u}_I^T \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \Delta \mathbf{u}_J. \quad (\text{A.18})$$

By substituting Equations (A.14) and (A.18) back into Equation (A.12), the variation in the residual becomes

$$\Delta G_k = \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \delta \mathbf{u}_I^T \mathbf{B}_I^T \bar{\mathbf{D}} \mathbf{B}_J \Delta \mathbf{u}_J d\Omega + \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \delta \mathbf{u}_I^T \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \Delta \mathbf{u}_J d\Omega. \quad (\text{A.19})$$

From Equation (A.7) we know that

$$G(\mathbf{u}_k) = \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \mathbf{B}_I^T \bar{\mathbf{S}} d\Omega - \bigcup_{r=1}^{n_r} \sum_{I=1}^m \delta \mathbf{u}_I^T \int_{\Gamma_r} N_I \mathbf{t} d\Gamma - \bigcup_{e=1}^{n_e} \sum_{I=1}^n \delta \mathbf{u}_I^T \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega = 0. \quad (\text{A.20})$$

Finally, substitution of the above into Equation (3.106) leads to

$$\begin{aligned} G(\mathbf{u}_{k+1}) &= \underbrace{\delta \mathbf{u}_I^T \left(\bigcup_{e=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \mathbf{B}_I^T \bar{\mathbf{S}} d\Omega \right)}_{\mathbf{F}_{int}} - \underbrace{\delta \mathbf{u}_I^T \left(\bigcup_{r=1}^{n_r} \sum_{I=1}^m \int_{\Gamma_r} N_I \mathbf{t} d\Gamma + \bigcup_{n=1}^{n_e} \sum_{I=1}^n \int_{\Omega_e} \rho_0 N_I \mathbf{b} d\Omega \right)}_{\mathbf{F}_{ext}} \\ &+ \underbrace{\delta \mathbf{u}_I^T \left(\bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \mathbf{B}_I^T \bar{\mathbf{D}} \mathbf{B}_J d\Omega + \bigcup_{e=1}^{n_e} \sum_{I=1}^n \sum_{J=1}^n \int_{\Omega_e} \left(\frac{\partial N_I}{\partial \mathbf{X}} \right)^T \mathbf{S} \frac{\partial N_J}{\partial \mathbf{X}} \mathbf{I} d\Omega \right)}_{\mathbf{K}} \Delta \mathbf{u}_J. \end{aligned} \quad (\text{A.21})$$

In Equation (A.21) we have indicated the internal force vector \mathbf{F}_{int} , the external force vector \mathbf{F}_{ext} and the stiffness matrix \mathbf{K} . In terms of the stiffness matrix and the force vectors the residual is given as

$$G(\mathbf{u}_{k+1}) = \delta \mathbf{u}_I^T \left(\mathbf{F}(\mathbf{u}_k)_{ext} - \mathbf{F}(\mathbf{u}_k)_{int} + \mathbf{K}(\mathbf{u}_k) \Delta \mathbf{u}_J \right). \quad (\text{A.22})$$

Appendix B

Ethical Considerations

The ethical clearance documents for the use of MRI scans of three healthy heart are given here. Ethical clearance was provided by the UCT Faculty of Health Sciences.

APPLICATION FORM

Please Note:



Any person planning to undertake research in the Faculty of Engineering and the Built Environment (EBE) at the University of Cape Town is required to complete this form before collecting or analysing data. The objective of submitting this application prior to embarking on research is to ensure that the highest ethical standards in research, conducted under the auspices of the EBE Faculty, are met. Please ensure that you have read, and understood the EBE Ethics in Research Handbook (available from the UCT EBE Research Ethics website) prior to completing this application form: <http://www.ebe-uct.ac.za/ethicsresearch/>

APPLICANT'S DETAILS

Name of principal researcher, student or external applicant		Merina Kamper
Department		Civil Engineering
Preferred email address of applicant		KAMPMER015@uct.ac.za
Your Degree: e.g. MSc, PhD, etc.		MSc
Credit Value of Research: e.g., 60/120/180/360 etc.		120
Name of Supervisor (if supervised):		Al-Prof Sebastian Shaballa
If this is a research contract, indicate the source of funding/sponsorship		NRF funded
Project Title		Active Contraction of the Left Ventricle with Cardiac Tissue Modelled as a Micromorphic Medium

- I hereby undertake to carry out my research in such a way that:
- there is no apparent legal objection to the nature or the method of research; and
 - the research will not compromise staff or students or the other responsibilities of the University;
 - the stated objectives will be achieved, and the findings will have a high degree of validity;
 - limitations and alternative interpretations will be considered;
 - the findings could be subject to peer review and publicly available; and
 - I will comply with the conventions of copyright and avoid any practice that would constitute plagiarism.

SIGNED BY	Full name	Signature	Date
Principal Researcher/ Student/external applicant	Merina Kamper		26/09/2018

APPLICATION APPROVED BY	Full name	Signature	Date
Supervisor (where applicable)	Al-Prof Sebastian Shaballa		26/09/2018
HOD (or delegated nominee)	Dr Yllon Ramphal		19 Oct 2018
Final authority for all applicants who have answered NO to all questions in Section1, and for all Undergraduate research (Including Honours).			
Chair : Faculty EBR Committee			
For applicants other than undergraduate students who have answered YES to any of the above questions.			



UNIVERSITY OF CAPE TOWN
Faculty of Health Sciences
Human Research Ethics Committee



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22 November 2018

HREC REF: 679/2018

Prof Ntobeko Ntusi

Cardiology

Medicine

J46.53

Dear Prof Ntusi

PROJECT TITLE: ACTIVE CONTRACTION OF THE LEFT VENTRICLE WITH CARDIAC TISSUE MODELLED AS A MICROMORPHIC MEDIUM (SUB-STUDY LINKED TO 554/2017 & R055/2015) (MSc Candidate - M. Kamper)

Thank you for submitting your study to the Faculty of Health Sciences Human Research Ethics Committee.

It is a pleasure to inform you that the HREC has **formally approved** the above-mentioned study.

Approval is granted for one year until the 30 November 2019.

Please submit a progress form, using the standardised Annual Report Form if the study continues beyond the approval period. Please submit a Standard Closure form if the study is completed within the approval period.

(Forms can be found on our website: www.health.uct.ac.za/fhs/research/humanethics/forms)

Please quote the HREC REF in all your correspondence.

Please note that the ongoing ethical conduct of the study remains the responsibility of the principal investigator.

Please note that for all studies approved by the HREC, the principal investigator **must** obtain appropriate Institutional approval, where necessary, before the research may occur.

The HREC acknowledges that the student, Marina Kamper will also be involved in this study.

Yours sincerely

PROFESSOR M. BLOCKMAN

CHAIRPERSON, FHS HUMAN RESEARCH ETHICS COMMITTEE

Federal Wide Assurance Number: FWA00001637.

Institutional Review Board (IRB) number: IRB00001938

This serves to confirm that the University of Cape Town Human Research Ethics Committee complies to the Ethics Standards for Clinical Research with a new drug in patients, based on the Medical Research Council (MRC-SA), Food and Drug Administration (FDA-USA), International Convention on Harmonisation Good Clinical Practice (ICH GCP), South African Good Clinical Practice Guidelines (DoH 2006), based on the Association of the British Pharmaceutical Industry Guidelines (ABPI), and Declaration of Helsinki (2013) guidelines.

The Human Research Ethics Committee granting this approval is in compliance with the ICH Harmonised Tripartite Guidelines E6: Note for Guidance on Good Clinical Practice (CPMP/ICH/135/95) and FDA Code of Federal Regulation Part 312.61, 312.62 and 312.63.